

TEMPERATURE DISTRIBUTION IN THIN PLATES
WELDED UNDERWATER

James Andrews Staub

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TEMPERATURE DISTRIBUTION IN THIN PLATES
WELDED UNDERWATER

by

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A.B. MATHEMATICS, ST. VINCENT COLLEGE
(1963)

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WELDED UNDERWATER

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JAMES ANDREW STAUB, JR.

Submitted in partial fulfillment of the requirements for the degrees of Naval Engineer and Master of Science in Mechanical Engineering at the Massachusetts Institute of Technology, June, 1971.

ABSTRACT

During underwater (wet) welding the metal within the heat-affected-zone experiences a rapid fluctuation in temperature which results in a loss of ductility and a susceptibility to stress-corrosion cracking. The object of this study was to develop and verify a model which would predict the temperature distribution such that the metallurgical and mechanical properties of the underwater weld could be determined.

The equation of heat flow in two-dimensions with source terms was converted to its finite difference equivalent and solved by successive iterations with a digital computer. The solution incorporated all non-linear terms in the original equation and accounted for the variation of the thermal properties of the metal with temperature. The interior boundary in the temperature field was found to have a definite shape which was a function of the important welding parameters. The shape function was correlated by experiment using methods established for normal surface welding. Heat losses were assumed to occur by boiling and radiation heat transfer.

The predicted and measured temperature distributions were compared. It was found that the model predicted peak temperatures with reasonable accuracy but that comparison of cooling curves showed considerable disagreement. Since the predicted cooling curve presented a sharper slope than the measured value it was considered that the assumption of conventional boiling heat transfer processes was invalid.

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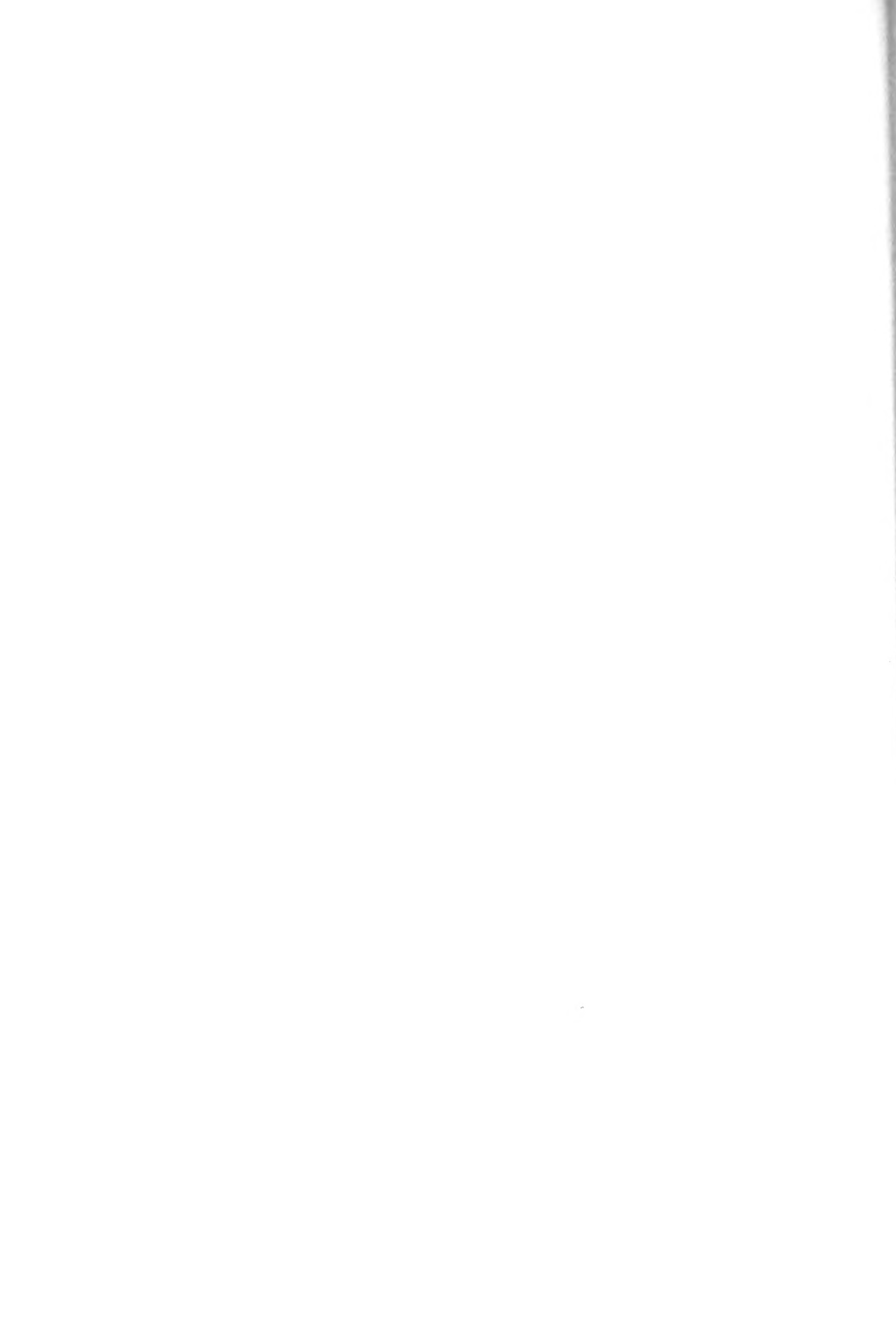


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LIST OF SYMBOLS

c_p	SPECIFIC HEAT (BTU/lb-°R)
E	ARC VOLTAGE (VOLTS)
H_m	HEAT CONTENT (BTU/FT ³)
I	ARC CURRENT (AMP.)
K_0, K_1	BESSEL FUNCTIONS
k	THERMAL CONDUCTIVITY (BTU/FT-HR-°R)
L	THICKNESS OF THE WORKPIECE
ℓ_m	MAX. POOL LENGTH (IN)
p	LUMPED PARAMETER
q	HEAT FLUX RATE (BTU/FT ² -HR)
q_c	CONVECTIVE HEAT LOSS
q_b	BOILING HEAT LOSS
q_r	RADIATIVE HEAT LOSS
r_m	RADIAL DISTANCE TO POOL BOUNDARY (IN)
T	TEMPERATURE (°R)
V	WELDING SPEED (FT/HR)
y_m	MAX. POOL HALF-WIDTH (IN)



GREEK LETTERS

α	THERMAL DIFFUSIVITY (FT /HR)
α_{min}	MINIMUM THERMAL DIFFUSIVITY FOR METAL
θ	ANGLE (DEGREES)
λ	$1/(2\alpha_{Min})$, HR/FT ²
ρ_L	MASS DENSITY OF SATURATED WATER (lb/ft ³)
ρ_V	MASS DENSITY OF SATURATED VAPOR (WATER)
σ	SURFACE TENSION (lb/ft)

SUBSCRIPTS

o	AMBIENT TEMPERATURE
M	MELTING TEMPERATURE

SUPERSCRIPTS

$*$	NORMALIZED QUANTITY
$-$	AVERAGE VALUE

CHAPTER I

INTRODUCTION

The object of this thesis was to develop a model of an underwater welding process such that the temperature history of the metal could be predicted. Particular attention was given to temperature changes within the heat affected zone. The model was constructed with consideration for a) the variation with temperature of the thermal properties of the base metal, b) a finite heat source of predictable configuration, c) the heat losses to the liquid environment, and d) the non-linear components of the heat flow equation. Three major areas within the literature were investigated: a) underwater, b) normal surface welding, and c) boiling heat transfer.

This study was conducted in recognition of the potential benefits that could be realized if underwater welding were a viable process. Predicted trends in ocean exploration and exploitation indicate the use of large structures that could only be fabricated or repaired on site within the liquid environment. Even today, considerable savings could be realized if ship hulls, off-shore drilling platforms, buoys and salvage work were constructed or repaired without explicit dependence on the requirement that the welding operation be conducted in air. Unfortunately, underwater welds are not structurally sound due to reduced ductility, and



susceptibility to stress corrosion and hydrogen cracking.

A reasonably accurate model, one that predicts peak temperatures and cooling rates, was seen as a necessity to examine the metallurgical changes occurring in the weld zone as function of welding parameter.

A. UNDERWATER WELDING: STATE-OF-THE-ART

Underwater welding, as a process, is best defined by considering its two subcategories: wet and dry welding. Dry underwater welding is a welding procedure in which the liquid environment is excluded from the vicinity of the workpiece by a structure such as a caisson. The structure is large enough to contain one or more workers plus the necessary welding equipment. The atmosphere within the chamber is usually at depth pressure and may, or may not, consist of life-supporting gases. This system's utility is restricted to symmetrical workpieces about which the chamber can be fitted and sealed. Hence, dry underwater welding is most used by oil companies laying undersea pipeline or repairing the submerged supporting structure on off-shore drilling rigs. Essentially, this procedure differs from normal air welding only by the pressure of the atmosphere surrounding the work.

Wet welding occurs when the liquid environment is allowed to interact directly with the joining process: the



most notable result being a rapid quenching of the liquid metal. The utility of this process is not restricted by the geometry of the workpiece, but rather by the quality of the product. No wet weld is considered safe when used in load bearing structures. The rapid quench produces a weld that has 50% of the ductility of air welds, and is subject to extreme stress corrosion cracking and hydrogen embrittlement. The object of the thesis, then, was to construct, and verify, a model that might predict the temperature cycles in a metal as it is being wet welded.

The necessity for such a model is twofold: a) considerable time and expense could be saved if current ocean structures could be repaired or fabricated in situ, and b) future projections of man's involvement with the ocean require the use of structures that can only be fabricated, by reason of their size, within the ocean environment. Currently, the most economical means of joining metal is by welding, and it would be extremely expensive if each new process, each new structural material, with their multitude of variables, had to be investigated individually by actual experimentation. Unfortunately, the state of wet welding has generally stagnated since its inception during WW II.

From its development in the 1940's until today, the most widely used wet welding process is shielded metal-arc. This is the same method used in air welding and is commonly



referred to as "covered electrode" or "stick electrode" welding. Generally, wet welding has merely borrowed whatever equipment was already developed for air welding. The only adaptation to the new environment being a coat of waterproof material for the electrode and an electrically-insulated holder. Only recently have the British begun to develop electrodes specifically designed for use under water. However, the details of the program are considered proprietary information.

The basic research conducted thus far has been an examination of the mechanical properties of the flux and the completed weld. Thus, Waugh and Eberlein²³ investigated flux coverings to determine those most suitable for underwater application.* Work at the Naval Engineering Station²⁶ produced some factors of safety that must be applied when wet welding. Hibshman and Jensen⁸ evaluated the properties of welds in low-carbon steel, under water and in air. The data revealed that ductility was markedly reduced. As far as it is known, however, no attempt has been made, either on a theoretical or experimental basis, to determine the temperature history of the metal during the wet welding process.

B. NORMAL SURFACE WELDING

A review of the literature for air welding, however, indicated that considerable effort has been devoted to

* Numbers refer to listing in Bibliography.



predicting the temperature cycles during the welding process. In 1935, Rosenthal¹⁶ obtained an exact solution to a modified heat flow equation. He simplified his model by neglecting non-linear terms, treating the conductivity as constant, and assuming a point source. However, the resulting equation for temperature distribution contained all the important welding parameters: weld speed, heat input, plate thickness, and the thermal properties of the metal. Furthermore, the equation was in quasi-steady state form so that the location of the point source was at the origin of a stationary coordinate system. Experimental verification by Rosenthal¹⁶ and Schmorber indicated that the model was reasonably accurate at some distance from the source of heat, but in adjacent areas (within the heat affected zone, or haz) the degree of error was unacceptably large. The primary cause for the discrepancy was due to the nature of the heat source which in actuality has a very definite shape. Nevertheless, Rosenthal's initial attempt provided the impetus which spanned a succession of attempts to include all the important variables in one well organized model.

Grosh, Hawkins, and Travant⁷ investigated the effect of considering the thermal properties as a function of temperature and obtained slightly better results. Barry³ modified Rosenthal's model by including possible heat losses to the environment but concluded that, at least for convection, the effect is negligible in comparison to the heat conducted



in the metal.

The most important attempts to derive a satisfactory model concerned those which examined the shape of the heat source. In 1952, Wells²⁷ proposed an equation from which the length and width of the weld pool could be determined as a function of the heat input. Subsequent experiments indicated that the relationship was reasonably accurate. Christensen, Davies, and Gjermundsen⁴ examined the shape of the weld pool and its relation to a wide variety of welding conditions and associated variables. While their objective was not to formulate a model, they did present experimental data from which viable conclusions were drawn: a) pool width and length increase with heat input for a given weld speed and plate thickness, b) the pool is wider than predicted by point source theory, and c) the pool has a distinct "tear-drop" shape. These basic observations permitted the derivation of a very successful model of temperature distribution during welding.

In 1967, Pavelic¹³ and Tanbakuchi²⁰ constructed and verified a model which combined all the disparate elements necessary to predict temperature variations in the haz. Unlike Rosenthal, they considered the variation of conductivity and diffusivity with temperature, a finite heat source, heat losses to the environment through convection and radiation, non-linearities in the heat flow equation,

and preheating of the metal from arc radiation. Their approach to the derivation of the model for a tungsten-inert-gas welding process formed the basis for this thesis.

C. BOILING HEAT TRANSFER

The literature on boiling heat transfer was investigated selectively with respect to possible models relating to sub-cooled pool boiling on horizontal plates at a non-uniform temperature.

The complexity of the boiling process has prevented researchers from developing a single formulation covering the wide range of temperatures over which boiling will occur. Figure 1-1, reported by Rohsenow,¹⁵ is considered to represent the regimes of boiling, each regime having a uniquely formulated model descriptive of heat transfer.

Rohsenow¹⁵ has correlated the data of Addoms for nucleate boiling on platinum wire in water at saturation temperature. Although the data was given for a specific geometry, Rohsenow observed that the same results applied to horizontal surfaces.

Correlations in the transition regime have not been firmly developed. Jordan,²⁵ reporting the data of Drew and Mueller, Farber and Scoriah, and Braunleh, concluded that if the data were plotted on a log-log graph, they would lie along a straight line from the critical flux (or, burnout)

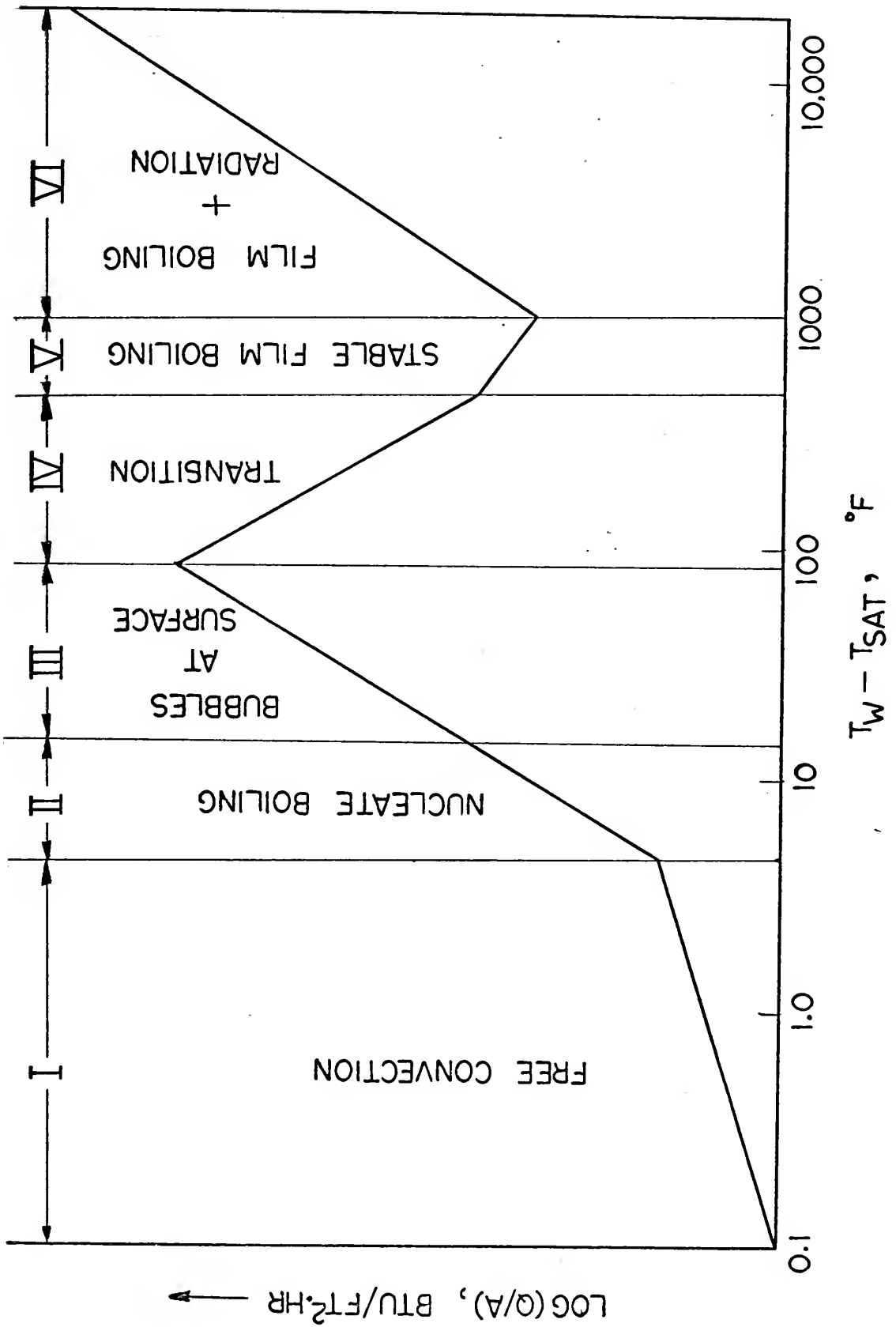


FIGURE 1-1



to the minimum film boiling heat flux. The slope of the line varied with the surface finish: the steepest slope occurring with a mirror smooth finish.

Chang, Botushansky, and Berenson²⁵ developed correlations for film boiling heat flux from horizontal surfaces. Jordan²⁵ has reported that the Borishansky relation fits the data for pool boiling in water.

D. SUMMARY OF PROCEDURE

The development of a model for predicting temperature changes in the heat affected zone of an underwater weld required the construction of a theoretical model followed by experimental verification. The model was to encompass the latest techniques, and should be reasonably accurate in determining peak temperatures and cooling rates.

The derivation of the model followed the approach taken by Pavelic¹³ and Tanbachuchi²⁰ in their work for surface GTA (Gas Tungsten Arc) welding. The two-dimensional equation of heat flow with external sources was reduced to quasi-steady state form and transformed to its finite difference equivalent. The external heat sources were examined and the proper relations chosen. Then, with the help of Pavelic's results, the equation describing the shape of the molten weld pool was derived. The resulting derivation included numerical constants which were determined through experimentation.

Finally, the computer program was written. The method for laying out the finite difference grid was that developed by Tanbachuchi. His solution was refined by having the shape of pool internally generated instead of being supplied as input.

The experiment was conducted in two parts. First, the correlation constants in the pool configuration equation were determined by measuring the dimensions of molten weld pools while varying the welding parameters: Heat input and welding speed. Then, temperatures occurring during welding were determined by laying weld beads on submerged plates in which thermocouples had been installed. The results of the final experiment were compared with the values predicted by the model.

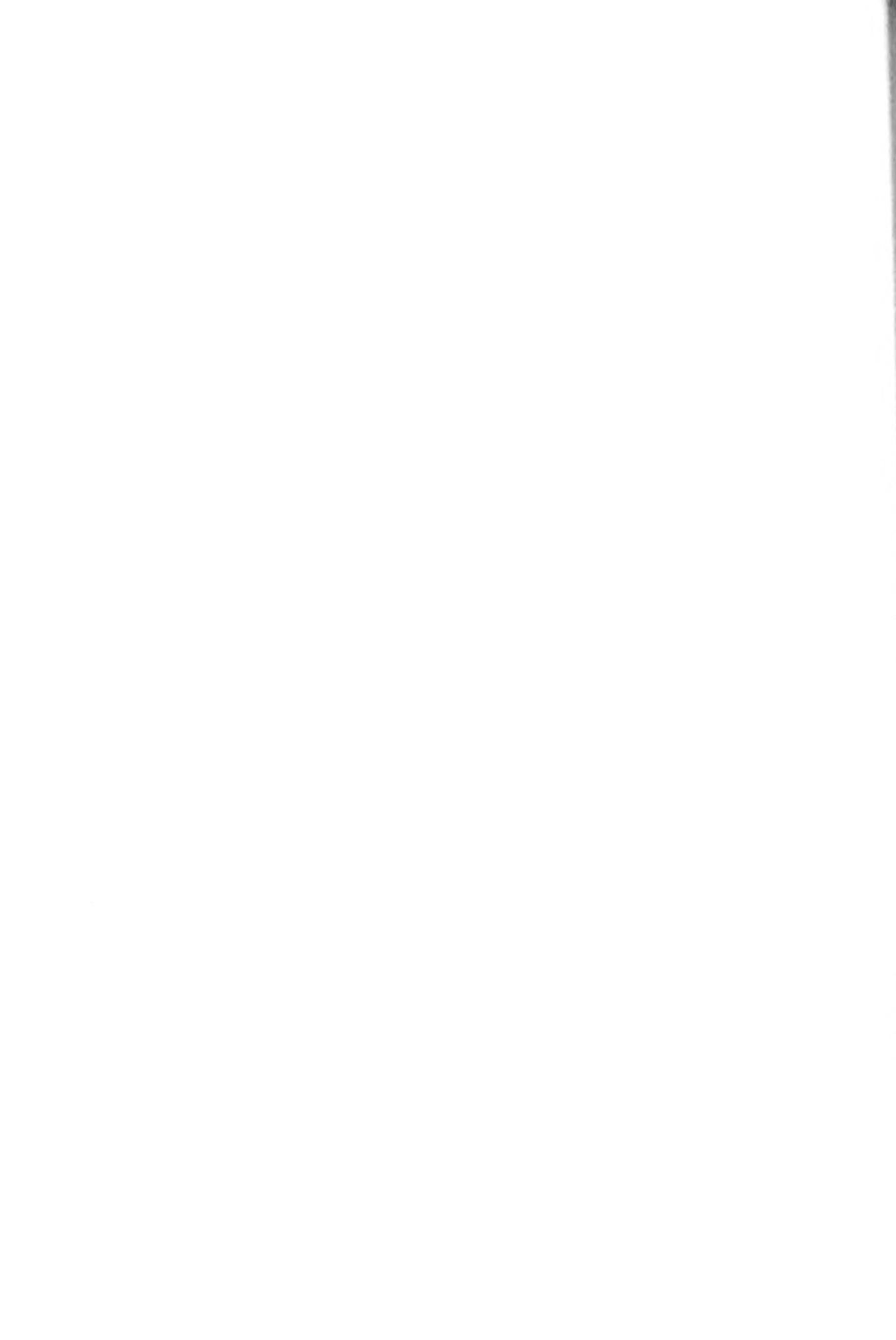


CHAPTER II

MATHEMATICAL MODEL AND COMPUTATIONAL TECHNIQUEA. PRELIMINARY OBSERVATIONS

Welding is a process of joining two or more materials together by applying sufficient quantities of heat to a local area at their interface such that melting occurs and a union is achieved. The amount of heat required is on the order of 10^4 BTU/HR and, for structural applications, the manner of generation is by electric arc. The method for directing the arc is by an electrode which may (consumable) or may not (non-consumable) contribute metal to the finished bond. The electrode concentrates the arc and creates a small pool of molten metal which is typically "tadpole" or "rain-drop" shaped when viewed in the plane perpendicular to the arc axis. As the electrode is drawn along the material a bead of weld (i.e., a chain of overlapping pools) is formed which represents the physical evidence of the interface bond.

The largest percentage of the heat developed by the arc goes into raising the temperature of the material to its melting point. The remainder is accounted for by conduction, convection, and radiation, joule heating, and pre-heating of the material in the vicinity of the arc. Pre-heating will occur in all but submerged arc processes and is due to the divergence of the arc which takes the appearance of a cone with the electrode at its apex. The central part of the



arc, or the plasma column, is responsible for the creation of the molten pool. The remainder of the arc, while not generating sufficient heat to melt the material, will cause preheating. Hence, heat may flow into the material directly from the pool and from the arc by pre-heating.

While the above heat balance has been observed in normal surface welding, its complete relevance to the underwater welding process is less certain. Conduction and radiation have importance, but convective losses must be modified by boiling heat transfer phenomena. Joule heating losses are usually estimated to account for 1-2% of machine power and, hence, may be neglected. A search of the literature did not reveal any information on the dynamics of an arc surrounded by water so that the effect of pre-heating could not be determined. For the experiment, it was decided that the effect, if it did exist, could be reduced to negligibility by the proper choice of electrodes. Consequently, an electrode that could be placed directly on the metal surface without shorting out was selected. These "drag" type electrodes operated in a manner such that the metal core melted faster than the flux covering the flux coat, which came in direct contact with the metal, formed a hollow barrel and prevented arc diffusion. Consequently, it was assumed that pre-heating was negligible and all heat flowed into the metal through the molten pool. Interaction with the

environment was assumed to occur by radiation and boiling heat transfer.

An approximation to the real welding process was then seen to be a boundary-value problem in two-dimensions with negative source terms. The interior boundary in the field was viewed to be formed by the contour of the molten pool; the exterior boundaries were considered as the four edges of a rectangular surface. The pool contour temperature was known to be the melting temperature of the material with the edges at ambient. Radiation and boiling heat transfer were included as the negative sources.

B. EQUATION OF HEAT FLOW

The governing differential equations are presented in this section. In the sections to follow the equations are reduced to their finite difference equivalents, the boundaries are defined, and the relations for the source terms are determined.

The general equation of heat flow with source terms is,

$$\nabla \cdot (K \nabla T) + W_i = \rho c \frac{DT}{Dt} ,$$

where k , ρ , and c are the conductivity, density and specific heat of the material and may be a function of distance, time, and temperature. W_i represents the source terms and has the units, $\text{BTU/FT}^3\text{-HR}$. In partial differential notation for two dimensions, the equation becomes,

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{1}{k} \frac{\partial K}{\partial T} \left[\left(\frac{\partial T}{\partial x} \right)^2 + \left(\frac{\partial T}{\partial y} \right)^2 \right] + \frac{W_i}{k} = \frac{\rho c}{k} \frac{\partial T}{\partial t} .$$

The equation in this form is for a coordinate system fixed to the control surface defining the temperature field. Since the interior boundary (i.e., the pool contour) moves through the field with velocity, V , the velocity of the electrode, the temperature at a given point will change with time. To eliminate the time dependence, the coordinate system was transferred to the moving boundary with the following transformation equations:

$$\bar{x} = x - Vt$$

$$\bar{t} = t$$

where \bar{x} and \bar{t} are coordinates in the new system. (The details of the calculation appear in Appendix 1.) With the transformation, the equation of heat flow became,

2.1

$$\frac{\partial^2 T}{\partial \bar{x}^2} + \frac{\partial^2 T}{\partial \bar{y}^2} + \frac{1}{k} \frac{\partial k}{\partial T} \left[\left(\frac{\partial T}{\partial \bar{x}} \right)^2 + \left(\frac{\partial T}{\partial \bar{y}} \right)^2 \right] + \frac{W_i}{k} = - \frac{V}{\alpha} \frac{\partial T}{\partial \bar{x}} .$$

The temperatures with respect to the new system could now be considered constant and were viewed to form isothermal contours in the plane of the field with the pool contour forming the innermost. Hence, the pool boundary could be called the isotherm of melting.

The equation was put in non-dimensional form by

defining normalizing relations. It was decided to make the exterior boundary conditions homogeneous and the conditions along the pool contour unity. Hence, the normalized temperature was defined,

$$T^* \equiv \frac{T - T_O}{T_M - T_O} \quad 2.2$$

where T_M is the melting temperature of the material (approximately 2700°F for mild steel) and T_O the ambient temperature.

The normalized dimensions were defined to be,

$$x^* \equiv \frac{x}{x_O} \quad , \quad y^* \equiv \frac{y}{x_O} \quad .$$

when these definitions were substituted back into equation 2.1, it was found that x_O should be defined as

$$x_O \equiv \alpha / V \quad ,$$

where α refers to a specific value of the thermal diffusivity. From a stability analysis of equation 2.1, recommended by Ames², it was determined that

$$\alpha = 2 \alpha_{Min}$$

where α_{Min} represents the minimum thermal diffusivity for the material. Finally, the normalized dimensions became

$$x^* = \frac{XV}{2 \alpha_{Min}} \quad , \quad y^* = \frac{YV}{2 \alpha_{Min}} \quad 2.3$$

by substitution of equations 2.2 and 2.3 into equation 2.1, the heat flow relation was obtained in non-dimensional form,

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{1}{k} \frac{dK}{dT} \left[\left(\frac{\partial T}{\partial x} \right) + \left(\frac{\partial T}{\partial y} \right)^2 \right] + \frac{4 \alpha_{Min}^2}{KV^2 (T_M - T_O)} (W)_1 = \frac{2 \alpha_{Min}}{\alpha} \frac{\partial T}{\partial x}$$

2.4

where the "Star" superscript has been dropped. (See Appendix 4 for details of the derivation.)

B. FINITE DIFFERENCE SOLUTION

Equation 2.4 was then converted to its finite difference equivalent so as to make it amenable to computer solution. Following the procedure recommended by Greenspan⁶ for elliptic non-linear partial differential equations. A finite difference grid node (Figure 2-9) was defined and the bounding temperatures were approximated by a Taylor expansion about T_O :

$$T_1 = T_O (\pm) S_i \frac{\partial T_O}{\partial x} + \frac{S_i^2}{2} \left[+ \frac{\partial^2 T_O}{\partial x^2} \right] + \dots \quad i = 1, 2, 3, 4.$$

Now, in the limit as the S_i (i.e., the grid spacing) approach zero the variation in temperature between the bounding temperatures along a given axis will represent the actual rate of change of temperature at the node. For unequal grid spacing, these changes were defined as:

$$\frac{\partial T_O}{\partial x} = a_0 T_0 + a_1 T_1 + a_2 T_2 \quad (2-5_{a,b})$$

$$\frac{\partial T_O}{\partial y} = \acute{a}_0 T_0 + \acute{a}_1 T_3 + \acute{a}_2 T_4$$



The weighing coefficients were found by comparison with Taylor's expansion through combination of like terms. The second partials were similarly defined

$$\frac{\partial^2 T_O}{\partial x^2} = a_3 T_0 + a_4 T_1 + a_5 T_2 \quad (2-6_{a,b})$$

$$\frac{\partial^2 T_O}{\partial y^2} = \acute{a}_3 T_0 + \acute{a}_4 T_3 + \acute{a}_5 T_4$$

and the coefficients found from Taylor's expansion. Substitution of equations 2.5 and 2.6 into 2.4 resulted in the finite difference equivalent for the equation of heat flow:

$$T_0 = \left\{ T_1 \left[\frac{2\alpha + S_2}{\alpha S_1 (S_1 + S_2)} \right] + T_2 \left[\frac{2\alpha - S_1}{\alpha S_2 (S_1 + S_2)} \right] + T_3 \left[\frac{2}{S_3 (S_3 + S_4)} \right] \right. \\ \left. + T_4 \left[\frac{2}{S_4 (S_3 + S_4)} \right] + \frac{4\alpha_{Min}^2}{KV^2 (T_M - T_O)} [W_1] \right. \\ \left. + \frac{1}{k} \frac{dK}{dT} \left[\left[\left(\frac{S_1 - S_2}{S_1 S_2} \right) T_O + \left(\frac{S_2}{S_3 (S_3 + S_4)} \right) T_1 + \left(\frac{S_1}{S_2 (S_1 + S_2)} \right) T_2 \right]^2 \right. \right. \\ \left. \left. + \left[\left(\frac{S_3 - S_4}{S_3 S_4} \right) T_O + \left(\frac{S_4}{S_3 (S_3 + S_4)} \right) T_3 + \left(\frac{S_3}{S_4 (S_3 + S_4)} \right) T_4 \right]^2 \right] \right\} / B \quad 2.7$$

where,

$$B = \left(\frac{2}{S_1 S_2} + \frac{2}{S_3 S_4} + \left(\frac{S_2 - S_1}{\alpha (S_1 S_2)} \right) \right)$$

The details of the derivation are presented in Appendix

Equation 2.7 operated on a grid (i.e., an assemblage of nodes) placed over the temperature field by assigning to each node point a value consisting of the weighted "average" of the four bounding temperatures. Since the exterior points in the field were at a normalized value of zero it was decided that calculations should begin at the pool contour ($T - 1$) and move outward. A series of iterations over the grid were performed until the desired accuracy was obtained.

C. THE BOUNDARY CONDITIONS

This section deals with the derivation of the equation describing the shape of the pool contour (i.e., the interior boundary in the temperature field) and a determination of the location of the exterior boundaries. The treatment of the problem follows the approach taken by Pavelic¹³ in his study of the tungsten inert gas process for normal air welding. Pavelic's method for finding the shape of the pool can be summarized as follows:

1. Rosenthal's two-dimensional equation with a point source of heat is analyzed to obtain the relations predicting the maximum dimensions of the pool: length and width.
2. These theoretical relations are correlated with experimentally-obtained data.



3. The contours of actual pools are measured and a curve fitted equation is obtained. This equation is a function of the maximum pool dimensions.

Hence, for any given set of welding variables, Pavelic could predict the contour of the pool. The details of Pavelic's analysis have been summarized in Appendix 3.

It was found that the maximum theoretical dimensions of pool are

$$Y_{\max} = r_m \left\{ 1 - \left[\frac{K_0(r_m)}{K_1(r_m)} \right]^2 \right\}^{\frac{1}{2}}$$

For the maximum half width, and

$$X_{\max} = 2r_m \frac{K_0(r_m)}{K_1(r_m)}$$

for the maximum length. The argument of the Bessel functions, K_0 and K_1 , is the radial distance from the origin of the arbitrary coordinate system to the pool contour and was found to be a function of the lumped parameter, p :

$$p = \frac{3.415 EI}{2\pi \alpha L H_m}$$

The relationship of x_{\max} , Y_{\max} , and r_m to the pool is presented in Figure 2.1. When plotted on a log-log graph, the pool width ($= 2Y_{\max}$) vs p and x_{\max} vs p appear as a straight line and, hence, experimental data could be correlated by

POOL CONTOUR VARIABLES

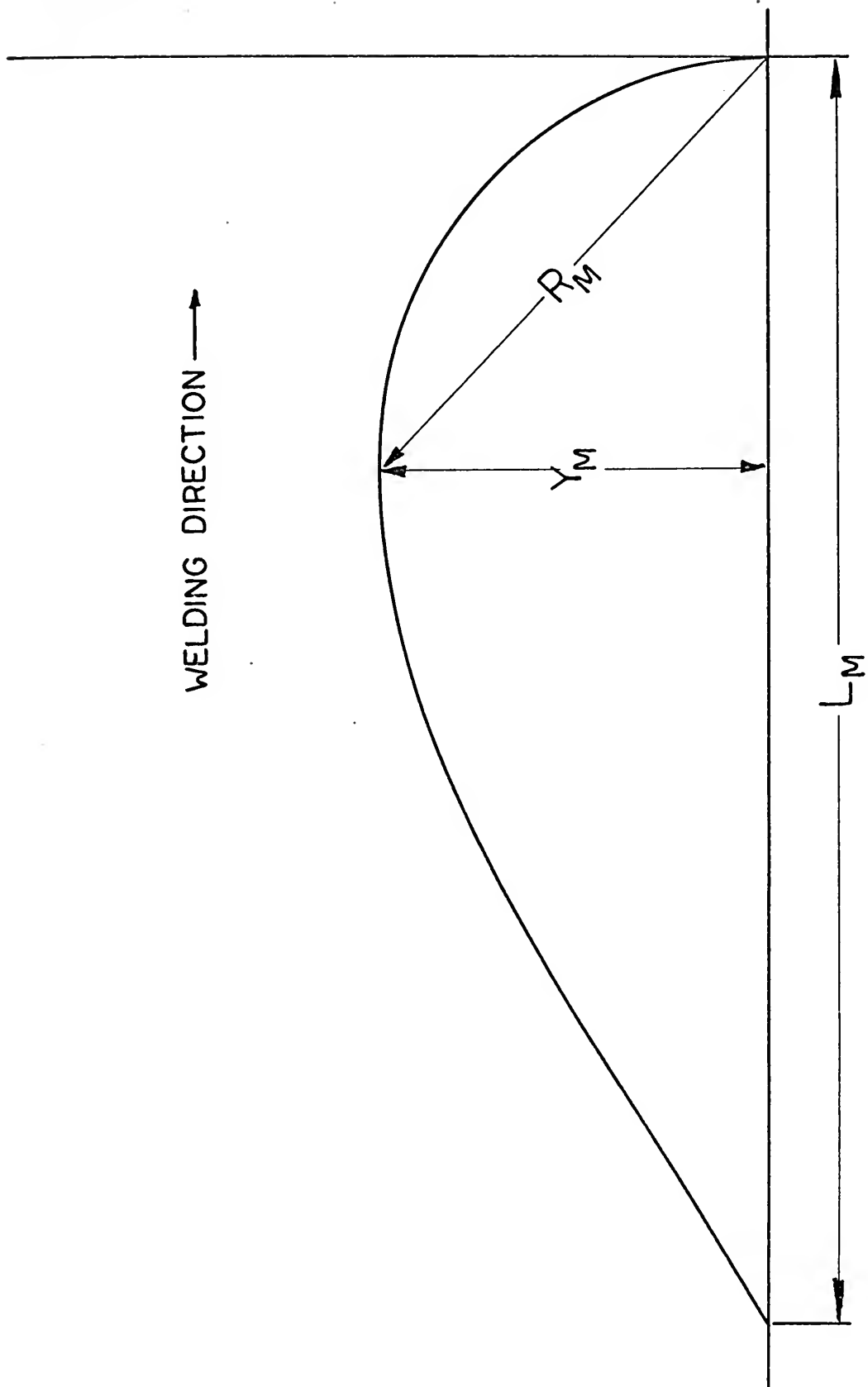


FIGURE 2-1

the relations:

$$2y_{\max} = ap^b$$

$$x_{\max} = cp^d$$

with a , b , c , and d becoming the correlation constants.

Experiments were conducted therefore to find values for the constants. Pools were made by welding beads on thin rectangular plates. At a predetermined point in the welding process the molten metal was ejected from the pool by a jet of high pressure air leaving a crater whose edges formed the desired contour. Only those craters nearest the ideal two-dimensional configuration were retained for measurement. An ideal 2-D crater has been defined in Figure 2.2. The maximum length and width were measured with a Bausch and Lomb microscope containing a movable micrometer table. A computer program was then written to fit a curve through the data points from which values for the correlation constants were obtained. The program with sample input is presented in Appendix 7.

Two major problems were recognized early in the attempt to produce craters for measurement: a) a search of the literature did not reveal any similar experiments conducted with stick electrodes, and b) it was very difficult to time the release of air to eject the metal during the under water welding process. Without the certainty that the metal was being ejected at the proper instant it was considered



2-D WELD POOL

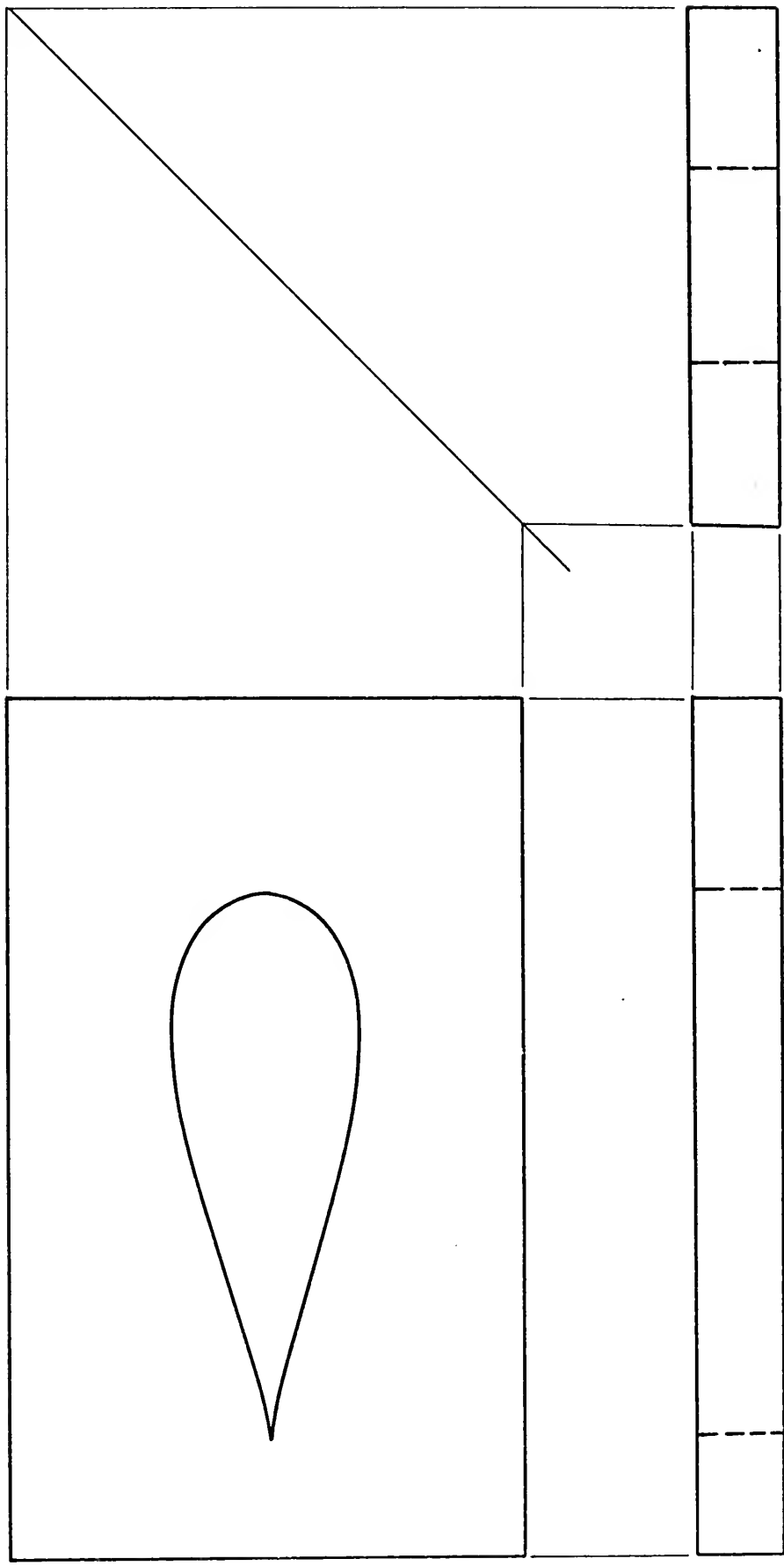


FIGURE 2-2



possible that the craters might be distorted and measurements valueless. The lack of prior experiments conducted in air denied a basis for comparison to determine if the underwater data were reasonable. Consequently, it was decided to conduct an initial series of runs in air where crater production could be closely controlled, merely to serve as a crude reference for the underwater results.

Plates of two different thicknesses, $1/8"$ and $1/4"$, were used and the data plotted as shown in Figures 2.3 and 2.4. The results for the $1/8"$ plate appeared to be in good agreement with the predicted relation, but those for the $1/4"$ plates were not. The lack of alignment was attributed to the failure to obtain two-dimensional pools in the larger specimens. The available power was insufficient to create the required configuration, and, as a result, the crater formed was only a hollow depression on the surface of the specimen. Significantly, an increase in power (ExI) would have resulted in a larger value for the lumped parameter, p , which would have shifted the data points to the right and brought them more in line with the predicted relationship. (It was noted, however, that a line drawn through the $1/4"$ data points was approximately parallel to the theoretical line. This relationship suggested that the penetration of the arc could also be correlated with the lumped parameter, p , so that the equation for a 3-D pool might eventually be derived. No attempt was made to do so, however.) From the plot for air welding it was observed



MAXIMUM 2-D POOL LENGTH WELDED IN AIR

32

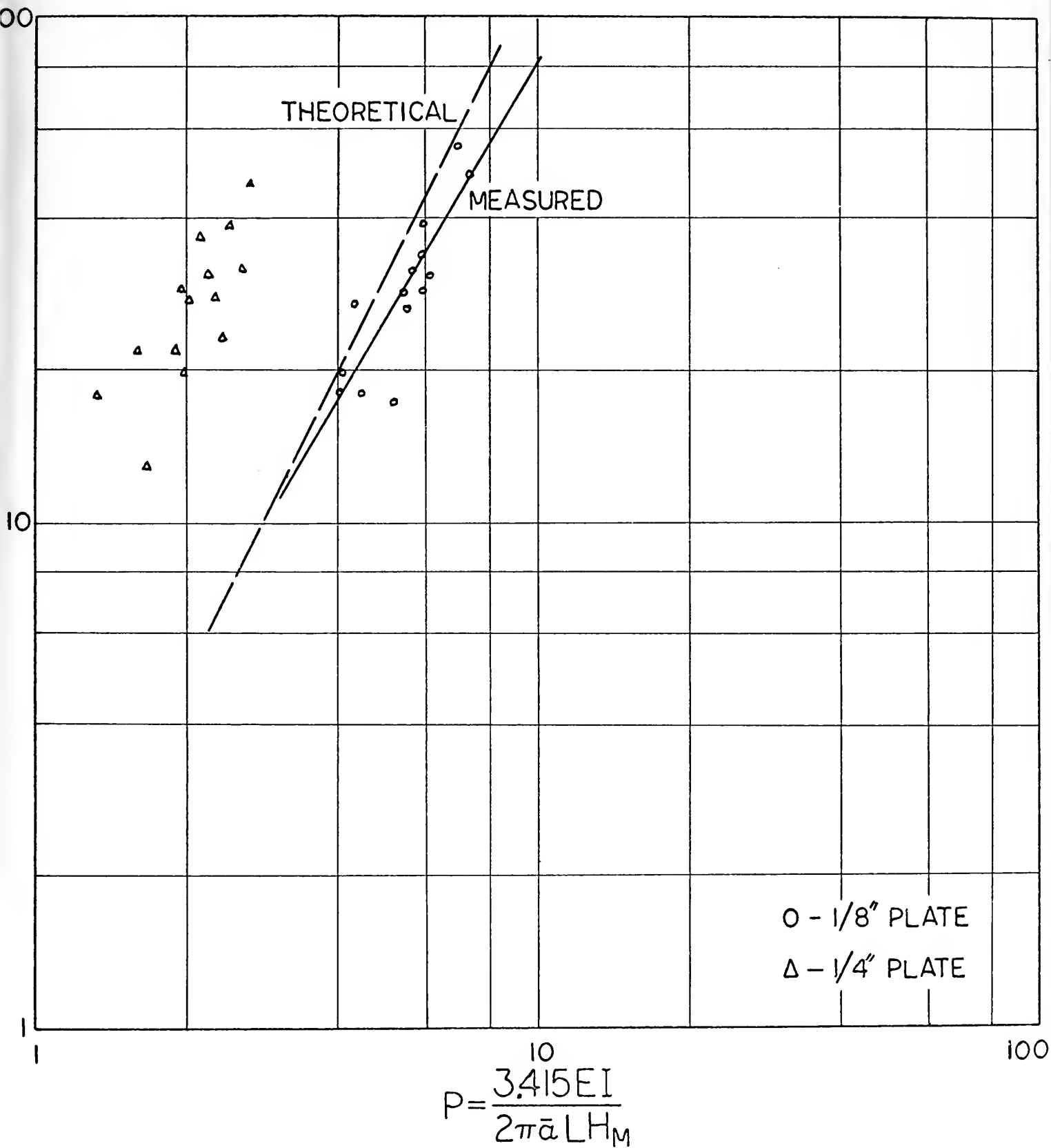


FIGURE 2-3

MAXIMUM 2-D POOL WIDTH WELDED IN AIR

33

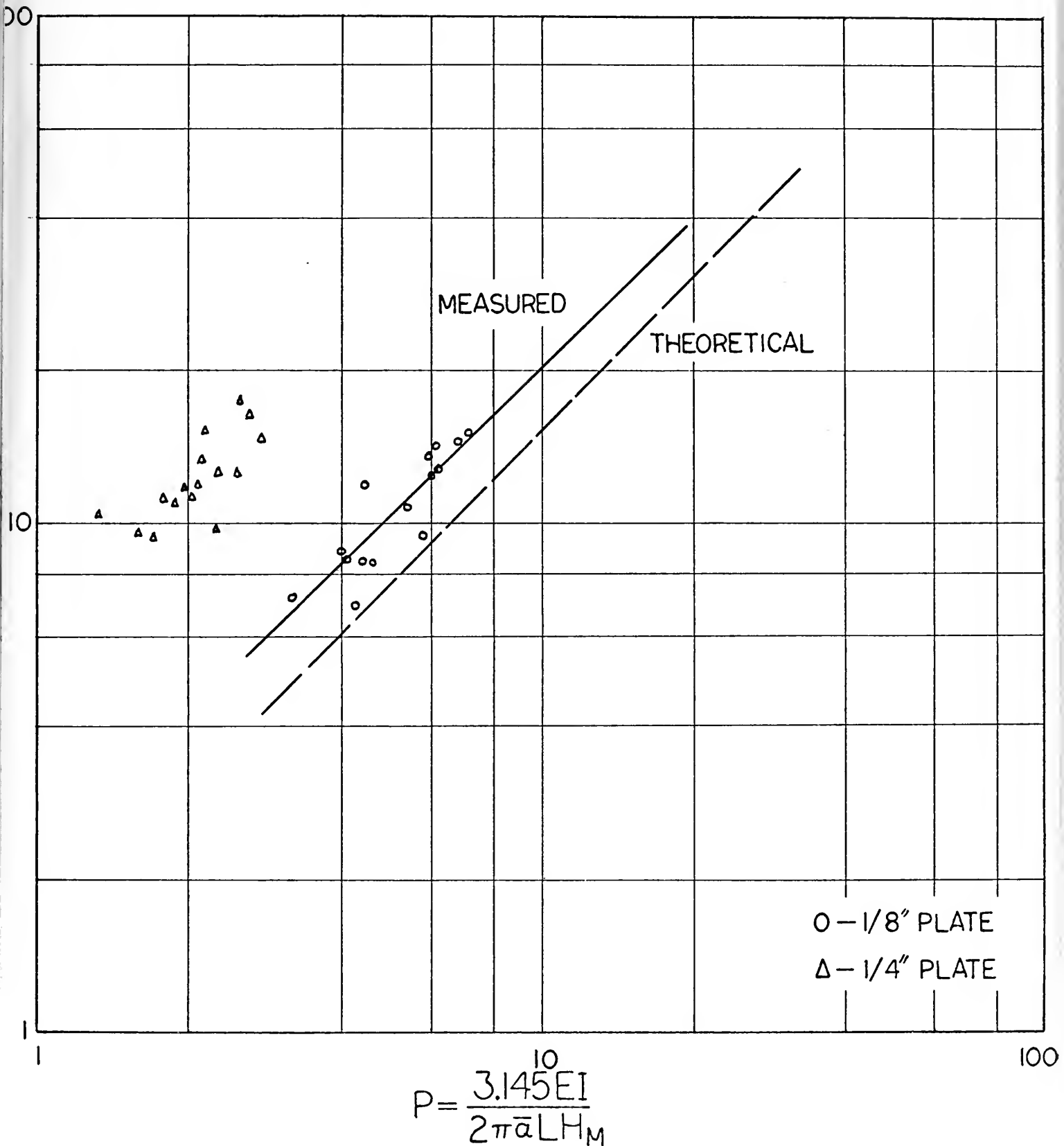


FIGURE 2-4



that a) 2-D pools would not be obtained in water for 1/4" thick specimens and b) the width of the underwater pool should be larger than predicted while the length should be smaller.

Pools were then created under water, measurements taken and the data plotted as shown in Figures 2.5 and 2.6. The location of the data points with respect to the theoretical line was noted and compared with air welded results. Since the location was as expected it was assumed that the measurements were reasonable. Curves were fitted through the data points. These curves were not found to be parallel either to the predicted or the air welded lines, and the reason for the divergence (considered more fully in Chapter IV) could not be adequately explained due to the small size of the data sample and data scatter. Nevertheless, the correlation constants obtained from the fitted curve were used. It was consequently assumed that the length and width of an underwater pool were related to the lumped parameter, p , according to the equations:

$$x_M^* = 4.42 p^{1.01}$$

$$W_M^* = 3.16 p^{0.67}$$

where x_m^* and W_M^* are dimensionless quantities and

$$x_M = \frac{24 x_m^*}{M^{\alpha} \text{Min}} \quad , \quad \text{in.}$$

$$W_M = \frac{24 W_M^*}{M^{\alpha} \text{Min}} \quad , \quad \text{in.}$$

MAXIMUM 2-D POOL WIDTH

35

UNDERWATER WELD

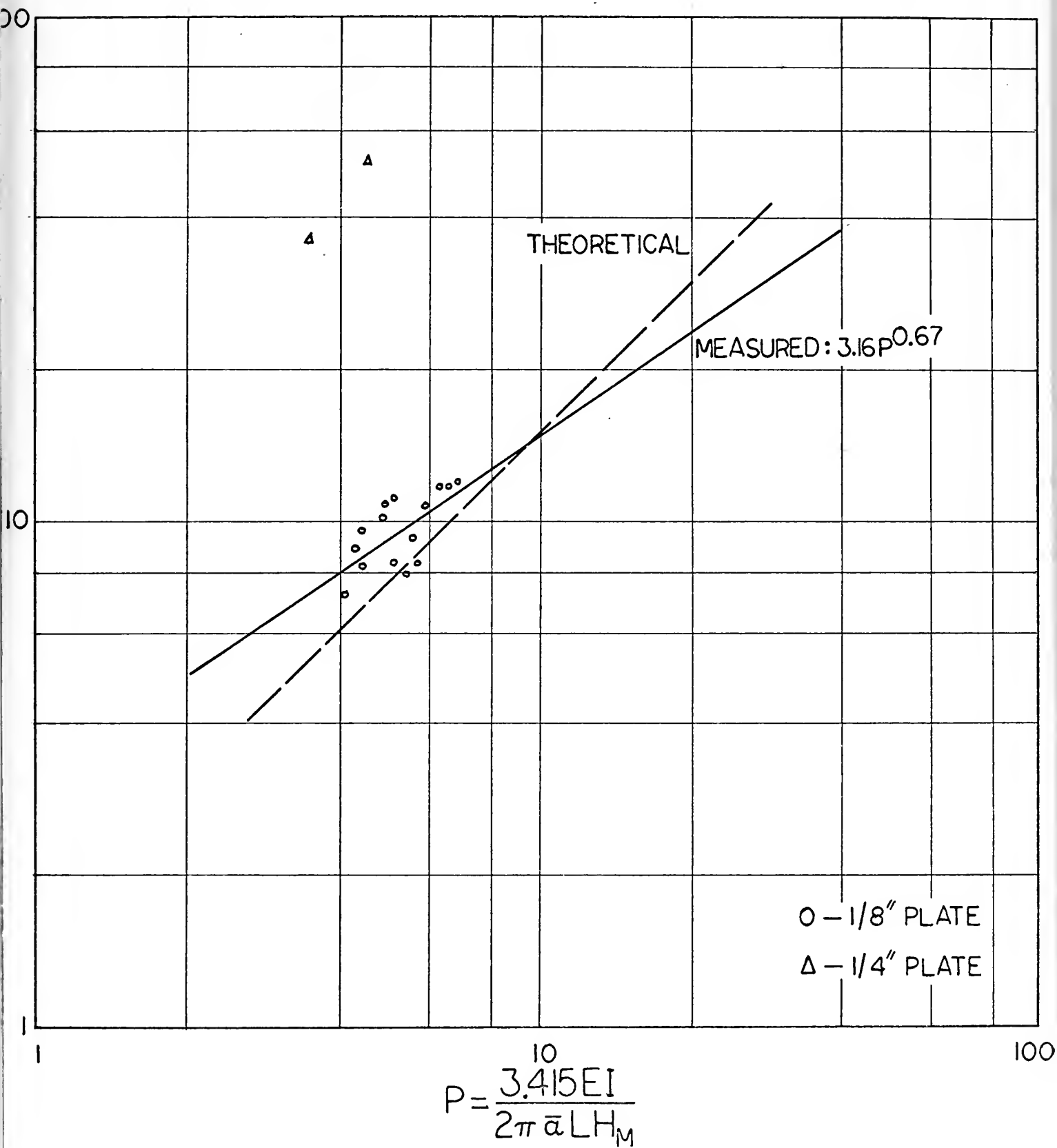


FIGURE 2-5

UNDERWATER WELD

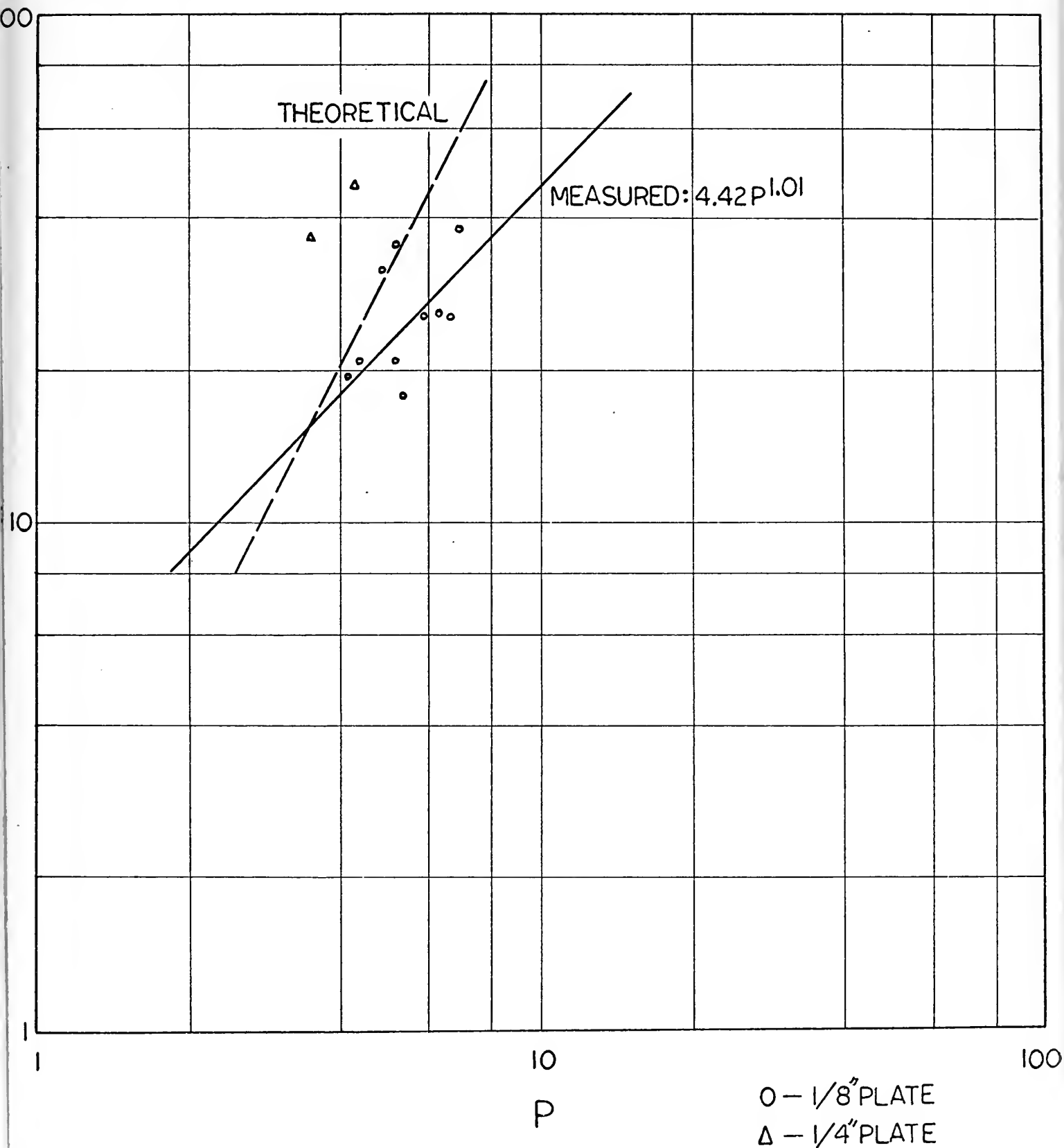


FIGURE 2-6

with the maximum pool dimension relations defined it was now possible to derive an equation that would predict the complete contour of the pool. Pavelic suggested curve fitting an equation of the form:

$$y^2 = Bx^a(x_m - x)^b \quad 2.8$$

to which the boundary conditions were applied,

- (1) $y = 0$ @ $x = x_m$
- (2) $y = 0$ @ $x = 0$
- (3) $\dot{y} = 0$ @ $y = W_m/2$, $x = x_c$
- (4) $\dot{y} = \infty$ @ $y = 0$, $x = 0$

where x_c is the distance to the point of maximum width.

After the applying of the boundary conditions, it was found that

$$B = \frac{(W_m/2)^2 (a+b)^{a+b}}{(ax_m)^a (bx_m)^b}$$

where a and b are the correlating constants. The details of the derivation have been included in Appendix 3.

Values for the constants were obtained by measuring the actual pool contours and fitting the resulting measurements with equation 2.8. A computer program, employing an M.I.T. non-linear least squares curve fitting routine, was written and has been presented in Appendix 7. It was hoped that a large number of pools would be available for measurement in order that the calculated correlating constants would be generally applicable. While a large sample size

was obtained for air welded pools, only one under water pool of acceptable quality was produced. Due to the accelerated solidification of the molten metal in water and crater walls took on a jagged appearance which made measurements difficult although the maximum dimensions were generally easily determined. Consequently, it was decided to base the calculation of the constants on data taken from the air welded pools and then compare the predicted contour with the measurements from the single under water pool. The result of this approach is presented in Figure 2.7. The predicted contour gave a wider head and narrower tail than was measured but the overall shape was considered satisfactory. Nevertheless, it was realized that the contour equation was based on a sample size of one (i.e., the single under water pool). However, it seemed reasonable to assume that the general shape of a pool would be independent of the atmosphere in which it was created since the dimensions were most strongly influenced by heat input and welding speed. With this assumption, the correlation constants

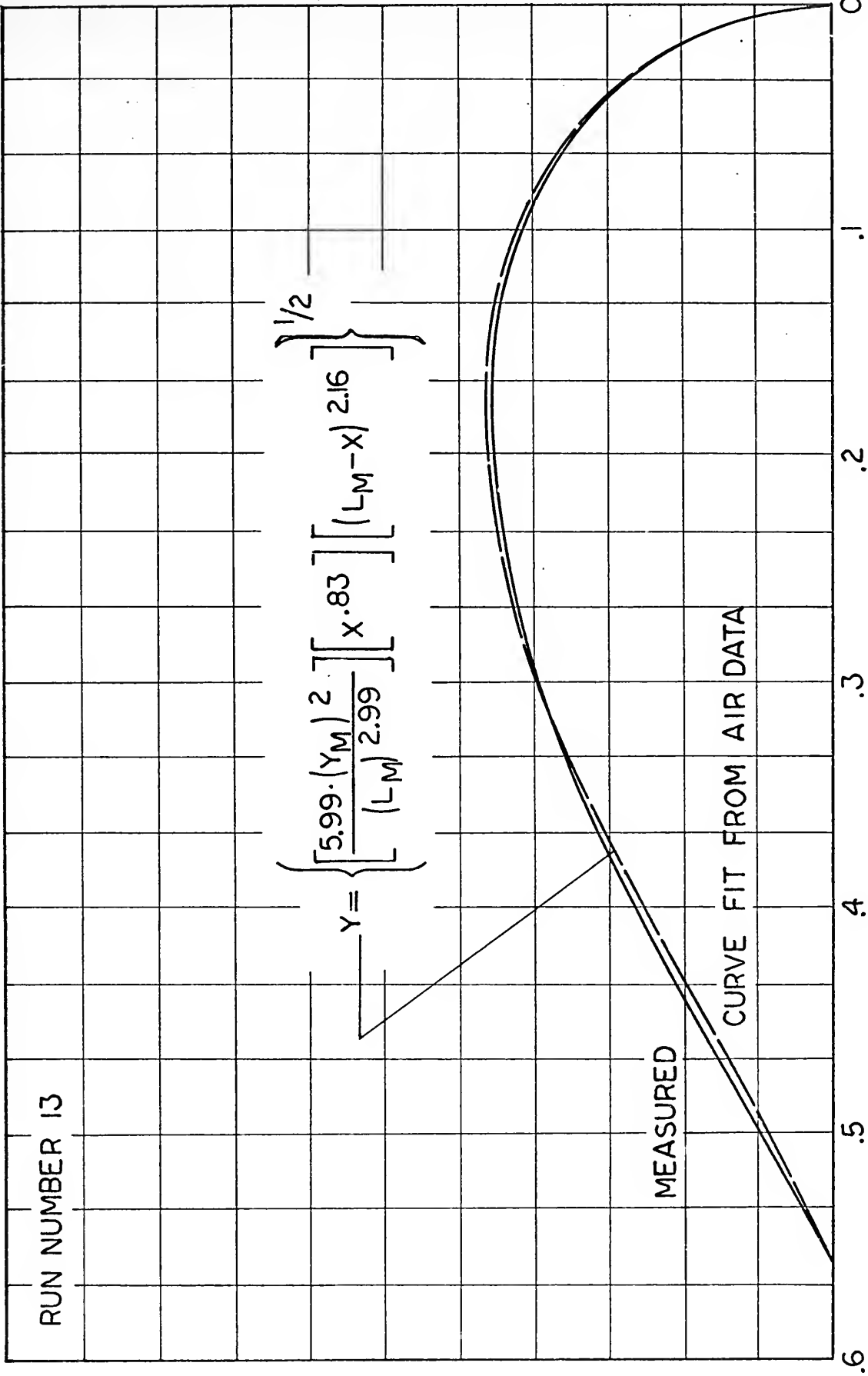
$$a = 0.83$$

$$b = 2.16$$

were used to predict the shape of the under water pool contours.



POOL CONTOUR



$$Y = \left\{ \left[\frac{5.99 \cdot (Y_M)^2}{(L_M)^{2.99}} \right] \left[X^{0.83} \right] \left[(L_M - X)^{2.16} \right] \right\}^{1/2}$$

LENGTH, IN.

FIGURE 2-7

D. DERIVATION OF THE HEAT LOSS TERMS, W_i

For this model heat losses were assumed to occur through boiling and radiation. Although convective heat flux could be considered, it was estimated that these losses would account for only 3% of the total heat input to the specimen (15,000 to 20,000 BTU/Hr) and were therefore neglected. Consequently, heat losses were considered according to three distinct regimes as depicted in Figure 2.8.

Nucleate boiling is characterized by the termination of bubbles at the heated surface which grow in size with increasing temperature. The agitation of the liquid near the surface caused by bubble activity -- formation, collapse, or ascent -- maintains a circulation of relatively cooler liquid near the surface which can result in heat fluxes on the order of 10^6 BTU/FT²-HR.

When the surface temperature reaches a value 50-90°F above saturation the number of bubbles forming has increased to the point where they begin to overlap and create a solid film of vapor. The vapor film is not immediately stable and may collapse back to the heated surface where the process repeats itself. In this transition region the cooler liquid is brought into contact with the surface only intermittently and heat flux rates are not as high as they would be during nucleation. As temperature increases the film grows more stable, insulating the surface and reducing heat transfer. Finally, the film becomes fully stable and the heat flux

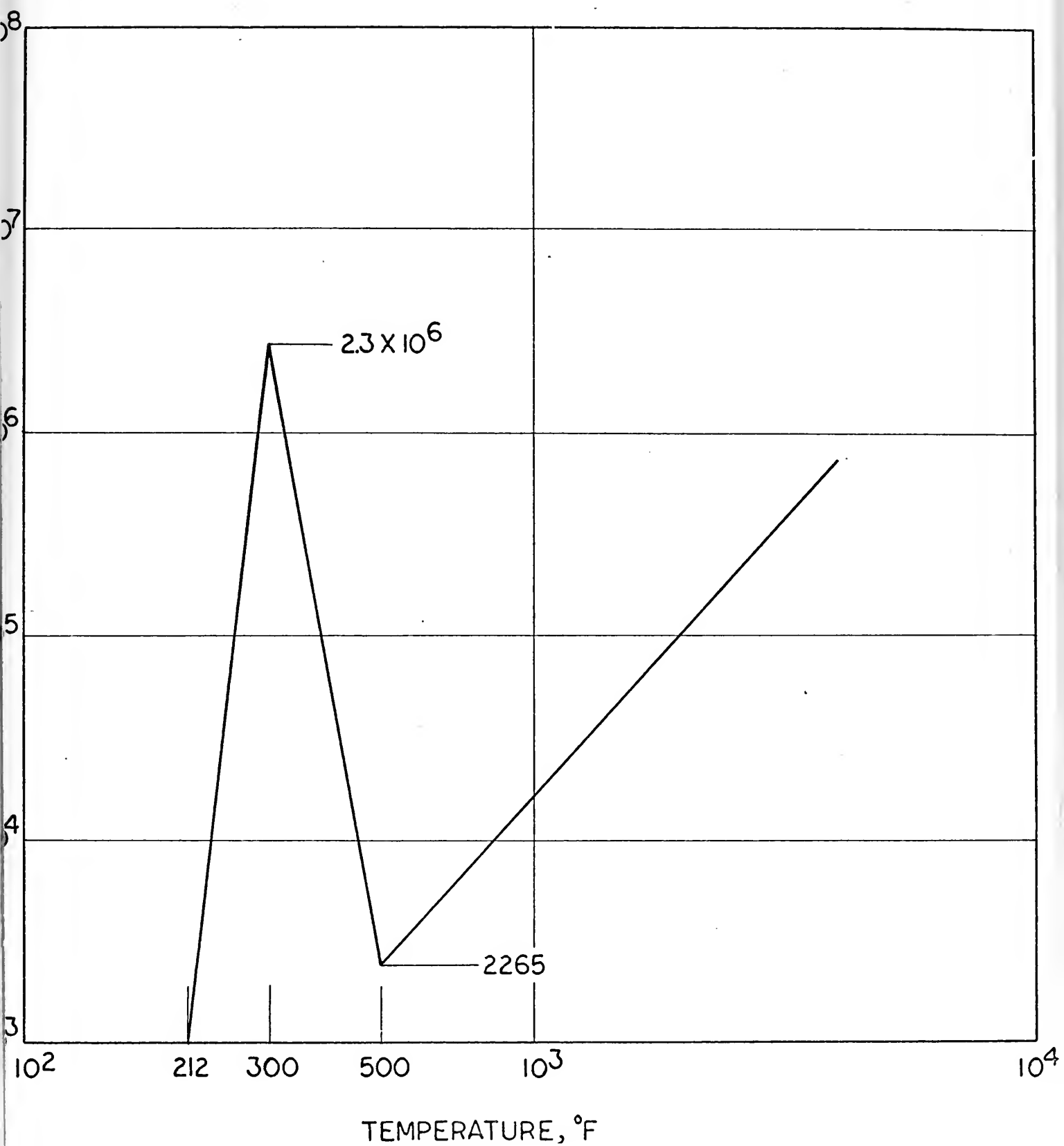


FIGURE 2-8

reaches its minimum value. At higher temperatures heat loss through radiation becomes the predominant term.

The complexity of the entire process required that each regime be treated separately in determining the appropriate relation for calculating heat flux. First, however, it was necessary to define the temperatures which bounded each regime. Since nucleation begins near saturation temperature, the lower bound was chosen to be 212°F. Burnout, or the end of nucleation, will occur at a temperature which depends upon the condition of the heated surface, the amount of dissolved gases in the liquid, and the ambient pressure. If the solubility of air at 14.7 psia is 6m l/l, then, from data reported by Rohsenow¹⁵ burnout will occur near 260°F. However, a mildly oxidized surface increases the temperature, by an unknown amount. Consequently, 300°F (as the nearest round number) was defined as the temperature at which nucleation terminates and the transition regime begins. For the minimum heat flux, the data of Borishansky and Berenson as reported by Jordan²⁵ indicated that transition ends near 500°F.

Once the boundary temperatures for each boiling regime were established it was necessary to determine the suitable heat flux relations. Rohsenow's¹⁵ correlation of Addoms' data suggested that heat flux in the nucleate boiling regime varied according to the cube of the temperature difference,

$$T_w - T_{sat}$$



The properties of the liquid and vapor were calculated at saturation and the proper relation was found to be

$$q_N = [1.542(T_W - T_{SAT})]^3, \quad \text{BTU/ft}^2\text{-hr.}$$

The details of the calculation have been presented in Appendix 2. As indicated by Jordan²⁵, the data for film boiling on a horizontal surface in water is predicted with reasonable accuracy by Borishansky's relation. In this regime the heat flux was found to be

$$q_F = 32.4(T_W - T_{SAT})^{0.75}, \quad \text{BTU/hr-ft}^2,$$

where calculations have been summarized in Appendix 2.

Heat flux during transition boiling has not been well defined. However, data reported in the literature indicates that measures fluxes tend to fall along a straight line plotted with log-log coordinates. This relationship suggested the derivation of an equation with the burnout and minimum film boiling heat fluxes as boundary conditions. As summarized in Appendix 2, the results of the derivation were found to be

$$q_T = 2265.11 \left[\frac{288}{(T_W - T_{SAT})} \right]^{5.86}$$

where

$$300^\circ \leq T_W \leq 500^\circ\text{F}.$$

The equation for radiation heat transfer was determined by assuming the surface of specimen exchanges heat with the surrounding water. Hence,

$$q_R = 0.1317 \times 10^{-8} (T_w^4 - T_{SAT}^4) \quad , \text{ BTU/hr-ft}^2 \quad .$$

E. COMPUTATIONAL TECHNIQUE

The computer program written to calculate the temperature distribution consisted of the four main operations:

1. The shape of the pool contour is generated.
2. The grid is layed out to form a computational block.
3. The grid nodes are initialized using Rosenthal's temperature distribution equation.
4. Successive iterations are carried out over the grid until the desired accuracy is obtained.

A listing of the program is presented in Appendix 7, and instructions on its use in Appendix 6.

Pool Contour Generation

A value for the lumped parameter, p , is calculated from the input data and the maximum dimensions of the pool are determined according to the correlation derived in Chapter II. Then, using the contour equation, which is a function of the maximum dimensions, the computer generates a series of points which define the shape of the interior boundary in the temperature field. If projected on the x-axis (i.e., the longitudinal center line) of the pool, the points would be

spaced 5/100 of an inch apart. This spacing allows from 10 (0.5 inch pool length) to 20 points in defining the shape of the boundary.

Grid Layout

Except for minor alterations, the method of arranging the grid is identical to approach developed by Tanbakuchi²⁰ and is sized approximately 100 lines to the inch in a rectangular array. The grid is configured such that the specified thermocouple location appears at the intersection of a row and a column. A schematic of the grid forming a computational block is presented in Figure 2.9. Columns are arranged parallel to the center line (i.e., column number 1) of the pool; rows perpendicular. The intersection of column 1 and row 1 mark the origin of the coordinate system. In the output, rows forward, or in the direction of welding, are identified as positive quantities.

Initializing the Grid Nodes

Nodes within and on the pool boundary are given values of one ($T^* = 1$) while the exterior boundary nodes are set to zero. As suggested by Tanbakuchi,²⁰ the intervening nodes are initialized with Rosenthal's temperature distribution equation. Initializing the nodes considerably reduces the number of iterations required to define the temperature field.

Calculation of Temperatures

A Gauss-Seidel type iteration, which makes use of the

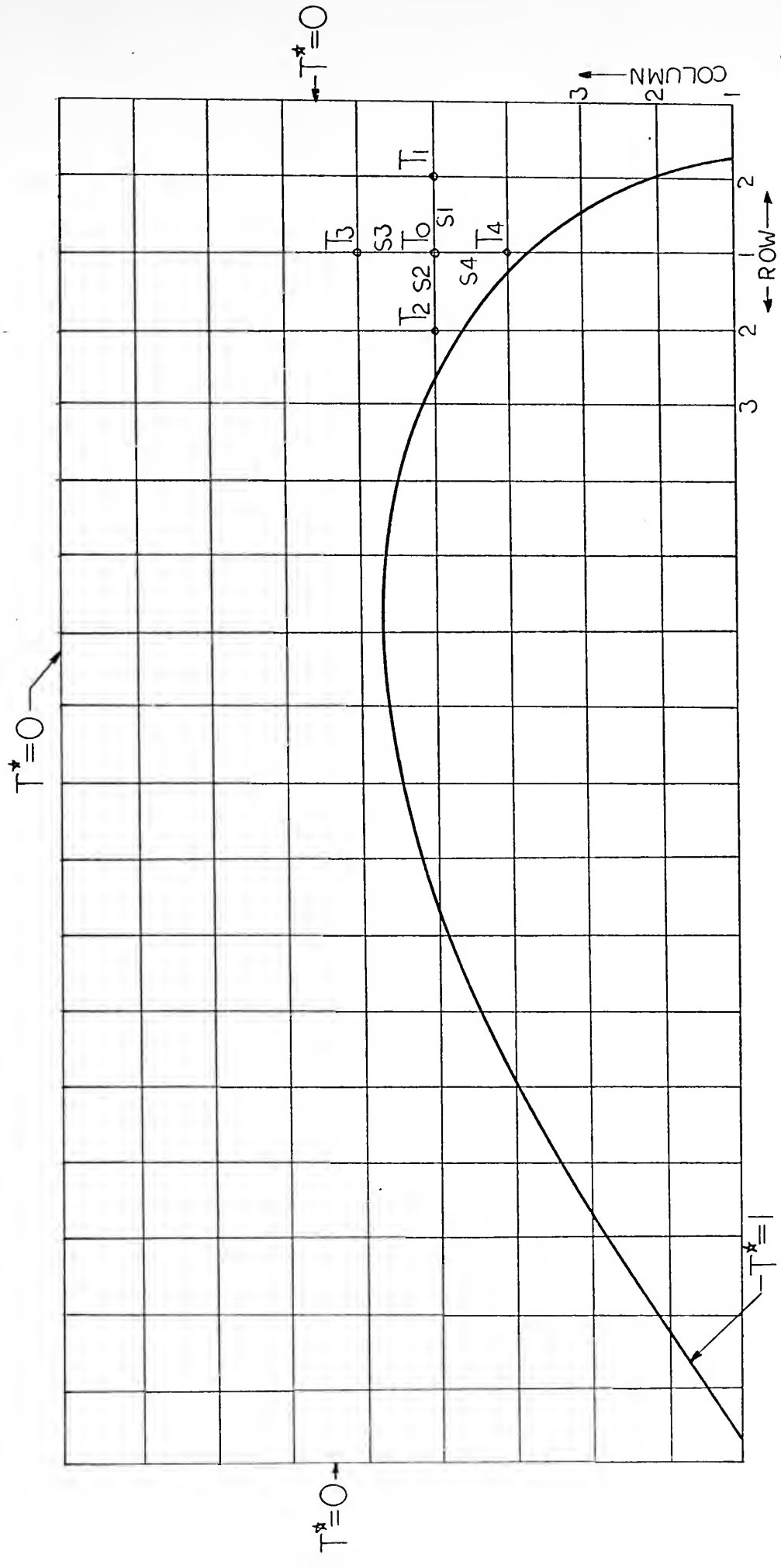


FIGURE 2-9

NOT TO SCALE

latest available temperatures, is performed until the specified accuracy is reached. The program begins each iteration cycle at the interior boundary to avoid excessive computation in that part of the field initially at zero normalized-temperature. Calculation of the value for the heat loss term is accomplished by a search routine which locates the node temperature on the boiling curve and computes the appropriate quantity.

CHAPTER III

UNDER WATER EXPERIMENT

The experimental verification portion of the thesis consisted of three steps: Production and measurement of molten pools made in air, production and measurement of molten pools made under water, and measurement of temperature changes in a specimen being welded under water.

A. EQUIPMENT

The same equipment shown schematically in Fig. 3-1, was used for all three steps mentioned above and consisted of the following:

1. Water Tank
2. Gravity Welding Stand
3. Power Source
4. Blow-Out System
5. Temperature Recorders
6. Current, Voltage, and Time Recorders.

Water Tank. The welding experiments were conducted in a steel tank measuring four feet long by three feet high by two feet deep. The exterior of the tank was sheathed in wood and rubber matting to prevent electrical shocks. A platform, containing clamps for securing the specimen, was fitted to the inside of the tank. The platform was located one foot from the top of the tank so that the electrode, but not the electrode holder, would be submerged in the water. The tank was



EQUIPMENT SCHEMATIC

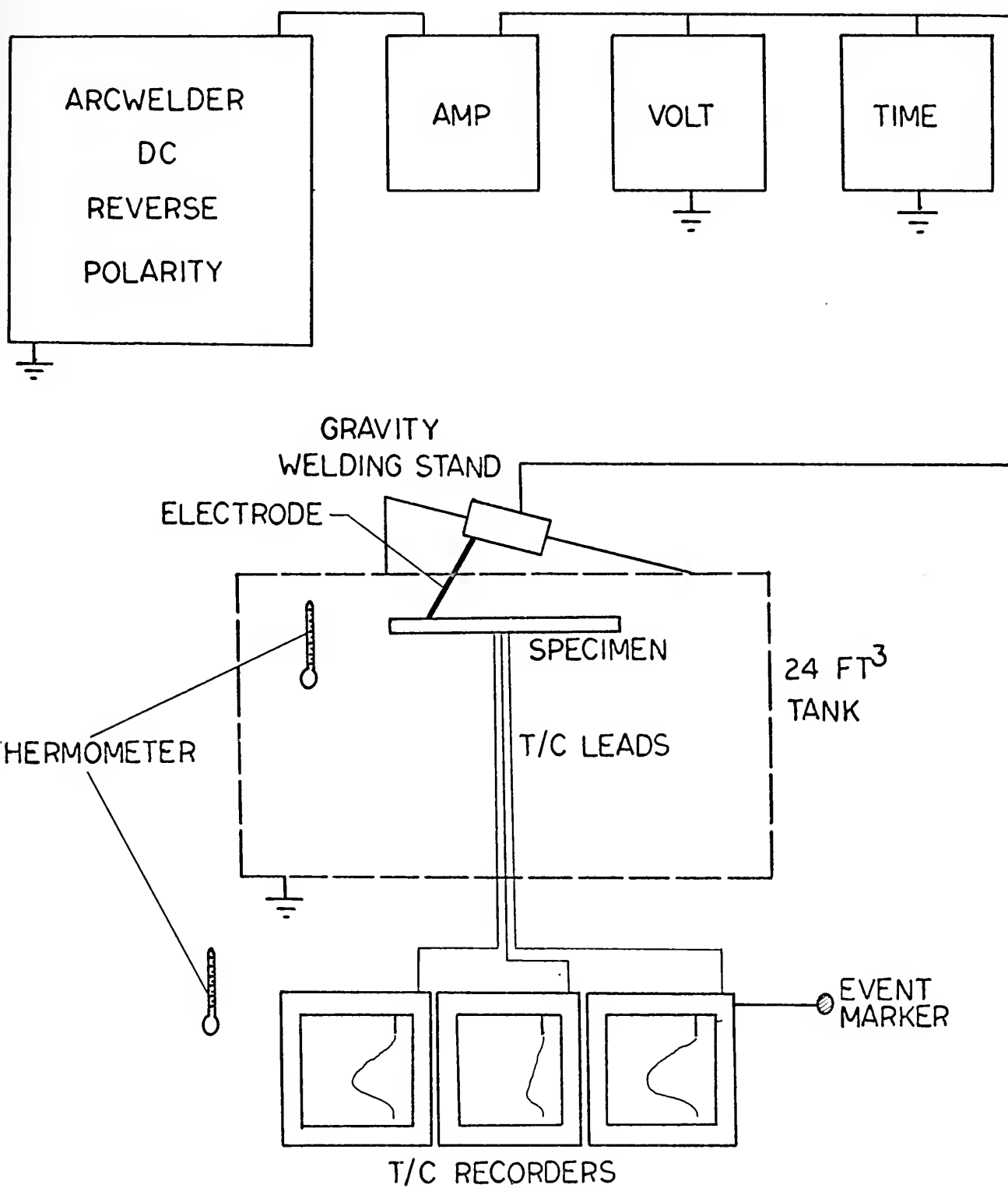


FIGURE 3-1



filled by a line supplying ordinary tap water (Boston, Mass.) and drained by gravity discharge. The tank was grounded to the power source.

Gravity Welding Stand. A Japanese gravity welder was modified and fitted to the top of the tank. The device consisted essentially of a common stick electrode holder fastened to a carriage which could be moved along an inclined rod. A guide was fitted to the rod so that the electrode would draw a straight bead on the specimen. The holder was adjusted to the carriage until the electrode maintained a constant angle of fifty-one degrees relative to the surface of the specimen in the plane of travel and ninety degrees in the plane perpendicular to direction of electrode travel. The tripping mechanism on the carriage, which rotated the electrode up and away from the specimen at the end of travel, was adjusted so that the arc would be broken after approximately six inches of weld bead was drawn.

Power Source. A Lincoln arc welder capable of supplying up to 40 volts and 350 amperes direct current was used as the power source. The device was set to operate on reverse polarity. This arrangement was selected as being similar to power sources for actual under water welding.

Pool Blowout System. A system for ejecting the molten metal from the specimen to form a weld crater was constructed and consisted of a nozzle, foot switch, and hose connected to a

400 psi air line. The nozzle was mounted on a swivel-necked holder so that adjustments could be made in directing the flow of air into the molten pool. The holder and nozzle combination were clamped to the welding platform inside the tank. Several preliminary experiments were conducted to determine the direction and angle of the jet. After considering several possible locations, it was decided that having the nozzle discharge along the axis of the bead but opposite to the direction of welding would give satisfactory results. The nozzle angle was adjusted so that the air would jet into the head of the pool, directing the molten metal back into the tank, away from the operator stationed behind the gravity welding stand. Once the proper relationships were determined, the nozzle was securely fastened in place for the duration of the experiment.

Temperature Recorders. Three clevis brush, Model 220, recording voltmeters were used to measure the output signal from the thermocouples. Each instrument contained two channels so that a total of six thermocouples could be recorded simultaneously. The event markers were connected electrically and repressing the key on any instrument would cause a simultaneous voltage signal to be recorded on all six channels. The instruments were calibrated at 200°, 700°, and 1000°F by inserting a thermocouple, of the same type and dimensions used in the experiment, into a controlled bath. The chart speed was adjusted to 25MM/sec and the gain to



1MV/Division (or, 50 MV total scale).

Common mercury-in-glass thermometers were selected to determine water and air temperatures. One thermometer was taped near the terminals of the recorders to measure the cold junction temperature; the other was suspended in the water to the depth of the specimen (approximately six inches below the surface at the far end of the tank).

Current, Voltage and Time Recorders. A General Electric D-C (0.400 amp.), Model 10H, recording ammeter was used to measure the current flowing in the welding circuit. An Esterline Angus (0.100V) recording voltmeter was selected to measure the operating voltage during welding. A timing device consisting of a General Time Co., Model 320-1, electric clock connected to a circuit breaker was used to measure the amount of time required to lay a given length of weld bead. When the arc was initiated the breaker would close and start the clock; breaking the arc would cause the clock to stop. Dividing the length of the bead by the elapsed time thus recorded served to indicate the welding speed.

All three instruments had been calibrated by the instrument lab at the Boston Naval Shipyard and were assumed to be recording properly.

B. MATERIALS

The criteria used in selecting materials for the

experiment was based upon their being readily available at low cost and having wide usage in the industry. Three main types of materials were used: Steel plate, welding rod, and thermocouple wire.

Steel Plate. All specimens were cut from cold rolled mild steel (A.I.S.I. 1010) sheets. Three different thicknesses were selected: $1/16"$, $1/8"$, and $1/4"$. For the pool measurement portion of the experiments the sheets were cut into 4" by 6" sections; the specimens in which temperatures would be measured were sized 6" by 12". Both sides of all specimens were cleaned by grit blasting to provide a uniform surface.

Welding Rod. E7018-A1, $3/16"$ Dia., stick electrodes were found to be suitable for the experiment. This metal core of rod tended to melt faster than the flux covering disintegrated permitting the electrode to be "dragged" across the specimen during the under water operation. This feature was critical since the arc would not be observed while under water. In essence, the flux covering formed a hollow cylinder which shielded the arc from the water, and maintained the proper length necessary to sustain the arc. Since the duration of submergence in water was short (20 to 30 sec!), the electrodes were not coated with any water-proofing materials.

Thermocouples. Chromel-Alumel Type K, $.020"$ Dia., thermocouples were selected to measure temperature changes occurring

during welding. The selection of the size required a compromise between the ability to withstand high temperatures (large diameters), and fast response (small diameters). Furthermore, the thermocouple junction should form a point (ideally) in the specimen to avoid thermal gradients. Consequently, the diameter chosen was the smallest considered practical for the experiment.

C. EXPERIMENTAL PROCEDURE

The experiment consisted of three parts: a) production of pool craters, b) temperature measurement, and c) reduction of data.

Production of Pool Craters. Molten pools were first produced by welding in air, then under water. The essential steps for this part of the experiment were the same for both processes:

1. Specimen aligned on work table.
 - 1a. If an under water pool was desired the tank was then flooded until the specimen was covered by six inches of water.
2. The power source was turned on and given a five minute warm up period.
3. An electrode was selected and inserted into the holder.
4. The holder and carriage on the welding stand were manipulated so that the tip of electrode was slowly brought to the point where the arc would start.

5. The carriage was drawn manually down the incline of the welding stand, laying a bead on the surface of the specimen.
6. The blow-out system was activated by depressing the foot switch at the same time the carriage reached the tripping point.
7. The voltage, current, time, ambient temperature, and bead length were recorded for the run.
8. The dimensions of the pool crater - maximum width length, and contour - were measured under a Bausch and Lomb microscope equipped with a movable, micro-meter table. For the experiment, the ranges of the variables involved were:

Current	100 - 300 amps
Voltage	20 - 30 volts
Welding Speed	50 - 125 Ft/Hr
Ambient Temp	70 - 73°F (air)
	40 - 43°F (water)

Current was varied by adjusting the rheostat on the power source in steps of 25 amps., while the voltage was allowed to seek its equilibrium value. A particular welding speed could not be chosen prior to a run since manual control was required. However, an acceptable range of speeds were produced by consciously moving the carriage along the guide rod. A mechanical system for better control of welding

speeds was attempted but the problem of starting the arc defeated the project.

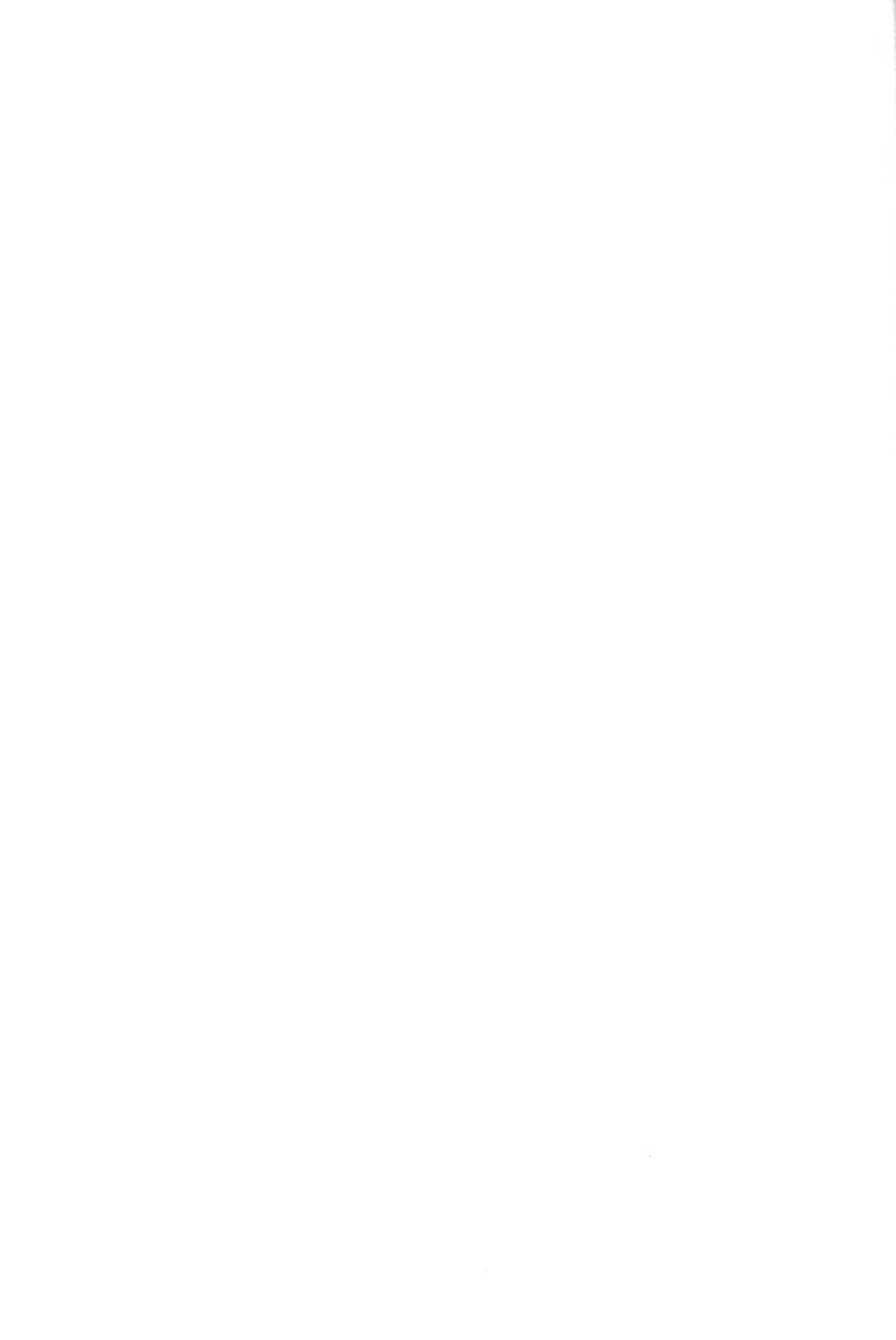
The problems connected with producing two-dimensional weld pools by stick electrodes were numerous and it is estimated that only 20% of the runs resulted in acceptable data. The welding stand did not allow the freedom of movement that is normally enjoyed in "Striking" an arc. The low voltages required so small a gap that the electrode was often grounded before the arc could start. In addition, a precise relationship between the heat input and welding speed necessary to produce two-dimensional pools was observed. Since the speed could not be predetermined, there was no assurance that an acceptable pool would result from a run. In practice, a number of runs at various speeds were conducted at a given heat input until the proper pool was obtained.

Temperature Measurement. All runs were conducted under water and the essential steps involved were:

1. Preparation of the specimen.
2. Aligning the specimen with the electrode.
3. Flooding the tank until the level of water was six inches above the surface of the specimen.
4. Turning on the power source and recording equipment.
5. Welding a bead on the surface of the specimen.
6. Recording temperature, time, voltage, current and bead length for the run.

The preparation of the specimen involved drilling holes at specified locations and mounting the thermocouples. Twelve .040" Dia. holes were drilled in each plate in a manner that assured that the axis of the hole was perpendicular to the surface of the plate. (See Fig. 3.2 for relative locations.) The centers of the holes formed a locus of points five degrees from the center line of the plate. An additional hole (#12) was drilled near the edge of the specimen to record boundary temperatures. (During the experiment, however, it was observed that placing a thermocouple at this location -- about six inches from the center line -- was unnecessary since the temperature variation one-half inch from the center line was negligible!). The distance from the center line to the center of the holes varied from 0.125" for thermocouple locations #1 and #9 to 0.440" for location #8; the center-to-center spacing along the center line of the specimen was approximately one-half inch. This arrangement permitted temperature measurement within the heat affected zone but allowed sufficient spacing for mounting thermocouples.

The thermocouples required special preparation to prevent their shorting out in the water. As shown in Fig. 3-3, water-proofing required sealing all joints with expoxy cement and covering the leads with a plastic sleeve shrunk to form a tight fit. (In their commercial form, the thermocouples had already been covered with asbestos and silicon



THERMOCOUPLE ARRANGEMENT

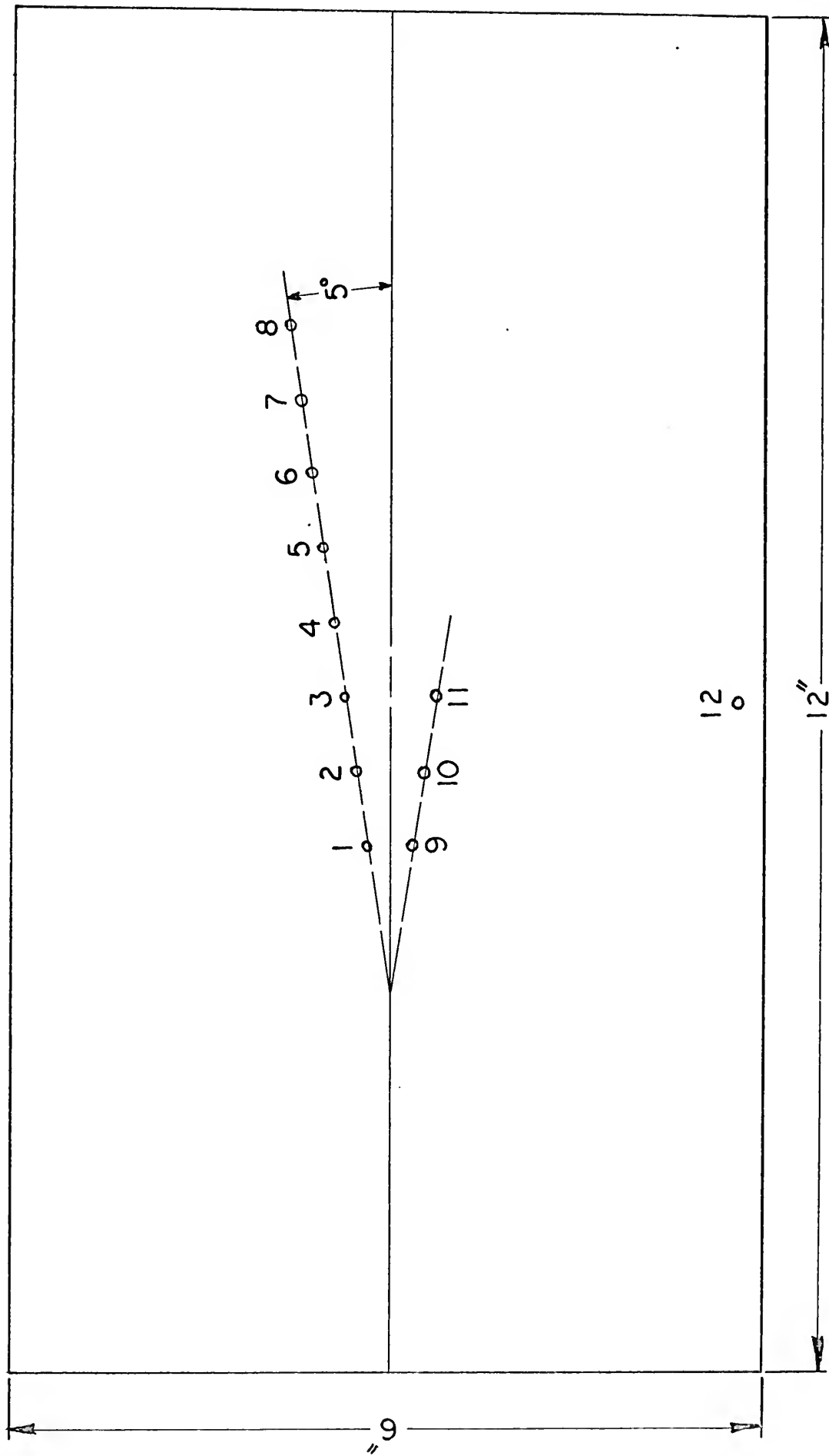


FIGURE 3-2

THERMOCOUPLE INSTALLATION

TOP SURFACE OF SPECIMEN

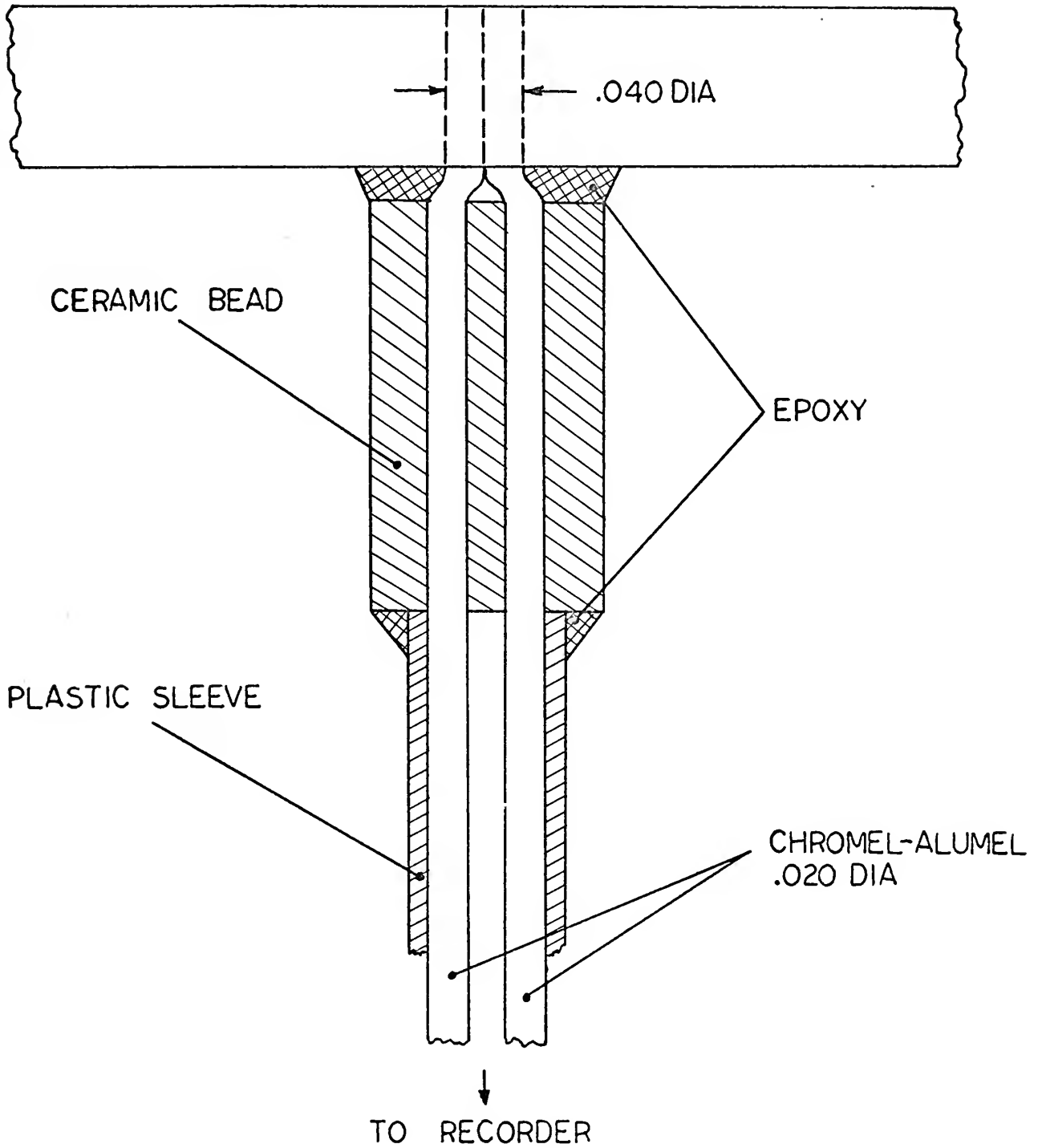


FIGURE 3-3



impregnated fiberglass.) A ceramic bead, containing separate holes for each thermocouple, was installed to prevent the leads from making a junction outside the plate. The exposed leads were cleaned with emery cloth and inserted through one of the holes in the specimen. A pair of pliers were then used to draw the leads through the hole until the ceramic bead was nearly flush with the bottom of the plate. The edges of the hole on the top surface of the specimen were given a light peening with a small prick punch, and the leads extending above the specimen were then cut off and ground flush with the surface. The purpose of the peening was to give the thermocouples a preferential junction at the surface of the specimen, and avoid, as much as possible, thermal gradients in the specimen resulting from pools that were not exactly two-dimensional. In this way, six thermocouples were installed in each specimen, leaving six open holes. Twelve specimens were prepared such that no specimen duplicated another with respect to thermocouple locations.

All specimens, with one exception, were $1/8$ " thick: the $1/16$ " plates burned through too easily and the penetration required for two-dimensional pools was not obtained with the $1/4$ " plates. One $1/4$ " specimen was prepared, however, in order to determine the response of the model to a situation where the shape of the heat source was not the ideal configuration.

A prepared specimen, containing six thermocouples,

was then placed on the table in the tank, aligned with the electrode, and clamped firmly in position. (It was first hoped that the electrode could be made to follow exactly the center line of the plate, thereby simplifying measurements taken to locate the thermocouples. Actual runs revealed, however, that although the bead would run straight it could not be made to follow any predetermined path with sufficient accuracy. Consequently, the positions of the thermocouples with respect to the longitudinal axis of the weld bead were re-calculated following the run.) The tank was filled with water until the specimens were submerged to a depth of six inches. The thermocouples were connected to the recorders and the power source turned on. The electrode was guided into position in preparation for starting the arc. At the instant of arc initiation the laboratory assistant depressed the event key on a recorder which placed marks on all six output tapes simultaneously. The carriage containing the electrode was then either allowed to move freely or was drawn by hand with a slight steady pressure down the incline of the welding stand until the tripping point was reached. The ambient temperatures (air and water), run time, current, voltage, and bead length were recorded.

D. DATA REDUCTION

The maximum dimensions of the weld pool were correlated to the lumped parameter, p , by means of a least squares curve

fitting routine available as an option from M.I.T.'s computer library. The details of the program have been presented in Appendix 7. The correlating parameters for the equation determining the contour of the weld pool were found by a non-linear least squares curve fitting routine. (Also see Appendix 7.)

The Y-displacement of a thermocouple was measured along a line running through the center of the thermocouple hole and perpendicular to the center line of the weld bead. The X-displacement was measured along the center line from the point of arc initiation to the intersection of the perpendicular. A computer program was written to reduce the thermocouple output. The program scaled the distances on the output tape by the ratio of the chart to the welding speed, and ensured that the measured temperatures were in the same coordinate system as the model. The program, with sample input data, appears in Appendix 7.

CHAPTER IV

RESULTS AND CONCLUSIONS

A. TEST RESULTS

The experimental portion of the thesis consisted of two parts: a) determining the correlation constants used to define the shape of the heat source, and b) measuring the temperature distribution in the specimen during welding. The results of the pool contour study have already been presented and described in Chapter II; it need only be added here that the results were in good agreement with theoretical predictions.

The results of comparing the temperature distribution as predicted by the model with that measured by experiment is represented in Figures 4-1 and 4-2. The coordinates have been presented in dimensionless form. The ordinates may be transformed to real temperatures by the relation

$$T = 2677T^* + 503 \quad , \quad ^\circ R$$

The abscissa for Figure 4-1 can be converted to inches with the relation

$$x = x^*/40.59 \quad , \quad \text{in.}$$

and for Figure 4-2

$$x = x^*/41.02 \quad , \quad \text{in.}$$

where the starred terms are the dimensionless values.

In the figures, Y_m is the distance from the weld center



FIG 4-1: CALCULATED AND MEASURED TEMPERATURE CURVES

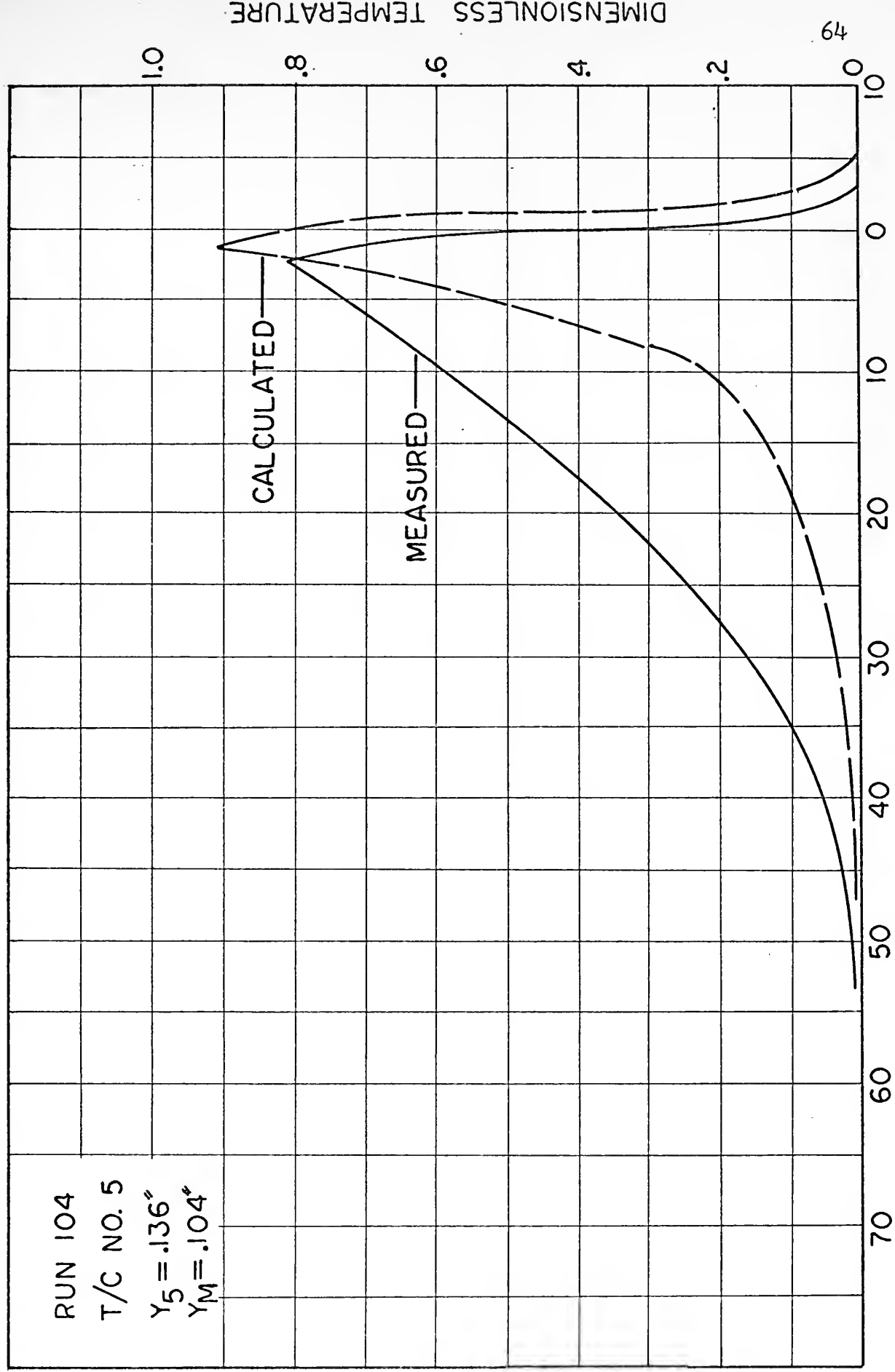
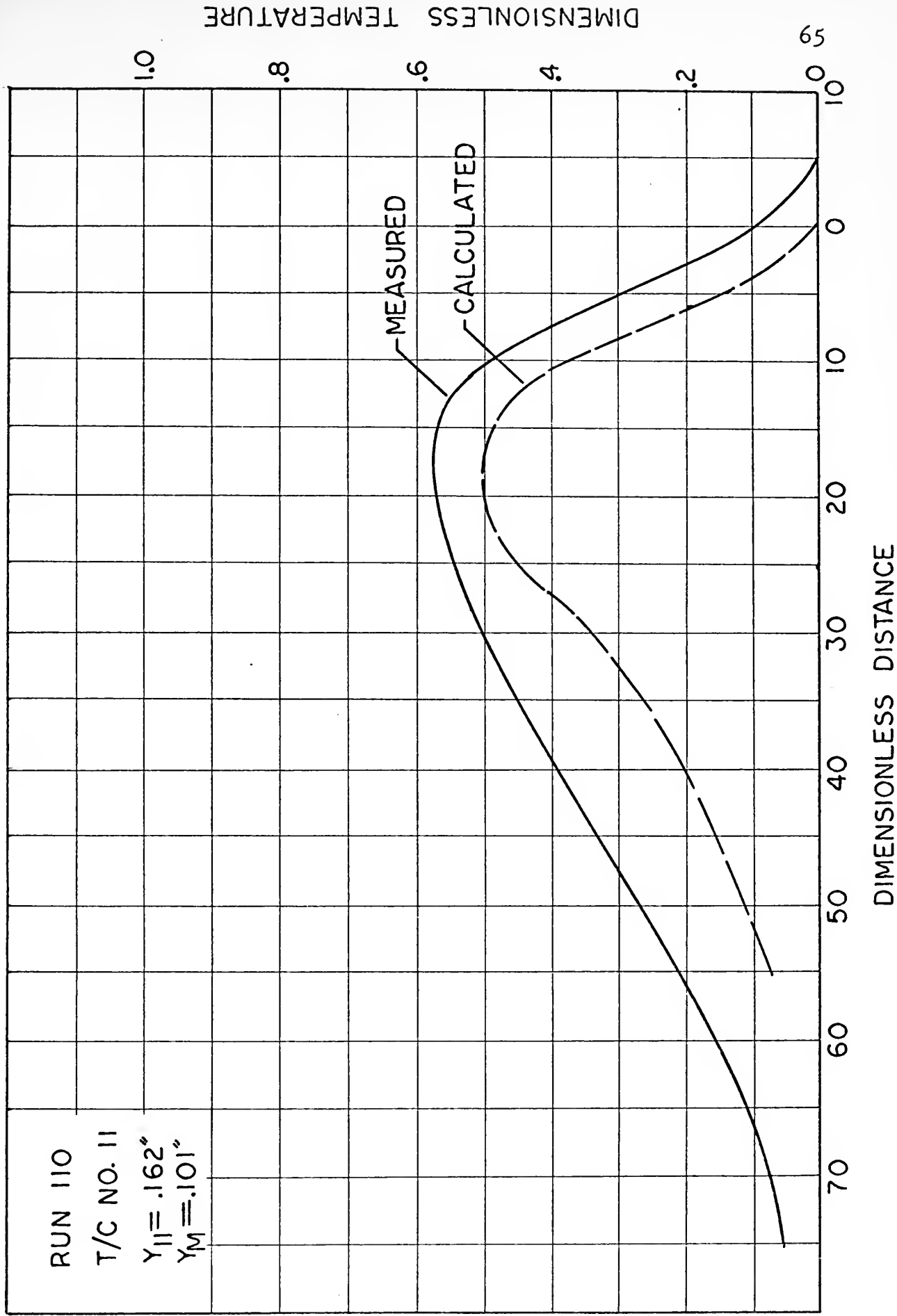




FIG 4-2: CALCULATED AND MEASURED TEMPERATURE CURVES





line to the point of maximum pool width; Y_5 (Figure 4-1) and Y_{11} (Figure 4-2) are the locations in the specimen at which the temperatures were measured.

Figure 4-3 has been included so that a comparison might be made between the predicted temperature distribution for underwater and air welding: the curve labeled "without boiling" represents air welding.

Figure 4-4 presents the predicted temperature distribution at various distances from the center line.

Figure 4-5 represents a comparison between measured and calculated cooling curve slopes as a function of distance from the maximum pool width. The slope of a given curve was measured at the average temperature and was defined to be the change in dimensionless temperature with respect to the change in dimensionless distance. For example, the average measured temperature for Run 104 (Figure 4-1) was found to be 0.4 and the slope at that point was -0.025.

B. ANALYSIS OF DATA

Prediction of peak temperatures within the haz was found to be reasonably accurate. The average error was 13% of the measured temperature with the largest variation occurring at locations nearest the molten pool.

The cooling portions of the predicted and measured curves were not found to be in good agreement. As a quantitative



FIG 4-3: TEMPERATURE COMPARISON CURVES

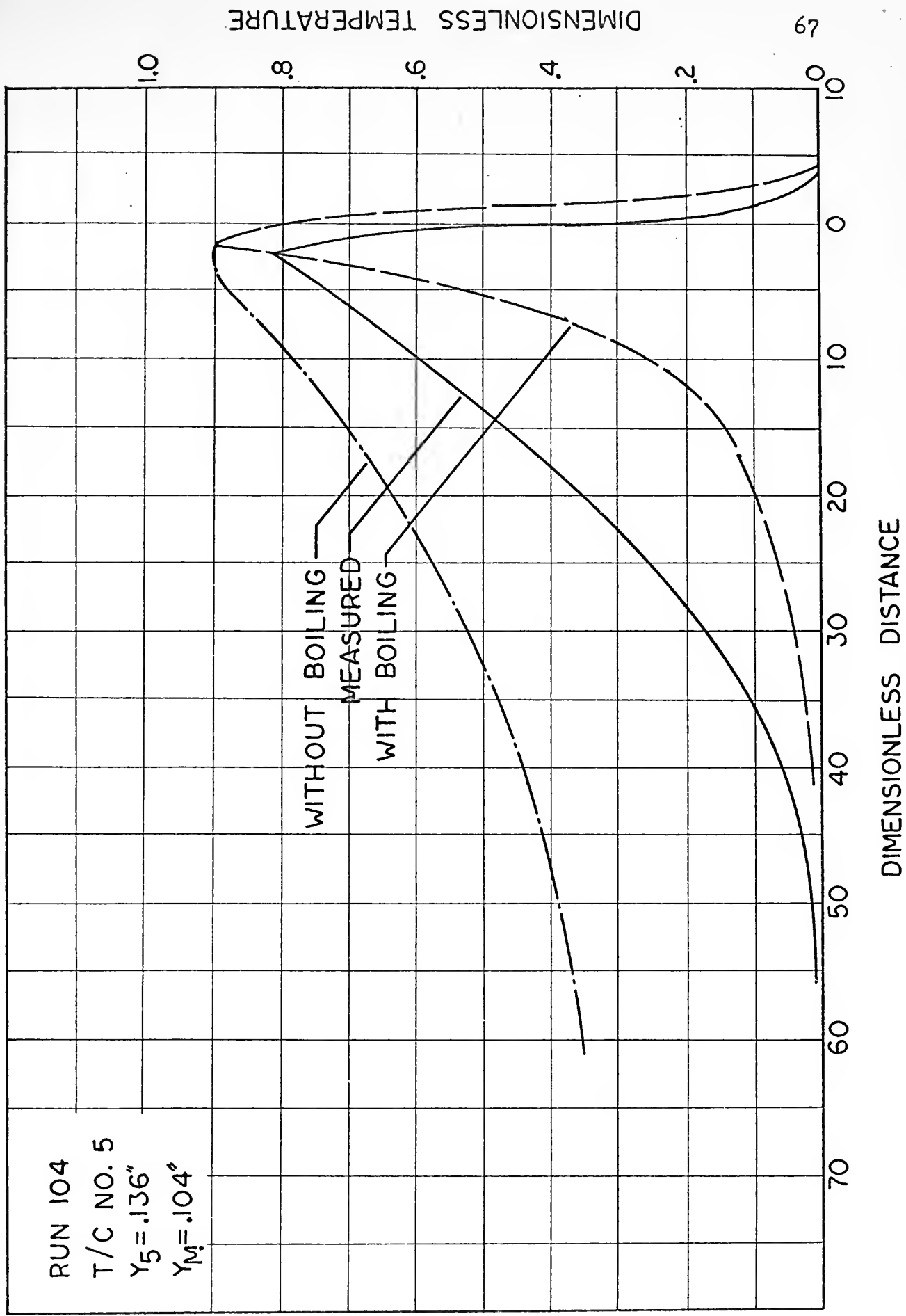




FIG 4-4: THEORETICAL TEMPERATURE DISTRIBUTION

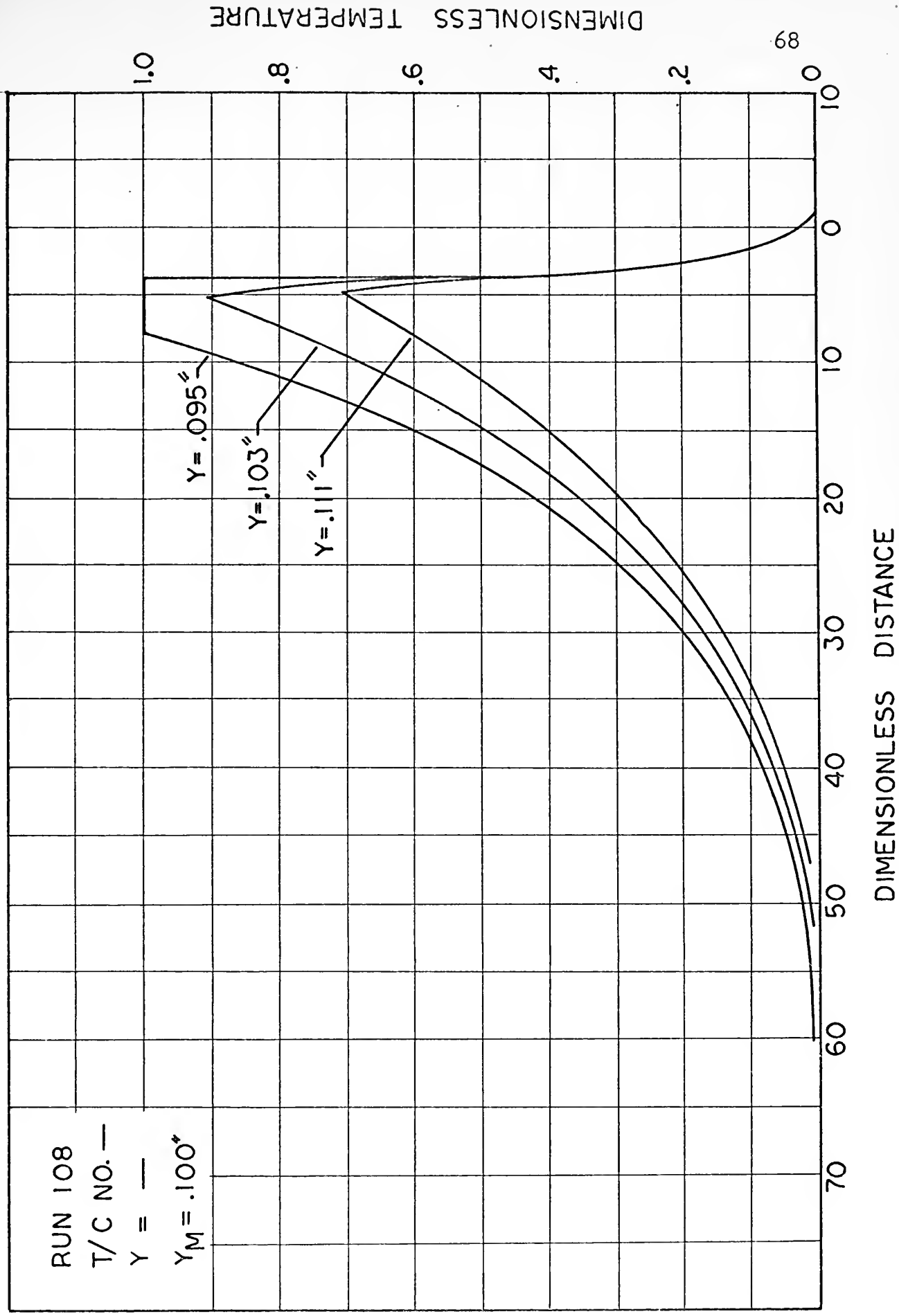
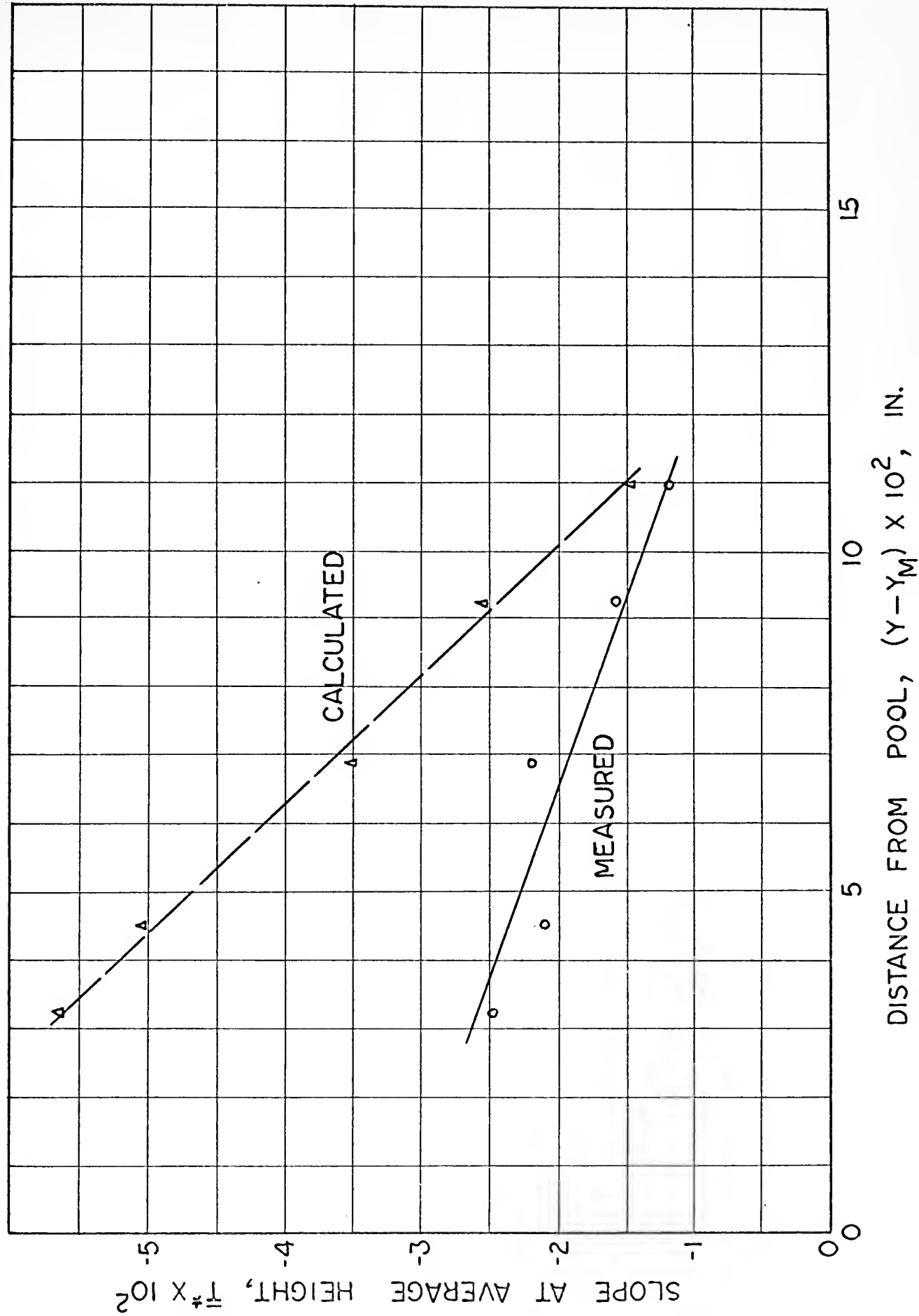


FIG 4-5: COMPARISON OF COOLING CURVE SLOPES





indication slopes were measured at the average temperatures of the distribution curves and plotted as in Figure 4-5. Near the molten pool the predicted slopes were in error by as much as 200% and converged with the measured values only at the edges of the haz.

Consequently, an evaluation of the effect of boiling heat transfer was made by neglecting such losses (i.e., as if the specimen were welded in air) with results presented in Figure 4-3. Such an analysis indicated that conventional boiling models were inappropriate when applied to the underwater welding process. However, nothing in the data suggested what mechanism might be controlling heat losses.

It can be speculated, nevertheless, that the gases generated from the oxidizing flux and possible dissociation of water molecules could disrupt the normal process of boiling heat transfer. These gases would collect at the tip of the electrode to form a protective envelope over a portion of the specimen's top surface. Within the envelope the predominant form of heat transfer would be radiation while exterior areas could support some boiling losses. The bottom surface (opposite to the side being welded) would not see the shielding gases and, consequently, heat losses would occur through boiling. This mixed form of heat losses would explain the slower rate of cooling observed during the experiment.

Unfortunately, the process just described has not been described in the literature and the development of the

appropriate model was beyond the scope of this research study.

C. CONCLUSIONS

The object of this thesis was to develop a model that would predict the temperature distribution in thin plates welded underwater.

It was found that the model was reasonably accurate in predicting peak temperatures within the heat-affected-zone, but that the degree of error in predicting cooling rates was unacceptably large. The initial assumption that heat losses can occur by conventional mechanisms of boiling heat transfer was, therefore, considered invalid. Rather, the proper model for heat losses must consider the effect of the gases generated during the welding operation.

Given the above conclusions, it is recommended that

1. Heat transfer fluxes be correlated for the case involving full boiling with radiation on one surface while the opposite surface supports only radiation, and
2. The study should consist of vertical as well as horizontal surfaces.



APPENDIX 1

TRANSFORMATION TO QUASI-STEADY STATE

$$K \left[\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} \right] + \frac{\partial k}{\partial x} \frac{\partial T}{\partial x} + \frac{\partial k}{\partial y} \frac{\partial T}{\partial y} + W_i = \rho c \frac{\partial T}{\partial t} \quad (1)$$

in the steady state system, the origin will be defined to be centered at the moving arc. Hence, the following transformation-equations apply:

$$x = \bar{x} + v_t$$

$$t = \bar{t}$$

where the primed quantities refer to the new system.

Hence,

$$\frac{\partial \bar{x}}{\partial x} = 1, \quad \frac{\partial \bar{x}}{\partial t} = -v, \quad \frac{\partial \bar{t}}{\partial t} = 1, \quad \frac{\partial \bar{t}}{\partial x} = 0$$

with these partial relationships, derivatives in the new system become,

$$\frac{\partial T}{\partial x} = \frac{\partial T}{\partial \bar{x}} \frac{\partial \bar{x}}{\partial x} + \frac{\partial T}{\partial \bar{t}} \frac{\partial \bar{t}}{\partial x} = \frac{\partial T}{\partial \bar{x}}$$

$$\frac{\partial^2 T}{\partial x^2} = \frac{\partial^2 T}{\partial \bar{x}^2}$$

$$\frac{\partial T}{\partial t} = \frac{\partial \bar{x}}{\partial t} \frac{\partial T}{\partial \bar{x}} + \frac{\partial T}{\partial \bar{t}} \frac{\partial \bar{t}}{\partial t} = -v \frac{\partial T}{\partial \bar{x}} + \frac{\partial T}{\partial \bar{t}}$$

Since temperature is constant in the steady state system

$$\frac{\partial T}{\partial \bar{t}} = 0$$

and

$$\frac{\partial T}{\partial t} = -v \frac{\partial T}{\partial \bar{x}}$$

Substituting into equation (1) and dropping prime notation,

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{1}{k} \left[\frac{\partial k}{\partial x} \frac{\partial T}{\partial x} + \frac{\partial k}{\partial y} \frac{\partial T}{\partial y} \right] + \frac{W_i}{k} = -\frac{v}{\alpha} \frac{\partial T}{\partial x}$$

where

$$\alpha = k/\rho c$$

The equation is simplified by noticing that

$$\frac{1}{k} \frac{\partial k}{\partial x} \frac{\partial T}{\partial x} = \frac{1}{k} \frac{\partial k}{\partial T} \left(\frac{\partial T^2}{\partial x} \right)$$

Finally,

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{1}{k} \frac{\partial k}{\partial T} \left[\left(\frac{\partial T^2}{\partial x} \right) + \left(\frac{\partial T^2}{\partial y} \right) \right] + \frac{W_i}{k} = -\frac{v}{\alpha} \frac{\partial T}{\partial x}$$



APPENDIX 2DERIVATION OF HEAT LOSS TERMS

A. Nucleate Boiling

The Rohsenow¹⁵ correlation is

$$\frac{C_L (T_w - T_{SAT})}{h_{fg}} = C_{Sf} \left[\frac{(g/A)_N}{u_L h_{fg}} \sqrt{\frac{g_O \sigma}{g(\rho_L - \rho_V)}} \right]^{\frac{1}{3}} Pr_L \quad 1.7$$

All quantities are taken at saturation (14.7 psia, 212°F) from the CRC HANDBOOK OF TABLES FOR APPLIED ENGINEERING SCIENCE.

$$\begin{aligned} C_L &= 1.005 \text{ BTU/lb-}^{\circ}\text{R} \\ Pr_L &= 1.9 \\ u_L &= 62.5 \times 10^{-7} \text{ lb-sec/ft}^2 \\ g_O/g &= 1 \\ h_{fg} &= 970.3 \text{ BTU/lb} \\ \rho_L &= 59.8 \text{ lb/ft}^3 \\ \rho_V &= 0.0374 \text{ lb/ft}^3 \\ \sigma &= 40.4 \times 10^{-4} \text{ lb/ft} \end{aligned}$$

Substitution of these values into the nucleate boiling equation yields,

$$q_N = [1.542(T_w - T_{SAT})]^3, \quad \text{BTU/hr-ft}^2.$$

in normalized form,

$$q_N^* = [1.542T^* (T_M - T_{SAT})]^3$$

where,



$$T^* = \frac{T_w - T_{SAT}}{T_M - T_{SAT}} .$$

B. Film Boiling

The Borishansky relation reported by Jordan²⁵ is

$$q_F = 0.425 \left[\frac{K_V^3 g \rho_V (\rho_L - \rho_V) (h_V - h_L)}{u_V [\nabla / g (\rho_L - \rho_V)]^{\frac{1}{2}}} \right]^{\frac{1}{4}} (T_w - T_{SAT})^{0.75}$$

where,

$$\rho_L = 59.8 \text{ lb/ft}^3$$

$$\rho_V = 0.0374 \text{ lb/ft}^3$$

$$\sigma = 40.4 \times 10^{-4} \text{ lb/ft}$$

$$h_{fg} = 970.3 \text{ BTU/lb}$$

$$u_V = 2.5 \times 10^{-7} \text{ lb-sec/ft}^2$$

$$K_V = 14.3 \times 10^{-3} \text{ BTU/hr-ft-}^{\circ}\text{R}$$

Substitution of these values into the film boiling equation yields,

$$q_F = 32.4 (T_w - T_{SAT})^{0.75}, \text{ BTU/hr-ft}^2 .$$

where the normalized form becomes,

$$q_F^* = 32.4 [T^* (T_M - T_{SAT})]^{0.75}$$

C. Transition Regime

The proposed relation for heat flux in this regime is

$$q_T = M (T_w - T_{SAT})^b$$

to which the boundary conditions are applied:

- 1) $q_T = q_N$ at $T_w = 300^{\circ}\text{F}$
- 2) $q_T = q_F$ at $T_w = 500^{\circ}\text{F}$



then,

$$1a) \quad M(300-212)^b = [1.542(300-212)]^3 = 2.34 \times 10^6$$

$$2a) \quad M(500-212)^b = 32.4(500-212)^{0.75} = 2265.11$$

for which

$$m = 2265.11$$

$$b = 5.86$$

Now, the transition heat flux equation becomes,

$$q_T = 2265.11 \left(\frac{288}{T_w - T_{SAT}} \right)^{5.86}$$

and the normalized relation,

$$q_T^* = 2265.11 \left[\frac{288}{T^*(T_M - T_{SAT})} \right]^{5.86}$$

D. Radiation Heat Transfer

For the heat flux during radiation,

$$q_R = \sigma F_e F_a (T_w^4 - T_{SAT}^4).$$

Since the specimen is completely enclosed by the liquid environment,

$$F_a = 1$$

and

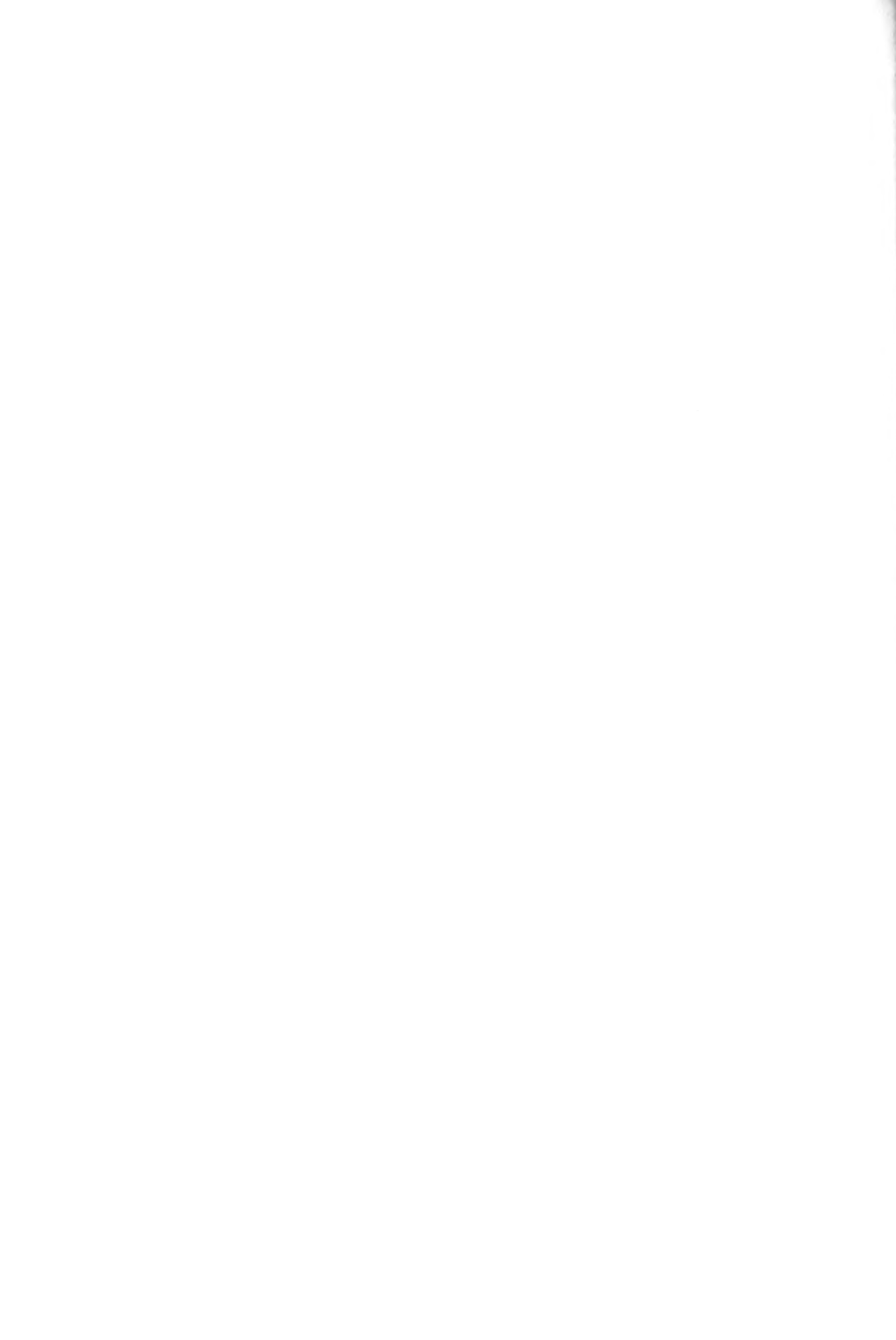
$$F_e = 1/(1/e_1 + 1/e_2 - 1)$$

the emissivity of steel cleaned by sandblasting is approximately,

$$e_1 = 0.8$$

while the water is very nearly a black body and

$$e_2 = 0.95 \quad .$$



The heat flux equation becomes,

$$q_R = 0.1317 \times 10^{-8} (T_W^4 - T_{SAT}^4)$$

and in normalized form,

$$q_R^* = 0.1317 \times 10^{-8} \left\{ \left[T^* (T_W - T_{SAT}) + T_{SAT} \right]^4 - T_{SAT}^4 \right\}$$



APPENDIX 3Derivation of Pool Boundary Equations

Pavelic's derivation of the pool boundary equation can be summarized as follows:

1. Rosenthal's point source equation is analyzed to obtain the maximum theoretical pool dimensions.
2. A correlation of measured data is obtained based on the predicted relations.
3. A two-dimensional equation is fitted through the boundaries of experimentally obtained pools.

1. The maximum dimensions of the pool.

Rosenthal's equation for the temperature distribution from a point source,

$$T - T_{\infty} = \frac{Q}{2\pi L\bar{K}} K_0(\bar{\lambda}Vr) \exp[-\bar{\lambda}Vx]$$

where

$$Q = 3.415 EI \quad , \text{ BTU/HR}$$

$$K_0(\bar{\lambda}Vr), \text{ Bessel Function}$$

$$\bar{\lambda}, \text{ Reciprocal of the average diffusivity}$$

$$V, \text{ Welding Speed (Ft/Hr)}$$

$$L, \text{ Plate Thickness (Ft).}$$

With the normalized relations,

$$\bar{\lambda}Vx = x^*$$

$$T^* = \frac{T - T_{\infty}}{T_m - T_{\infty}}$$

The temperature equation becomes

$$T^* = \frac{3.415 EI}{2\pi\bar{\alpha}L H_m} K_O(r^*) \exp[-x^*]$$

Define the lumped welding parameter,

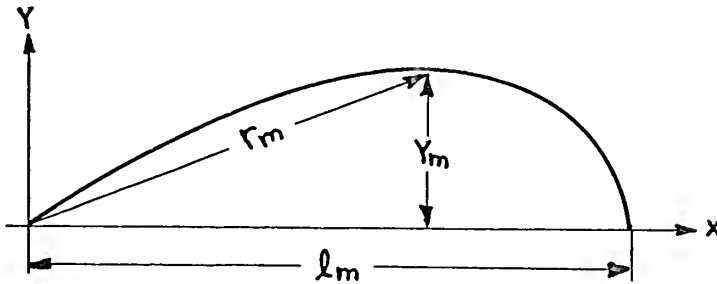
$$p^* = \frac{3.415 EI}{2\pi\bar{\alpha} L H_m}$$

where,

$$H_m = \bar{\rho} \bar{e} (T_M - T_\infty).$$

so that (with star notation dropped),

$$T = pK_O(r) \exp[-x]$$



The maximum dimensions are found since,

$$\frac{\partial T}{\partial x} = 0 \quad \text{at maximum width} \quad 1.$$

$$r = x \text{ and } y = 0 \text{ at maximum length.} \quad 2.$$

From condition 1., the value for the maximum width is

$$Y_m = r_m \left[1 - \left[\frac{K_O(r_m)}{K_1(r_m)} \right]^2 \right]^{\frac{1}{2}}$$

and from condition 2.,

$$l_m = 2r_m \frac{K_O(r_m)}{K_1(r_m)}$$



where r_m is defined at $T = 1$ and

$$P = \frac{\text{Exp}[-r_m K_O(r_m)/K_1(r_m)]}{K_O(r_m)}$$

2. Pool Contour Equation:

An equation is arbitrarily selected to match the boundary conditions for the pool,

$$\dot{y} = \infty \quad , \quad \text{at} \quad x = 0, \ell_m$$

$$\dot{y} = 0 \quad , \quad \text{at} \quad y = y_m$$

$$y = 0 \quad , \quad \text{at} \quad x = 0, \ell_m$$

A candidate is then,

$$y^2 = B X^m (\ell_m - x)^n$$

From the boundary condition, $\dot{y} = 0$ at $y = y_m$,

$$B = \frac{y_m^2}{\left(\frac{m \ell_m}{n+m}\right)^m \left(\frac{n \ell_m}{n+m}\right)^n}$$

where "m" and "n" are parameters to be determined experimentally.



APPENDIX 4

NORMALIZING THE HEAT FLOW EQUATION

The objective is to nondimensionalize all quantities. Consequently, the dimensionless temperature is defined to be,

$$T^* = \frac{T - T_o}{T_m - T_o}$$

where,

T_o = Ambient temperature

T_m = Melting temperature of the metal

and,

$T^* = 1$, at the weld pool

$T^* = 0$, at the specimen edge

the coordinate dimensions are transformed by a length scale, H, so that

$$x^* = x/H$$

$$y^* = y/H$$

By making the appropriate substitutions, we convert the equation of heat flow

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{1}{k} \frac{\partial k}{\partial T} \left[\left(\frac{\partial T}{\partial x} \right)^2 + \left(\frac{\partial T}{\partial y} \right)^2 \right] + \frac{1}{KL} (-\dot{q}_R - \dot{q}_B) = -\frac{v}{\alpha} \frac{\partial T}{\partial x}$$

to non-dimensional form,

$$\frac{\partial^2 T}{x^2} + \frac{\partial^2 T}{y^2} + \frac{1}{k} \frac{dk}{dT} \left[\left(\frac{\partial T}{x} \right)^2 + \left(\frac{\partial T}{y} \right)^2 \right] + \frac{H^2}{LK} [-q_R - q_B] = -\frac{VH}{\alpha} \frac{\partial T}{\partial x}$$

with the star notation dropped.

A value for H can be obtained by examining the



convergence criterion* for the related linear equation*,

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{H^2}{K} (-\dot{q}_R - \dot{q}_B) + \frac{VH}{\alpha} \frac{\partial T}{\partial x} = 0$$

Ames² suggests that the solution will converge when

$$0 < S < 2\alpha/VH$$

where the value, S , is the grid spacing for the numerical solution. We arbitrarily establish the upper bound on S to be one so that

$$H = 2\alpha/v$$

Since a small grid spacing is desirable for accuracy, we allow the diffusivity, α , to assume its least value.

Hence,

$$H = 2\alpha_{\text{Min}}/v$$

Finally, the complete non-dimensional equation of heat flow becomes:

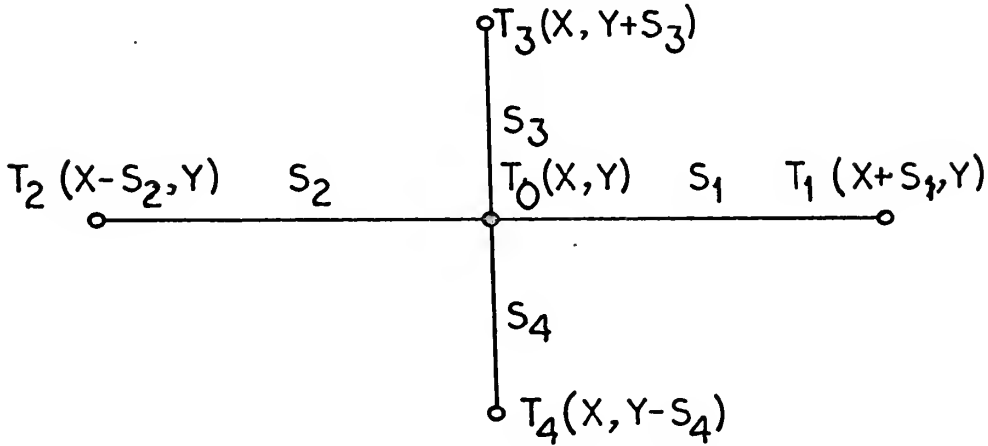
$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{1}{k} \frac{dK}{dT} \left[\left(\frac{\partial T}{\partial x} \right)^2 + \left(\frac{\partial T}{\partial y} \right)^2 \right] + \frac{4\alpha_{\text{Min}}^2}{LV^2K} (-\dot{q}_R - \dot{q}_B) = -\frac{2\alpha_{\text{Min}}}{\alpha} \frac{\partial T}{\partial x}$$



APPENDIX 5

FINITE DIFFERENCE EQUATION OF HEAT FLOW

Consider the following grid node:



The S_i represent grid spacings between nodes and T_0 will be a weighted "average" of the values at the four surrounding nodes. The following treatment is a summary of the approach employed by Tanbachuchi²⁰ and presented in detail by Greenspan.⁶

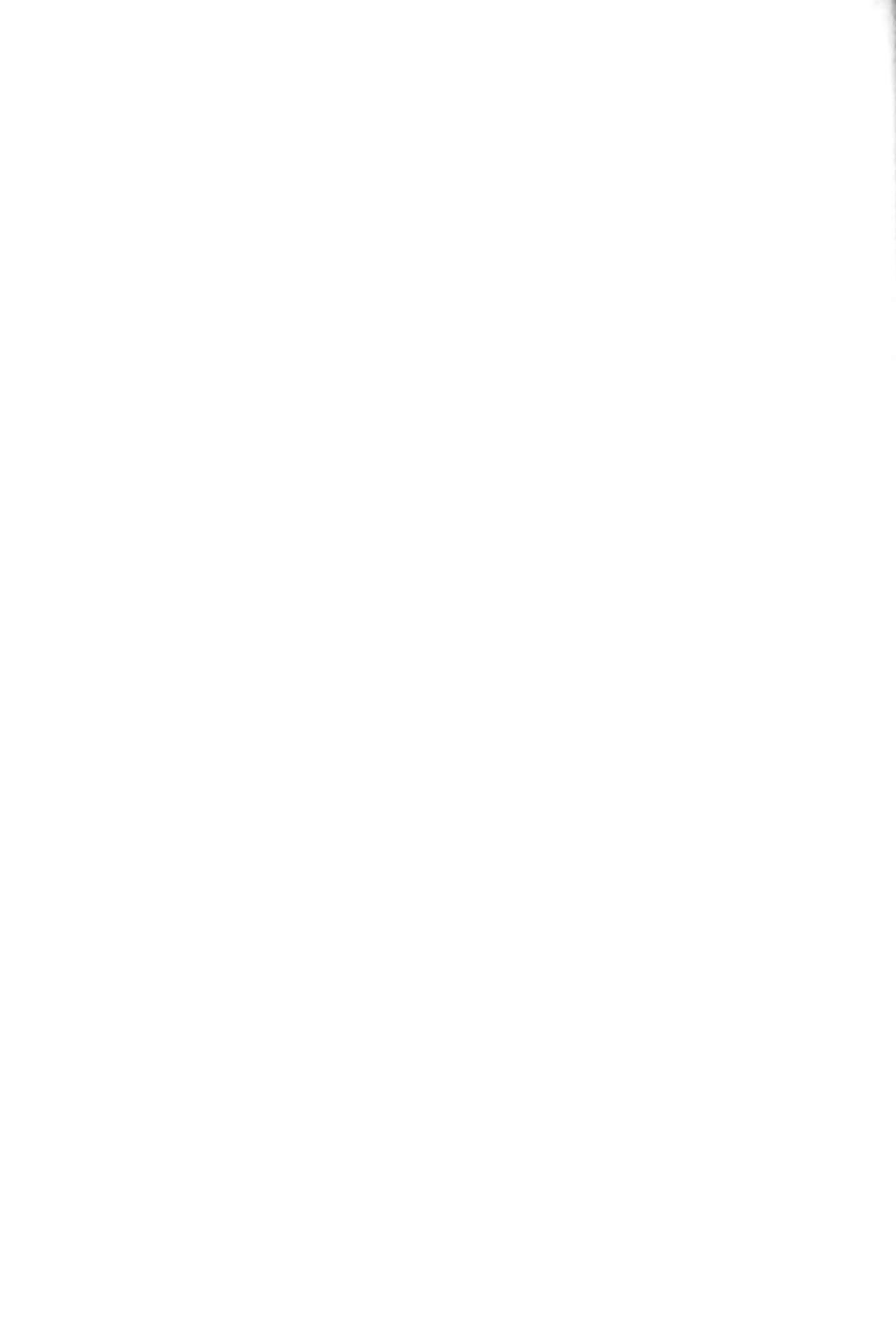
First, Taylor's expansion of the bounding temperatures about T_0 is taken:

$$T_1 = T_0 + S_1 \frac{\partial T}{\partial x} + \frac{S_1^2}{2} \frac{\partial^2 T}{\partial x^2} + \dots \quad (1)$$

$$T_2 = T_0 - S_2 \frac{\partial T}{\partial x} + \frac{S_2^2}{2} \frac{\partial^2 T}{\partial x^2} + \dots \quad (2)$$

$$T_3 = T_0 + S_3 \frac{\partial T}{\partial y} + \frac{S_3^2}{2} \frac{\partial^2 T}{\partial y^2} + \dots \quad (3)$$

$$T_4 = T_0 - S_4 \frac{\partial T}{\partial y} + \frac{S_4^2}{2} \frac{\partial^2 T}{\partial y^2} + \dots \quad (4)$$



where;

$$a_i = f(S)$$

by comparing equation (5) with equation (1) and (2) and collecting like terms, we find

$$a_o = \frac{s_1 - s_2}{s_1 s_2} \quad , \quad a_1 = \frac{s_2}{s_1 (s_1 + s_2)} \quad , \quad a_2 = \frac{-s_1}{s_2 (s_1 + s_2)}$$

Similarly, for the second derivative at T_o

$$\frac{\partial^2 T}{\partial x^2} = a_o t_o + a_1 T_1 + a_2 T_2 \quad (6)$$

where comparison with equation (1) and (2) yields

$$a_o = \frac{-2}{s_1 s_2} \quad , \quad a_1 = \frac{2}{s_1 (s_1 + s_2)} \quad , \quad a_2 = \frac{2}{s_2 (s_1 + s_2)}$$

changes in the y-direction are treated in like manner.

Substituting the different forms for $\partial T / \partial x$, $\partial T / \partial y$, $\partial^2 T / \partial x^2$, and $\partial^2 T / \partial y^2$ in the equation of heat flow, we find the value of T_o as a function of T_1 , T_2 , T_3 , T_4 , and the grid spacing.



APPENDIX 6

Notes on the use of the computer model

A. Cost and Size

The computer program is Fortran coded and consists of approximately 1300 input cards. Runs were done on an IBM 360 using the G compiler. The program requires between 400 K and 450 K bytes of core and generally took six minutes to execute. With low priority specified, then, costs averaged around thirty dollars per run.

B. Stability

The model is data sensitive and will diverge rapidly with the slightest provocation. Stability can be controlled by input of the proper grid multipliers:

Y MULT 1, Y MULT 2, GX 12, and GX 34 . The model will not converge if these multipliers are greater than one. The particular values required for convergence depend on the magnitude of the welding speed: higher speeds require smaller multipliers. As a reference, it was found that a speed of 50 ft/hr and a multiplier of 0.75 will assure convergence. Unfortunately, as the multipliers are decreased the core requirements increase to the point where the program exceeds capability.

C. Output

The output is presented in terms of node point temperatures.



Columns run parallel to the welding direction with column one identifying the long axis of the weld pool. Rows run perpendicular to the welding direction. The intersection of row one and column one mark the location of the origin of the system. Rows with positive incremental distances are read forward of row one in the direction of welding; these with negative distances in the opposite direction.

The output may be plotted in two ways. Each node temperature may be transferred to a grid, thereby identifying the isothermal contours surrounding the molten pool, or, the temperatures along a particular column may be so plotted to indicate how the temperature varies at some desired point from the center line of the pool (column one being at zero distance).

D. Description of Input Cards

1. First sixty cards are Bessel functions: argument, I_0 , I_1 , K_0 , K_1 . Format (5F 10.5)
2. NOPRP. Format (I3). Number of temperatures for the thermal properties of the specimen.
3. TRED, TKRD, TDRD. Format (3F10.5). Temperature, Conductivity, and Diffusivity of the specimen.
4. VOLT, AMP. Format (2F10.5). Voltage and amps for particular welding run.



5. TM, THIKRD, DISTV, TIME. Format (4F10.5). Melt-
ing Temp. ($^{\circ}\text{R}$), Specimen Thickness (IN), arc
travel distance (In), Time (Min).
6. TA. Format (F10.5). Ambient Temp. ($^{\circ}\text{R}$).
7. CRAD. Format (E12.4). Radiation Heat Transfer
Coefficient = $\sigma (E_t + E_b)$, BTU/ft²-hr-R⁴.
8. AVCON. Format (F10.5). Average Conductivity of
Specimen, BTU/ft-hr-R.
9. XLONG, YLONG, XLONF. Format (3F10.5). Length (In)
and Width (In) of Computational Block. XLONF is
the distance in inches from front of pool to forward
edge of block.
10. 6X12, 6X34, YMULT1, YMULT2. Format (4F10.5).
Grid multipliers which control the fineness of the
grid: forward of the arc, behind the arc, from the
base line to the thermocouple, and from the thermo-
couple to the edge of block.
11. NWANT. Format (I3). Specifies the number of grid
lines between each two points on the molten pool
boundary. (2 works well.)
12. ITCHK. Format (I3). Number of iterations.
(100-150).
13. ITCHK. Format (I3). Number of iterations at which
output is printed to check convergence.
14. ICONV. Format (I3). Number of convergence checks.

15. ITRNT. Format (I3). Number of iterations forward of the pool.
16. NDONT. Format (I3). Number of rows that need not be printed since convergence was not attained.
17. IFPRT. Format (I3). "1" if printed output is desired; "0" if not.
18. NEVIS1. Format (I3). First column to be printed.
19. NEVIS2. Format (I3). Last column to be printed.
20. NEVISI. Format (I3). Steps between first and last column.
21. EPSI. Format (F10.5). Tolerance for calculated Temperatures. (.001)
22. AKHAR. Format (F10.5). Coefficient in Rosenthal's equation for initializing grid.
23. JWAN. Format (I3). Number of curves to be plotted. (2 maximum)
24. NTC. Format (I3). Number of thermocouples or points, required (max of 2).
25. YTC(I). Format (F10.5). Distance in inches for each thermocouple from base line.
26. ITCN(I). Format (I5). Thermocouple number, or any dummy identification.



APPENDIX 7
COMPUTER MODEL FOR
TEMPERATURE DISTRIBUTION



```

DIMENSION X(50),Y(50),XN(50),YN(50),XNO(50),TCPL(2)
DIMENSION CONTF(40),RFOF(40),WHATF(40),XNF(25),YND(99),YNF(25)
DIMENSION XFF(75),YFF(75)
DIMENSION XNEWB(80),YNEWB(80),CKJBF(40),AFF(35),SLOPF(40),XND(99)
DIMENSION XFB(80),YFB(80),AXF(50,70),AYF(50,70),AXB(300,70)
DIMENSION AYB(300,70),TNF(50,70),TNB(300,70),NPRT(6),XPB(300)
DIMENSION XGRP(50),YGRP(50),XTGR(300),TGRP(300),JGRP(15),XPF(75)
DIMENSION YFC(2)
DIMENSION MAJOR(50)
DIMENSION BIORD(60),BKORD(60),BKIRD(60),RRED(60)
DIMENSION BIIRD(60)
DIMENSION TRED(50),TKRD(50),TDRD(50)
DIMENSION ITCN(2)
COMMON BIORD,BIIRD,BKORD,BKIRD,RRED
COMMON TRED,TKRD,TDRD
COMMON QBOIL
COMMON TM,TA,AMIN,TDEL,CN,CCONV,CRAD,SPD,THICK
COMMON TKXTD,TDXTD,TKMTD,TDMTD,NGPRP
WRITE(6,99)
99  FORMAT(2X,6H CHECK)
READ 4002,(RRED(I),BIORD(I),BIIRD(I),BKORD(I),BKIRD(I),I=1,60)
4003  FORMAT(5F10.5)
READ 4001,NGPRP
4001  FORMAT(I3)
READ 4002,(TRED(I),TKRD(I),TDRD(I),I=1,NGPRP)
4002  FORMAT(3F10.5)
READ 4003,VOLT,AMP
READ 4004,TM,THIKRD,DISTV,TIME
PRINT 4004, TM,THIKRD,DISTV,TIME

C
C
READ 4006,TA
READ 4005,CRAD
READ 4006,AVCON
READ 4002,XLUNG,YLONG,XLONF
PRINT 4002,XLUNG,YLONG,XLONF

```



```

C
C
READ 4004,GX12,GX34,YMULT1,YMULT2

```

```

READ 4001,NWANT
READ 4001,ITCHK
READ 4001,ITPCK
READ 4001,ICONV
WRITE(6,99)
READ 4001,ITFRNT
READ 4001,NDONT
READ 4001,IFPRT
READ 4001,NEVIS1
READ 4001,NEVIS2

```

```

C
C
READ 4001,NEVISI
READ 4006,EPSI
READ 4006,AKHAR
PRINT 4006,AKHAR

```

```

C
C
READ 4001,JWAN
4006 FORMAT(F10.5)
4008 FORMAT(F20.5)
4003 FORMAT(2F10.5)
6000 FORMAT(2F10.5)
4004 FORMAT(4F10.5)
4005 FORMAT(E12.4)
AMIN=TDRD(1)
DO 2001 I=2,NOPRP
IF(AMIN-TDRD(I))2001,2001,2002
2002 AMIN=TDRD(I)
2001 CONTINUE
THICK=THIKRD/12.
SPD=DISTV*5./TIME
CN=SPD/(24.*AMIN)

```




```

TDEL=TM-TA
WRITE(6,99)
IF(JWAN-2)1000,1000,1001
1001 JWAN=JWAN-2
READ 4001,(JGRP(I+2),I=1,JWAN)
JWAN=JWAN+2

C
C      CALCULATE POINTS ON POOL CONTOUR
C
1000 T=3.415+VOLT*AMP/(2.3.1416*AVCON*THICK*TDEL)
YMN=3.16+T*.67/2.
XMN=4.42+TA*.01
YM=YMN/CN
XM=XMN/CN
N1=XM/.03
PRINT 4001,N1
X(1)=0.
Y(1)=0.
DO 1002 I=2,N1
X(I)=X(I-1)+.03
Y(I)=((1.5/XM)*2.99)*5.86*YMN+2*X(I)*.83*.83*(XM-X(I))*2.16)*.5
XN(I)=X(I)*CN
YN(I)=Y(I)*CN
1002 WRITE(6,99)
XN(1)=0.
YN(1)=0.
XN(N1+1)=XMN
YN(N1+1)=0.
X(N1+1)=XMN/CN
Y(N1+1)=0.
N=N1+1
YMAX=0.
J=0
DO 1003 I=1,N1
IF(Y(I)-YMAX)1003,1004,1004
1004 YMAX=Y(I)

```



```

J=I
1003 CONTINUE
  XWN=X(J)*CN
  YWN=Y(J)*CN
  WRITE(6,99)
  PRINT 6000,(XN(I),YN(I),I=1,N)

C
C
C
  NORMALIZE THE INPUTS

  TKXTD=(TKRD(NOPRP)-TKRD(NOPRP-1))/(TRED(NOPRP)-TRED(NOPRP-1))
  TDXTD=(TDRD(NOPRP)-TDRD(NOPRP-1))/(TRED(NOPRP)-TRED(NOPRP-1))
  TK4TD=(TKRD(1)-TKRD(2))/(TRED(1)-TRED(2))
  TDMTD=(TDRD(1)-TDRD(2))/(TRED(1)-TRED(2))
  IF(N-2)2998,2998,2999
2999 CONTINUE
  CALL DBLW(RF,AU,XN(N),XWN,YWN,XWN/2.,SCF)
  PRINT 1,RF,AU,XN(N),XMN,YMN,SCF
  FORMAT(6F10,5)
  GO TO 2997
2998 SCF=1.
  CALL ORIGIN(RF,AMIN/AMIN,XN(N),XN(N)/2.,1.,SCF)

C
C
C
  FIND THE ORIGIN FOR THE EXP POINTS

2997 DO 2004 I=1,N
2004 XND(I)=RF-XN(I)
  PRINT 2,(XND(I), I=1,N)
  FORMAT(3F10,5)
  READ 4001,NTC
  IF(NTC-1)2996,2995,2995
2996 GY13=YMULT1
  GY24=YMULT2
  YLG13=YLUNG/2.
  GO TO 2994
2995 READ 7001,(YTC(I),I=1,NTC)
7001 FORMAT(F10,5)

```



```

2993 IF(NTC-1)2996,2993,2992
    GY24=YMULT2
    GO TO 2991
2992 GY24=AUTSCL(YTC(2)-YTC(1),YMULT2/CN)*CN
2991 GY13=AUTSCL(YTC(1),YMULT1/CN)*CN
    YLG13=YTC(1)
2994 XFP=XLQNF
    XL=XLONG
    YL=YLONG
C
C      SET UP GRID
C
C      FIND LNUMBER OF NODES
    NF=(XFP*CN+RF)/GX12+1.
    NB=((XL-XFP)*CN-RF)/GX34+1.
    NP1=YLG13*CN/GY13+1.
    NP2=(YL-YLG13)*CN/GY24
    NPT=NP1+NP2
    NP1T=NP1-1
    NVF=NF-1
    NVT=NP1-1
C
    WRITE(6,99)
    IF(NTC-1)2889,2889,2888
2989 NPRT(3)=NVT
    GO TO 2987
2988 NPRT(3)=(YTC(2)-YTC(1))/(GY24/CN)+YTC(1)/(GY13/CN)+1.
2987 NPRT(2)=NP1
    NPRT(1)=1
    DO 2995 J=1,NP1T
    DO 2995 I=1,NVF
    AXF(I,J)=GX12
    AYF(I,J)=GY13
    TNF(I,J)=0.

```



```

2005 CONTINUE
DO 2006 J=NPI,NVT
DO 2006 I=1,NVF
AXF(I,J)=GX12
AYF(I,J)=GY24
TNF(I,J)=0.
2006 CONTINUE
C
C
C
      LAYOUT GRID IN BACK OF POOL

NVB=NB-1
DO 2007 J=1,NP1T
DO 2007 I=1,NVB
AXB(I,J)=GX34
AYB(I,J)=GY13
TNB(I,J)=0.
2007 CONTINUE
DO 2800 J=NPI,NVT
DO 2800 I=1,NVB
AXB(I,J)=GX34
AYB(I,J)=GY24
TNB(I,J)=0.
2800 CONTINUE
C
C
C
      FIT EXP POINTS TO GRID

CK=0.
DO 2009 I=1,N
IF(XND(I))2008,2008,2009
2008 IF(CK-1.)2010,2009,2009
2010 CK=CK+1.
NPF=I-1
2009 CONTINUE
C
C
C
      WRITE(6,99)
C

```




```

C      FIND INTERSECTIIN OF POUL WITH Y-AXIS
C
      CALL POLFN(XNU(NPF),YN(NPF),XNJ(NPF+1),YN(NPF+1),1,ACENT,CSKOC,
1SLOPC,CUNSC,RCEN,WHATC,REGIN,R1,R2,SCF)
      IF(WHATC-1)2011,2012,2011
2011 IF(WHATC-2)2013,2014,2013
2012 CALL COLUMN(YPS,0,ACENT,CSKOC,1,1,SCF)
      IF (N-2)2,15,2015,2700
2700 IF(ABS(YPS)-YN(NPF+1))2015,2015,2013
2014 YPS=RCEN
      GO TO 2015
2013 YPS=CUNSC
2015 JATOF=YPS/GY13+1.
      ZARUF=JATOF
      AYE(1,JATOF)=ZARUF*GY13-YPS
      AYB(1,JATOF)=AYE(1,JATOF)
      XGRPL(1)=0
      YGRPL(1)=YPS
      DU 2016 J=1,JATOF
      TNF(1,J)=1.
2016 TNB(1,J)=1.
      IF(JATOF-1)2017,2017,2018
2018 AYE(1,JATOF-1)=2.*GY13-AYE(1,JATOF)
      AYB(1,JATOF-1)=AYE(1,JATOF-1)
2017 NCF=NPF+1
C
      WRITE(5,99)
C
C      NUMBER THE EXP POINTS FROM THE ORIGIN
C
      DU 2019 I=1,NCF
      IF(I-1)2020,2020,2021
2020 XFF(I)=0
      YFF(I)=YPS
      GO TO 2019
2021 NTF=NCF-I+1

```



```

XFF(I)=XNO(NTF)
YFF(I)=YN(NTF)
2019 CONTINUE
C
C
C
FIT CURVE THRU POINTS IN FRONT OF THE POOL
C
NF1=NPF
NF2=NF1
2022 NXM=3
DU 2029 I=1,NF1
IF(NXM-1)2023,2024,2024
2023 CALL PNTFND(XFF(I),YFF(I),XFF(I+1),YFF(I+1),GX12,GY13/20,10,NWANT,
1NEXSF,XBT,YBT,SCF)
IF(NEXSF-1)2029,2026,2026
2026 NXM=NEXSF
I2=I+1
NF2=NF1+NEXSF
XND(I)=XFF(I)
YND(I)=YFF(I)
XND(I2)=XBT
YND(I2)=YBT
DU 2027 I3=I2,NF2
XND(I3+1)=XFF(I3)
YND(I3+1)=YFF(I3)
2027 YND(I3+1)=YFF(I3)
NF3=NF2+1
DU 2028 I4=I,NF3
XFF(I4)=XND(I4)
YFF(I4)=YND(I4)
2028 YFF(I4)=YND(I4)
2029 CONTINUE
2024 IF(NF2-NF1)2030,2030,2499
2499 NF1=NF2
GO TO 2022
2030 NCF=NF1+1
C
C
PRINT 4001,NCF
C

```



```

WRITE(6,99)
NPF=NF1
ZARCL=0.
DO 2031 I=1,NCF
  IF(NPF-I)2031,2032,2032
2032 CALL POLFN(XFF(I),YFF(I),XFF(I+1),YFF(I+1),1.,AFF(I),CKOBF(I),
  1SLOPF(I),CONTF(I),RFOF(I),WHATF(I),REGF,R1,R2,SCF)
  NL1=XFF(I)/GX12
  NL2=XFF(I+1)/GX12
  FOLCN=NL1-NL2
  NCLDF=ABS(FOLCN)
  NCNEW=NCLDF+1
  DO 2033 K=1,NCNEW
    IF(NCLDF-K)2033,2034,2034
2034 ZARCL=ZARCL+1.
    IGRPL=2
    IF(ABS(XFF(I+1)-ZARCL*GX12)-GX12/20.)2035,2035,2036
2036 IF(WHATF(I)-1.)2037,2038,2037
2037 IF(WHATF(I)-2.)2039,2040,2039
2038 CALL COLUMN(YCALF,ZARCL*GX12,AFF(I),CKOBF(I),(YFF(I+1)+YFF(I))/2.,
  11.,SCF)
    IF(YCALF-(YFF(I)+GY13))2041,2041,2039
2040 YCALF=SQRT(RFOF(I)**2-(ZARCL*GX12)**2)
    GO TO 2041
2039 YCALF=SLOPF(I)*ZARCL*GX12+CONTF(I)
    GO TO 2041
2035 YCALF=YFF(I+1)
2041 JCALF=YCALF/GY13+1.
    ICALF=ZARCL+1.
    ZARIF=JCALF
    AYF(ICALF,JCALF)=ZARIF*GY13-YCALF
    XGRPL(IGRPL)=ZARCL*GX12
    YGRPL(IGRPL)=YCALF
    DO 2042 J=1,JCALF
2042 TNF(ICALF,J)=1.
    IF(JCALF-1)2033,2033,2043

```



```

2)343 AYF(ICALF,JCALF-1)=2.*GY13-AYF(ICALF,JCALF)
2)333 CONTINUE
2)331 CONTINUE
C
WRITE(6,99)
ICENF=XFF(NCF)/GX12+1.
ZARCF=ICENF
AXF(ICENF,1)=ZARCF*GX12-XFF(NCF)
TNF(ICENF,1)=1.
IF(ICENF-1)2044,2044,2045
2)345 AXF(ICENF-1,1)=2.*GX12-AXF(ICENF,1)
2)344 ZARRW=1.
DO 2046 I=1,NCF
IF(NPF-I)2046,2047,2047
2)347 NK=NCF-I+1
NR0W1=YFF(NK)/GY13
NR0W2=YFF(NK-1)/GY13
FDJRN=NR0W1-NR0W2
NR0DF=ABS(FDJRN)
NRNEW=NR0DF+1
DJ 2048 KR0WF=1,NRNEW
IF(NR0DF-KR0WF)2048,2049,2049
2)349 ZARRW=ZARRW+1.
IGRPL=3
IF(ABS(YFF(NK-1)-ZARRW*GY13)-GY13/20.2050,2050,2051
2)351 IF(WHATF(NK-1)-1.2052,2052,2053,2052
2)352 IF(WHATF(NK-1)-2.2054,2054,2055,2054
2)353 CALL ROWPS(XCALF,ZARRW*GY13,AF(NK-1),CKOBF(NK-1),XFF(NK)+XFF(NK-
11))/2.,1.,SCF)
IF(XCALF-(XFF(NK)+GX12)2056,2056,2054
2)355 XCALF=SQRT(RFOF(NK-1)*2-(ZARRW*GY13)**2)
GJ TJ 2)356
2)354 XCALF=(1./SLOPF(NK-1))*(ZARRW*GY13)+(-1./SLOPF(NK-1))*CONTF(NK-1)
GJ TJ 2)356
2)350 XCALF=XFF(NK-1)
2)356 ICALR=XCALF/GX12+1.

```




```

JCALR=ZARRW+1.
ZARIR=ICALR
AXF(ICALR,JCALR)=ZARIR*GX12-XCALF
XGRPL(IGRPL)=XCALF
YGRPL(IGRPL)=ZARRW*GY13
IF(ICALR-1)2048,2048,2057
2057 AXF(ICALR-1,JCALR)=2.*GX12-AXF(ICALR,JCALR)
2048 CONTINUE
2046 CONTINUE
C
WRITE(6,99)
C
C      FIT CURVE BEHIND ARC
C
XFB(1)=0.
YFB(1)=YPS
NPB=N-NPF
NCB=NPB+1
NMR=1
DO 2058 I=2,NCB
  NMR=NMR+1
  NTB=NPB+NMR
  XFB(I)=XND(NTB)
  YFB(I)=YN(NTB)
  NB1=NPB
  NB2=NB1
2067 NEXTM=1
DO 2059 I=1,NB1
  IF(NEXTM-1)2060,2061,2061
2060 CALL PNTFND(XFB(I),YFB(I),XFB(I+1),YFB(I+1),GX34,GY13/2.,1.,NWTANT,
    1NEXB,XBTB,YBTB,SCF)
  IF(NEXB-1)2059,2062,2062
2062 NEXTM=1
  I2=I+1
  NB2=NB1+1
  XND(I)=XFB(I)

```



```

YND(I)=YFB(I)
XND(I2)=XBTB
YND(I2)=YBTB
DO 2063 I3=I2,NB2
XND(I3+1)=XFB(I3)
2063 YND(I3+1)=YFB(I3)
NB3=NB2+1
DO 2064 I4=I,NB3
XFB(I4)=XND(I4)
2064 YFB(I4)=YND(I4)
2059 CONTINUE
2061 IF(NB2-NB1)2065,2065,2066
2065 NB1=NB2
GO TO 2067
2065 NCB=NB1+1
C
WRITE(6,99)
NPB=NB1
ZARCB=J0
JMAX=1
C
PRINT 4001,NCB
DO 2068 I=1,NCB
IF(NPB-I)2068,2069,2069
2069 CALL PULFN(XFB(I),YFB(I),XFB(I+1),YFB(I+1),I0,AFOB,CKOBB,SLOPB,
1CONTB,RFOB,WHATB,REGB,R1,R2,SCF)
NCL1=XFB(I)/GX34
NCL2=XFB(I+1)/GX34
FDLCN=NCL1-NCL2
NCLDF=ABS(FDLCN)
NBNEW=NCLDF+1
DO 2070 K=1,NBNEW
IF(NCLDF-K)2070,2071,2071
2071 ZARCB=ZARCB-1
IGRPL=4
IF(ABS(XFB(I+1))-ZARCB*GX34)-GX34/20.12072,2072,2073

```



```

2073 IF(WHATB-1.)2074,2075,2074
2074 IF(WHATB-2.)2076,2077,2076
2075 CALL COLUMN(YCALB,ZARCB*GX34,AFOB,CKOBB,(YFB(I+1)+YFB(I))/2.,1.,SC
1F)
      IF(YCALB-(YFB(I)+2.*GY13))2078,2078,2076
2077 YCALB=SQRT(RFOB**2-(ZARCB*GX34)**2)
      GO TO 2078
2076 YCALB=SLOPB*ZARCB*GX34+CCNTB
      GO TO 2078
2072 YCALB=YFB(I+1)
2078 JCALB=YCALB/GY13+1.
      IF(JMAX-JCALB)2079,2080,2080
2079 JMAX=JCALB
      IMAX=-ZARCB+1.
2080 ICALB=-ZARCB+1.
      ZARIB=JCALB
      AYB(ICALB,JCALB)=ZARIB*GY13-YCALB
      XGRPL(IGRPL)=ZARCB*GX34
      YGRPL(IGRPL)=YCALB
      DO 2081 J=1,JCALB
2081 TNB(ICALB,J)=1.
      XNEWB(ICALB)=ZARCB*GX34
      YNEWB(ICALB)=YCALB
      IF(JCALB-1)2077,2070,2083
2083 AYB(ICALB,JCALB-1)=2.*GY13-AYB(ICALB,JCALB)
2070 CONTINUE
2068 CONTINUE
C
      WRITE(6,99)
      ICENB=ABS(XFB(NCB))/GX34+1.
      ZARCB=ICENB
      AX3(ICENB,1)=ZARCB*GX34-ABS(XFB(NCB))
      TNB(ICENB,1)=1.
      IF(ICENB-1)2084,2084,2085
2085 AXB(ICENB-1,1)=2.*GX34-AXB(ICENB,1)
2084 ZARWB=0.

```



```

2337 IF(AXB(ICENB,1)-GX34)2086,2087,2086
      ICNEW=ICENB
      GO TO 2088
2386 ICNEW=ICENB+1
2388 XNEWB(ICNEW)=XFB(NCB)
      YNEWB(ICNEW)=0.
      XNEWB(1)=0.
      YNEWB(1)=YPS
      ARW=1.
      OVER=0.
      DO 2089 I=2, ICNEW
2390   NKB=ICNEW-I+2
      NRW1=YNEWB(NKB)/GY13
      NRW2=YNEWB(NKB-1)/GY13
      BDORN=NRWB1-NRWB2
      NROCB=ABS(BDORN)
      IF(NROCB-1)2089,2091,2091
2091   CALL POLFN(XNEWB(NKB),YNEWB(NKB),XNEWB(NKB-1),YNEWB(NKB-1),1.,AFOB
1,CKOBB,SLOPB,CONTB,RFOB,WHATB,REGB,R1,R2,SCF)
      DO 2092 KROB=1,NROCB
      ZARWB=ZARWB+ARW
      IGKPL=5
2312 IF(WHATB-1)2093,2094,2093
2093   IF(WHATB-2)2095,2096,2095
2094   CALL RWPS(XCALB,ZARWB*GY13,AFOB,CKOBB,(XNEWB(NKB)+XNEWB(NKB-1))/
12.,1.,SCF)
      IF(ABS(XCALB)-ABS(XNEWB(NKB)-GX34))2097,2097,2095
2096   XCALB=SQRT(RFOB**2-(ZARWB*GY13)**2)
      GO TO 2097
2095   XCALB=((1./SLOPB)*((ZARWB*GY13)+(-1./SLOPB)*CONTB)
      GO TO 2097
2311 XCALB=XNEWB(NKB-1)
2397 ICALB=ABS(XCALB/GX34)+1.
      ZARIB=ICALB
      JCALB=ZARWB+1.
      IF(JCALB-JMAX)2098,2099,2100

```




```

2199 IF(OVER-1.)2101,2102,2102
2101 ARW=0.
    OVER=1.
    GO TO 2103
2198 IF(OVER-1.)2104,2102,2102
2100 JVER=1.
2102 ARW=-1.
    GO TO 2103
2104 ARW=1.
2103 AXB(ICALB,JCALB)=ZARIB*GX34-ABS(XCALB)
    XGRPL(IGRPL)=XCALB
    YGRPL(IGRPL)=ZARWB*GY13
    IF(ICALB-1)2102,2102,2106
2106 AXB(ICALB-1,JCALB)=2.*GX34-AXB(ICALB,JCALB)
2102 CONTINUE
2103 CONTINUE
2107 IMAX1=IMAX-1
    DO 2108 J=1,JMAX
    DO 2109 I=1,IMAX
        IF(TWB(I,J)-1.)2110,2108,2110
2110 IF(AXB(I,J)-GX34)2111,2109,2112
2111 AXB(I,J)=GX34-AXB(I,J)
        GO TO 2108
2112 AXB(I,J)=GX34
2109 CONTINUE
2108 CONTINUE
C
    WRITE(6,99)
C
C
C
        GRID LAYOUT FINISHED
C
    DO 2113 I2=1,NF
    ZGR=NF-I2
2113 XTGR(I2)=GX12+ZGR
    DO 2114 I2=1,NVB
    ZGR=I2

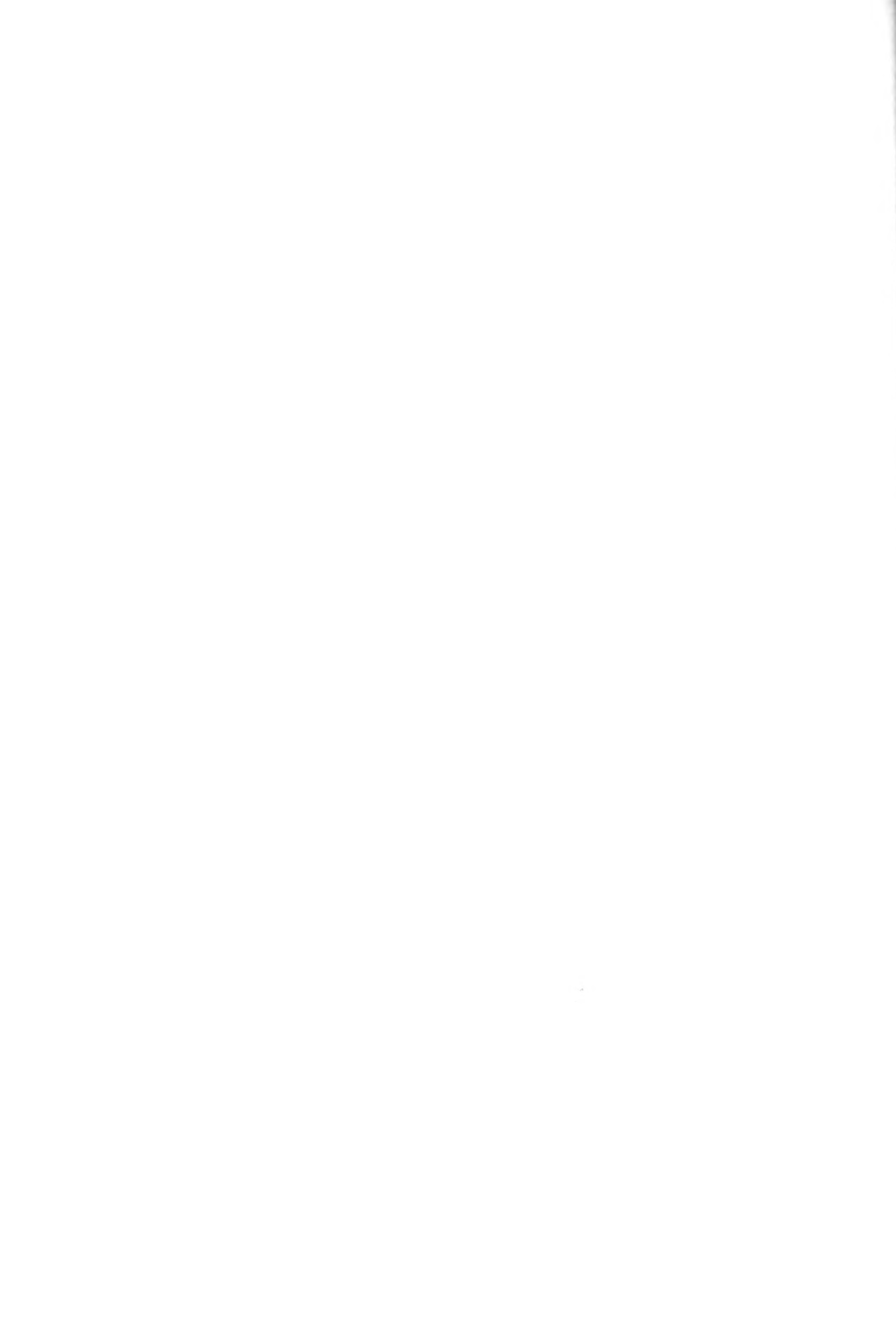
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ID=NF+I2
2114 XTGR(ID)=-GX34*ZRGR
      IGT=NB+NVF
      AFG=IGT-NDONT
C
C      INITIALIZE NODES IN FRONT OF ARC
C
      CALL BESSEL(-XNO(N),BIOJ,BIJ,BKOJ,BKIJ)
      ATMP=1.
      CTMP=EXP(ATMP*XNO(N))/(AKHAR*BKOJ)
      DO 2115 J=1,NVT
      AX8(NVB,J)=GX34*.1
      TNF(NF,J)=J,
2115 TNB(NB,J)=J,
      DO 2116 I=1,NF
      AYF(I,NVT)=GY24*.1
2116 TNF(I,NPT)=J,
      DO 2117 I=1,NB
      AYB(I,NVT)=GY24*.1
2117 TNB(I,NPT)=J,
      DO 2118 J=1,NP1
      DO 2118 I=1,NF
      IF (TNF(I,J)-1.)2119,2118,2120
2119 ZARX=I-1
      ZARY=J-1
      TNF(I,J)=DRAJH(CTMP,ATMP,ZARX*GX12,ZARY*GY13)
      IF(TNF(I,J)-1.)2118,2120,2121
2121 TNF(I,J)=J,
2118 CONTINUE
      ZANP1=NP1-1
      NODP2=NP2+1
      DO 2121 K=2,NODP2
      DO 2121 I=1,NF
      J=K+NP1-1
      IF(TNF(I,J)-1.)2122,2121,2123
2122 ZARX=I-1

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```

ZARY=K-1
TNF(I,J)=DRAJH(CTMP,ATMP,ZARX*GX12,ZANP1*GY13+ZARY*GY24)
IF(TNF(I,J)-1.)2121,2123,2123
2123 TNF(I,J)=.9
2121 CONTINUE
C
WRITE(6,99)
C
C      INITIALIZE NODES BEHIND ARC
C
DO 2124 J=1,NP1
DJ 2124 I=1,NVB
IF(TNB(I,J)-1.)2125,2124,2127
2125 ZARX=1-I
ZARY=J-1
TNB(I,J)=DRAJH(CTMP,ATMP,ZARX*GX34,ZARY*GY13)
IF(TNB(I,J)-1.)2124,2127,2127
2127 TNB(I,J)=.9
2124 CONTINUE
DO 95 J=1,NP1
CK=.5
DJ 95 I=1,NVB
IF(TNB(I+1,J)-TNB(I,J))96,95,95
96 IF(CK-1.)97,98,98
97 CK=CK+1.
ZARX=I
XNUM=ZARX*GX34
ZARX=I
DEN=ZARX*GX34
TNB(I,J)=TNB(I,J)*(XNUM/DEN)**2
95 CONTINUE
DJ 2128 K=2,NODP2
DJ 2128 I=1,NVB
J=K+NP1-1
IF(TNB(I,J)-1.)2129,2128,2131
2129 ZARX=1-I

```



```

ZARY=K-1
TNB(I,J)=DRAJH(CTMP,ATMP,ZARX*GX34,ZANP1*GY13+ZARY*GY24)
IF(TNB(I,J)-1.)2128,2131,2131
2131 TNB(I,J)=.9
2128 CONTINUE
DJ 9J K=2,NODP2
CK=.1
DO 90 I=1,NVB
J=K+NP1-I
IF(TNB(I+1,J)-TNB(I,J))91,90,90
91 IF(CK-1.)92,93,93
92 CK=CK+.1
ZARX=I
XNUM=ZARX*GX34
ZARX=I
DEN=ZARX*GX34
TNB(I,J)=TNB(I,J)*(XNUM/DEN)*.2
93 CONTINUE
3)
C
C
C
      SKART CALCULATING TEMP PROFILES
ITER=-1
NGRAPH=0
ITPRT=ITPCK
DO 2132 J=1,NVT
2132 MAJBUR(J)=IMAX
2133 IF(ITER-ITCHK)2134,2135,2136
2134 ITER=ITER+1
ITPRT=ITPRT+1
IF(ITPRT-ITPCK)2137,2138,2138
2138 DO 2139 IPRT=1,ICONV
NPRTD=NPRTD(IPRT)
PRINT 9999
5999 FORMAT(1H1)
PRINT 25.0,(ITER,NPRTD,I,AXF(I,NPRTD),AYF(I,NPRTD),TNF(I,NPRTD),
1I=1,NF)

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```

2500 FORMAT(15X,18HNO. OF ITERATIONS=I3,3X,9HCOLUMN= ,I3,3X,6HROW= ,
      1I3,3X,9HX-INCR.= ,F10.5,3X,9HY-INCR.= F10.5,3X,7HTEMP= ,F10.5)
      PRINT 9999
2139 PRINT 2500, (ITER,NPRTD,I,AXB(I,NPRTD),AYB(I,NPRTD),TNB(I,NPRTD),
      1I=1,NB)
      PRINT 9999
      DO 2140 I2=1,NF
      ID=NF-I2+1
2140 TGRP(I2)=TNF(ID,NP1)
      DO 2141 I2=1,NVB
      ID=NF+I2
2141 TGRP(ID)=TNB(I2+1,NP1)
      ITPRT=)
      NGRAPH=NGRAPH+1
2137 DO 2145 J=1,NVT
      IJID=MAJBUR(J)
2222 DO 2147 I=IMAX,IJDID
      IF(TNB(I,J)-1.0)2148,2147,2148
2148 IF(J-1)2147,2149,2150
2149 BBMN=AYB(I,J)
      T4MN=TNB(I,J+1)
      GO TO 2151
2150 BBMN=AYB(I,J-1)
      T4MN=TNB(I,J-1)
2151 CALL DIFEQ(TNB(I,J),TNB(I,J),TNB(I-1,J),TNB(I+1,J),TNB(I,J+1),T4MN
      1,AXB(I-1,J),AXB(I,J),AYB(I,J),BBMN,1.0)
2147 CONTINUE
      DO 2152 I=IJDID,NVB
      IF(TNB(I,J)-1.0)2153,2152,2153
2153 IF(J-1)2152,2154,2155
2154 BBMN=AYB(I,J)
      T4MN=TNB(I,J+1)
      GO TO 2156
2155 BBMN=AYB(I,J-1)
      T4MN=TNB(I,J-1)
2156 CALL DIFEQ(TNB(I,J),TNB(I,J),TNB(I-1,J),TNB(I+1,J),TNB(I,J+1),T4MN,AX

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1B(I-1,J),AXB(I,J),AYB(I,J),BBMN,1.0)
IF(ABS(TNB(I-J)-TNB(I,J))-EPSI)2157,2157,2158
2158 TNB(I,J)=TNBDI
2152 CONTINUE
2157 IF(I-NVB)2159,2160,2160
2159 MAJOR(J)=I
TNB(NB,J)=1.0*TNB(NVB,J)-0.1*TNB(NVB-1,J)
GO TO 2161
2159 MAJOR(J)=I+1
2151 IF(IMAX-2)2162,2162,2163
2163 DO 2164 I=2,IMAX
I=IMAX-I+1
IF(TNB(I,J)-1.0)2165,2164,2165
2165 CALL DIFEQ(TNB(I,J),TNB(I,J),TNB(I-1,J),TNB(I+1,J),TNB(I,J+1),TNB(
I,J-1),AXB(I-1,J),AXB(I,J),AYB(I,J),AYB(I,J-1),1.0)
2154 CONTINUE
2162 IF(TNF(1,J)-1.0)2166,2167,2166
2166 CALL DIFEQ(TNF(1,J),TNF(1,J),TNF(2,J),TNF(1,J+1),TNF(1,J-
1),AXF(1,J),AXB(1,J),AYF(1,J),AYF(1,J-1),1.0)
TNB(1,J)=TNF(1,J)
2167 IF(ITER-ITERNT)2168,2168,2146
2168 DO 2169 I=2,NVF
IF(TNF(I,J)-1.0)2170,2169,2170
2170 IF(J-1)2169,2171,2172
2171 BBMN=AYF(I,J)
T4MN=TNF(I,J+1)
GO TO 2173
2172 BBMN=AYF(I,J-1)
T4MN=TNF(I,J-1)
2173 CALL DIFEQ(TNF(I,J),TNF(I,J),TNF(I+1,J),TNF(I-1,J),TNF(I,J+1),T4MN
1,AXF(I,J),AXF(I-1,J),AYF(I,J),BBMN,1.0)
2169 CONTINUE
2146 CONTINUE
DO 2174 KD=1,NB
2174 TNB(KD,NPT)=1.0*TNB(KD,NVT)-0.1*TNB(KD,NVT-1)
DO 2175 KD=2,NF

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2175 TNF(KD,NPT)=1.01*TNF(KD,NVT)-.01*TNF(KD,NVT-1)
GO TO 2133
C
C
C      OUTPUT FORMS
2135 PRINT 3000
3000 FORMAT(44X,29H 材料数据 MATERIAL DATA 材料数据,/)
PRINT 3001
3001 FORMAT(20X,31HMATERIAL 材料 MILD STEEL,/)
PRINT 3002,IM
3002 FORMAT(51X,34HMELTING TEMPERATURE,(DEG.R) 材料熔点 F12.5,/)
PRINT 3003,THIKRD
3003 FORMAT(64X,24HTHICKNESS,(IN.) 材料厚度 F12.5,/)
PRINT 3004
3004 FORMAT(45X,29H 焊接速度: WELDING DATA 焊接速度,/)
PRINT 3005,SPD
3005 FORMAT(56X,35H SPEED OF TRAVEL, (IN/MIN) 材料速度 F12.5,/)
PRINT 3006,VOLT
3006 FORMAT(64X,26H VOLTAGE, (VOLTS) 材料电压 F12.5,/)
PRINT 3007,AMP
3007 FORMAT(65X,25H CURRENT, (AMPS) 材料电流 F12.5,/)
PRINT 3008
3008 FORMAT(45X,33H 熔池数据: MOLTEN POOL DATA 熔池数据,/)
XWARD=XWN/CN
YWARD=YWN/CN
PRINT 3009,X(N)
3009 FORMAT(67X,22H LENGTH, (IN) 材料长度 F12.5,/)
PRINT 3010,XWARD,IMAX
3010 FORMAT(5X,05H DISTANCE FROM THE FRONT OF THE POOL TO MAX. WIDTH, (
1IN) 材料距离 F12.5,5X,27H CORRESPONDING ROW 材料行 I3)
PRINT 3011,YWARD,JMAX
3011 FORMAT(41X,29H MAXIMUM WIDTH, (IN) 材料最大宽度 F12.5,2X,30H CORRESPOND
ING COLUMN 材料列 I3,/)
PRINT 3012
3012 FORMAT(28X,65H 材料数据: DATA SUPPLIED FOR CURVE FITTING OF THE WELD
1 POOL 材料数据,/)

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PRINT 3013
3013 FORMAT(23X,20H DISTANCE FROM FRONT,2X,26H NORMALIZED VALUE MEASURE
10,2X,13H DISTANCE FROM THE,8X,11H MORMALIZED)
PRINT 3014
3014 FORMAT(15X,3HNO,6X,18HUF THE POOL,(IN,1),2X,16H FROM THE ORIGIN,1
11X,17H LINE OF SYMMETRY,9X,6H VALUE)
DO 3015 I=1,N
3015 PRINT 3016,I,X(I),XNO(I),Y(I),YN(I)
3016 FORMAT(15X,13,10X,F10.5,13X,F10.5,13X,F10.5,13X,F10.5)
PRINT 3017
3017 FORMAT(////)
PRINT 3018
3018 FORMAT(43X,33H MEASUREMENT DATA 测量数据,/)
DO 3019 I=1,NTC
READ 10,ITCN(I)
11 FORMAT(15)
PRINT 3020,ITCN(I)
3020 FORMAT(64X,23H THERCOUPLE NO,0000000,5X,15)
PRINT 3021,YTC(I)
3021 FORMAT(39X,46H DISTANCE FROM LINE OF SYMMETRY,(IN)0000000 F12.5)
JTHRM=NPRT(I+1)
PRINT 3022,JTHRM
3022 FORMAT(38X,51H COLUMN CORRESPONDING TO THE ABOVE DISTANCE 0000000,
15X,13,/)
3019 CONTINUE
PRINT 3023
3023 FORMAT(35X,49H COMPUTATION DATA AND ASSUMPTIONS 计算数据,/)
1)
PRINT 3024,CRAD
3024 FORMAT(29X,61H RADIANT HEAT TRANSFER COEF,(BTU/FT*2-HR-DEG.R*4)
1000000 E12.4,/)
PRINT 3026,TA
3026 FORMAT(52X,36H AMBIENT WATER TEMP,(DEG.R)0000000 F12.5,/)
PRINT 3027,AMIN
3027 FORMAT(32X,56H MINIMUM VALUE OF THERMAL DIFFUSIVITY,(FT*2/HR)000
10000 F12.5,/)

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PRINT 3028,CN
3028 FORMAT(53X,34H NORMALIZING DISTANCE,(IN) 000000 F1205,/)
PRINT 3029,TDEL
3029 FORMAT(48X,40H NORMALIZING TEMPERATURE,(DEGR) 000000 F1205,/)
PRINT 3030,SCF
3030 FORMAT(68X,21H SCALE FACTOR 000000 F1205,/)
ITER=ITER+1
PRINT 3031,ITER
3031 FORMAT(60X,29H NUMBER OF ITERATIONS 000000,5X,I3,/)
JGRP(1)=NPRT(2)
JGRP(2)=NPRT(3)
NGRAPH=-1
PRINT 3032
3032 FORMAT(///)
PRINT 3033
3033 FORMAT(42X,36H 000000000000 CURVE PLOTTING DATA 0000000000,///)
DO 3034 J=1,JWAN
JD=JGRP(J)
IF(JD-NP1)3035,3035,3036
3035 ZJD=JD-1
DJD=ZJD*GY13
GO TO 3037
3036 ZJD1=NP1-1
ZJD2=JD-NP1
DJD=ZJD1*GY13+ZJD2*GY24
3037 PRINT 3038,J,DJD,JD
3038 FORMAT(6X,52H DIMENSIONLESS DISTANCE FOR CURVE WITH SYMBOL CODE,(I
13,9H) 000000,F1005,2X,29H CORRESPONDING CLUUMN 000000,I3,/)
DO 3039 I2=1,NF
ID=NF-I2+1
3039 TGRP(I2)=TNF(ID,JD)
DO 3040 I2=1,NVB
ID=NF+I2
3040 TGRP(ID)=TNB(I2+1,JD)
NGRAPH=NGRAPH+1
3034 CONTINUE

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IF(IFPRT-1)2136,3042,3042
3042 DO 3043 I=1,NB
    ZARIB=I-1
3043 XPB(I)=-ZARIB*GX34
    DO 3044 I=1,NF
    ZARIB=I-1
3044 XPF(I)=ZARIB*GX12
    DO 3045 J=NEVIS1,NEVIS2,NEVIS1
    IF(J-NP1)3046,3046,3047
3046 ZARIB=J-1
    YPRNT=ZARIB*GY13
    GO TO 3048
3047 ZARIB=NP1-1
    ZARB2=J-NP1
    YPRNT=ZARIB*GY13+ZARB2*GY24
3048 PRINT 3049
3049 FORMAT(10X,71H XXXXXXXXXX CALCULATED TEMP. IN FLAT PLATE BEING WELDED
    1 UNDERWATER XXXXXXXXXX,///)
    PRINT 3050,(J,YPRNT,I,XPF(I),AXF(I,J),AYF(I,J),TNF(I,J),I=1,NVF)
3050 FORMAT(5X,8HCOLUMN=I3,17H CUR. D-L DIS. = F8.3,2X,5HROW=I3,17H,
    1CUR. D-L DIS. = F8.3,2X,7HX-IN, = F6.3,2X,7HY-IN. = F6.3,2X,10HD-L
    2 TEMP =,F8.5)
    PRINT 3051
3051 FORMAT(///)
    PRINT 3050,(J,YPRNT,I,XPB(I),AXB(I,J),AYB(I,J),TNB(I,J),I=1,NVB)
3045 CONTINUE
2136 CONTINUE
CALL EXIT
END
SUBROUTINE BESSEL (Z,BIOT,BIIT,BKOT,BKLT)
    DIMENSION BIORD(60),BIIRD(60),BKORD(60),BKIRD(60),RRED(60)
    DIMENSION TRED(50),TKRD(50),TDKRD(50)
    COMMON BIORD,BIIRD,BKORD,BKIRD,RRED
    COMMON QBOIL
    COMMON TRED,TKRD,TORD
    COMMON TM,TA,AMIN,TDEL,CN,CCUNV,CRAD,SPD,THICK

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COMMON TKXTD,TDXTD,TKMTD,TKMID,TDMTD,NOPRP
VALF(A,B,C,D,E)=A+(B-A)/(C-D)*(E-D)
IF(Z-0.05)8,9,9
IF(Z-2.095)1,1,2
Z2=2.095*Z
I=Z2
BIJT=VALF(BIORD(I),BIORD(I+1),RRED(I+1),RRED(I),Z)
BIIT=VALF(BIIRD(I),BIIRD(I+1),RRED(I+1),RRED(I),Z)
BKJT=VALF(BKORD(I),BKORD(I+1),RRED(I+1),RRED(I),Z)
BKIT=VALF(BKIRD(I),BKIRD(I+1),RRED(I+1),RRED(I),Z)
GO TO 7
IF(Z-3.095)3,3,4
Z1=Z
GO TO 5
Z1=3.095
BIOT=EXP(Z1)/SQRT(6.28318*Z1)*(1.0+1.0/(8.0*Z1))
BIIT=BIOT*(1.0-3.0/(8.0*Z1))/(1.0+1.0/(8.0*Z1))
BKJT=SQRT(1.57/Z)*EXP(-Z)*(1.0-1.0/(8.0*Z))
BKIT=BKOT*(1.0+3.0/(8.0*Z))/(1.0-1.0/(8.0*Z))
GO TO 7
BIJT=1.0+Z**2/4.0
BIIT=Z/2.0*(1.0+Z**2/8.0)
BKJT=ALOG(2.0/Z)-0.5772
BKIT=1.0/Z
RETURN
END
SUBROUTINE ORIGIN(RF,A,BIN,RINTL,TN,S)
DIMENSION BIORD(60),BIIRD(60),BKORD(60),BKIRD(60),RRED(60)
DIMENSION TRED(50),TKRD(50),TDRD(50)
COMMON BIORD,BIIRD,BKORD,BKIRD,RRED
COMMON TRED,TKRD,TDRD
COMMON TM,TA,AMIN,TDDEL,CN,CCONV,CRAD,SPD,THICK
COMMON QBQIL
COMMON TKXTD,TDXTD,TKMTD,TKMID,TDMTD,NOPRP
R1=RINTL/S
B=BIN/S

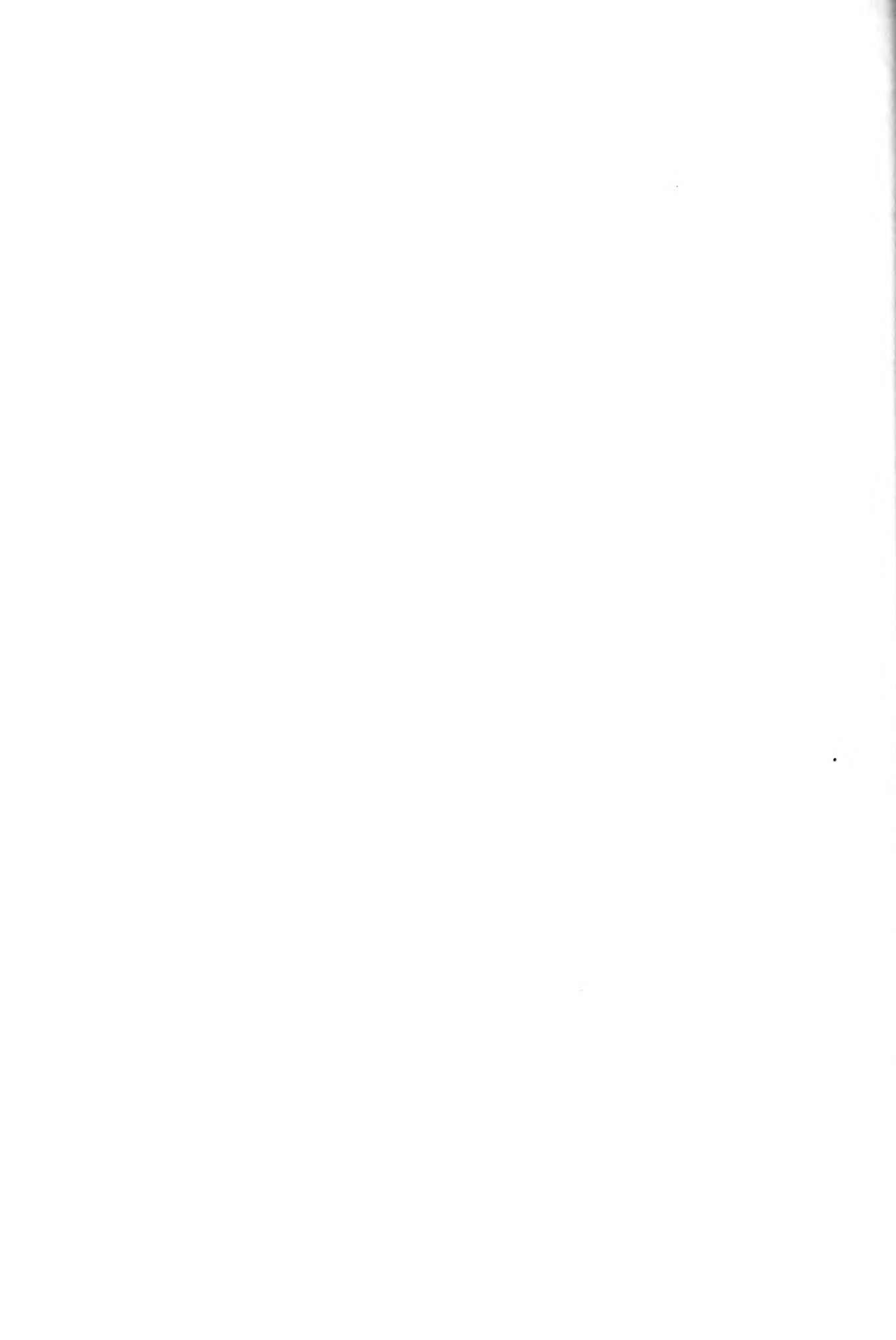
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102 CALL BESSEL(A*R1,BI01,BI11,BK01,BK11)
CALL BESSEL(A*(B-R1),BI02,BI12,BK02,BK12)
TABE=TN*EXP(A*B)*BK02-BK01
DRIV=A*(TN*EXP(A*B)*BK12+BK11)
RF=R1-TABE/DRIV
IF(ABS(R1-RF)-.001)100,100,101
101 R1=RF
GO TO 102
100 RF=RF*S
RETURN
END
SUBROUTINE PROP(TK,TKDRV,TD,TND,TRDD,TMIN,TMAX,TKMNT,TDMNT,TKMXT,
1TKMXT,TKXDR,TDXDR,TKMDR,TDMDR)
DIMENSION BIORD(60),BIIRD(60),BKORD(60),BKIRD(60),RRED(60)
DIMENSION TRED(50),TKRD(50),TORD(50)
COMMON BIORD,BIIRD,BKORD,BKIRD,RRED
COMMON TRED,TKRD,TORD
COMMON QBOIL
COMMON IM,JA,AMIN,TDEL,CN,CCONV,CRAD,SPD,THICK
COMMON TKXTD,TDXTD,TKMTD,TDMTD,NQPRP
VINTF(A,B,C,D,E)=A+(B-A)/(C-D)*(E-D)
VINTD(A,B,C,D)=(A-B)/(C-D)
T=TND-TDEL+TA
IF(T-TMIN)208,208,209
IF(T-TMAX)201,202,202
T2=T/TRDD
I=T2
IF(TRED(I)/TRDD-T2)203,204,205
IB=I
GO TO 206
205 IB=I-1
TK=VINTF(TKRD(IB),TKRD(IB+1),TRED(IB+1),TRED(IB),T)
TD=VINTF(TORD(IB),TORD(IB+1),TRED(IB+1),TRED(IB),T)
TKDRV=VINTD(TKRD(IB),TKRD(IB+1),TRED(IB),TRED(IB+1))
IF(TD-AMIN)210,207,207
204 TK=TKRD(I)

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208 TD=TDRD(I)
   TKDRV=VINTD(TKRD(I+1),TKRD(I-1),TRED(I+1),TRED(I-1))
   IF(TD-AMIN)210,207,207
   TK=TKMIN+TKMDR*(T-TMIN)
   TD=TDMIN+TDMDR*(T-TMIN)
   TKDRV=TKMDR
209 IF(TD-AMIN)210,207,207
   TK=TKMAX+TKXDR*(T-TMAX)
   TD=TDMAX+TDXDR*(T-TMAX)
   TKDRV=TKXDR
210 IF(TD-AMIN)210,207,207
   TD=AMIN
211 RETURN
   END
SUBROUTINE ROWPS(X,YIN,A,C,XINTL,TN,S)
DIMENSION BIORD(60),BIIRD(60),BKORD(60),BKIRD(60),RRED(60)
DIMENSION TRED(50),TKRD(50),TDRD(50)
COMMON BIORD,BIIRD,BKORD,BKIRD,RRED
COMMON TRED,TKRD,TDRD
COMMON QB0IL
COMMON TM,TA,AMIN,TDEL,CN,CCONV,CRAD,SPD,THICK
COMMON TKXTD,TDXTD,TKMTD,TDMTD,NGPRP
TABLEF(XDIS,BFUTO)=C*EXP(-A*XDIS)*BFUTO-TN
DRIVEF(XDIS,RDIS,BFUTO,BFUTL)=-C*A*EXP(-A*XDIS)*(BFUTO+XDIS/RDIS*
1BFUTL)
X1=XINTL
Y=YIN/S
302 R1=SQRT(X1**2+Y**2)
   CALL BESSEL(A*R1,BIORD,BIIRD,BKORD,BKIRD)
   X=X1-TAREF(X1,BKORD)/DRIVEF(X1,R1,BKORD,BKIRD)
   IF(ABS(X-X1)-.001)300,300,301
301 X1=X
   GO TO 302
303 RETURN
   END
SUBROUTINE COLUMN(Y,X,A,C,YINTL,TN,S)

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DIMENSION BIORO(60),BIIRD(60),BKORD(60),BKIRD(60),RRED(60)
DIMENSION TRED(50),TKRD(50),TDRD(50)
COMMON BIORO,BIIRD,BKORD,BKIRD,RRED
COMMON TRED,TKRD,TDRD
COMMON QB0IL
COMMON TM,TA,AMIN,TDEL,CN,CCONV,CRAD,SPD,THICK
COMMON TKXTD,TDXTD,TKMTD,TDMTO,NCPRP
Y1=YINTL/S
402 R1=SQRT(X1**2+Y1**2)
CALL BESSEL(A=R1,BIUCI,BIICL,BKOCL,BKICL)
TAOOR=R1*(TN*EXP(A*X)/C-BKOCL)/(Y1*A*BKICL)
Y=Y1-TAOOR
IF(ABS(Y-Y1)-.0001)400,400,401
401 Y1=Y
GO TO 402
Y=Y*S
403 RETURN
END
SUBROUTINE POLFN(X1,Y1IN,X2,Y2IN,TN,A,CPSKO,SLOPE,CONST,R,WHAT,REG
1 IN,R1,R2,S)
DIMENSION BIORO(60),BIIRD(60),BKORD(60),BKIRD(60),RRED(60)
DIMENSION TRED(50),TKRD(50),TDRD(50)
COMMON BIORO,BIIRD,BKORD,BKIRD,RRED
COMMON TRED,TKRD,TDRD
COMMON QB0IL
COMMON TM,TA,AMIN,TDEL,CN,CCONV,CRAD,SPD,THICK
COMMON TKXTD,TDXTD,TKMTD,TDMTO,NCPRP
Y1=Y1IN/S
Y2=Y2IN/S
R1=SQRT(X1**2+Y1**2)
R2=SQRT(X2**2+Y2**2)
Z1=R1+X1
Z2=R2+X2
IF(R2-R1)500,501,502
IF(Z2-Z1)504,504,503
502 IF(Z2-Z1)503,504,504

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5 J3  A=ALOG(R1/R2)/(2.*(Z2-Z1))
      CALL BESSEL(A*R1,B10,B11,BK0,BK1)
      CPSKO=EXP(A*X1)*TN/BK0
      WHAT=1.
      GO TO 505
5 J1  R=R1
      WHAT=2,
      GO TO 505
504  WHAT=3.
505  IF(X1+X2)506,507,507
505  REGIN=-1.
      GO TO 508
507  REGIN=1.
508  SLOPE=(Y2-Y1)*S/(X2-X1)
      CONST=Y1IN-SLOPE*X1
      RETURN
      END
SUBROUTINE PNTFND(X1,Y1,X2,Y2,GRDX,GRDY,TN,NW,NEXT,XBTWN,YBTWN,S)
DIMENSION BIORD(60),BIIRD(60),BKORD(60),BKIRD(60),RRED(60)
DIMENSION TRED(50),TKRD(50),TDRD(50)
COMMON BIORD,BIIRD,BKORD,BKIRD,RRED
COMMON TRED,TKRD,TDRD
COMMON QB0IL
COMMON TM,TA,AMIN,TDEL,CN,CCUNV,CRAD,SPD,THICK
COMMON TKXTD,TDXTD,TKMTD,TDMTD,NOPRP
SLINF(SLA,SLARG,SLB,SLC)=SLA+SLARG+SLB*SLC
CIRCLF(RCIR,ARCIR)=SQRT(RCIR**2-ARCIR**2)
NPT1=X1/GRDX
NPT2=X2/GRDX
FNPT=NPT2-NPT1
NPT=ABS(FNPT)
IF(NPT-NW)601,602,602
602  NEXT=1
      XBTWN=(X1+X2)/2.
      CALL PULFND(X1,Y1,X2,Y2,TN,AF,CK,SLOP,CONT,R,WHAT,REG,R1,R2,S)
      IF(WHAT-1.)635,636,635

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635 IF(WHAT-2.)637,638,637
636 CALL COLUMN(YBTWN,XBTWN,AF,CK,(Y1+Y2)/2.,TN,S)
633 IF(YBTWN-((Y1+Y2)/2.+GRDY))639,639,637
      YBTWN=CIRCLF(R,XBTWN)*S
      GO TO 639
637 YBTWN=SLINF(SLOP,XBTWN,1.,CONT)
      GO TO 639
631 NEXT=0
633 RETURN
      END
      SUBROUTINE ORBLW(RFS,A,XL,XL1,W,RGS,S)
      DIMENSION BIOR(60),BIIRC(60),BKORD(60),BKIRD(60),RRED(60)
      DIMENSION TRED(50),TKRD(50),TDRD(50)
      COMMON BIOR,BIIRC,BKORD,BKIRD,RRED
      COMMON TRED,TKRD,TDRD
      COMMON QBQIL
      COMMON TM,TA,AMIN,TDDEL,CN,CCONV,CRAD,SPD,THICK
      COMMON TKXTD,TDXTD,TKMTD,TDMTD,NCPRP
      RF=RGS
      CRIT=1.0
832 XW=RF-XL1
      RV1=SQRT(XW**2+XL1**2/4.)
835 FUNT=(RW1+RF-XL1)*ALOG(RF/(XL-RF))-2.*RF*ALOG(RW1/(XL-RF))
      FDRV=ALOG(RF/(XL-RF))-2.*RF/RW1
      RW=RW1-FUNT/FDRV
      IF(ABS(RW-RW1)-.001)803,803,804
834 RW1=RW
      GO TO 805
833 G=ALOG(RF/RW)
      H=XW+RW-2.*RF
      A=.5*G/H
      CALL BESSEL(A*RW,BIO,BI1,BKO,BK1)
      BKOD=-BK1
      BKID=-BKO-BK1/(A*RW)
      RWD=XW/RW
      GJ=(RW-RWD*RF)/(RW*RF)

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HD=RWD-1.
AD=.5*(GD*H-HD*G)/H**2
TABE=BKU+XW*BK1/RW
DRIV=(AD*RW+ARWD)*(BKOD+XW*BK1D/RW)+(RW-RWD*XW)*BK1/RW**2
RFS=RF-TABE/DRIV
CRITE2=(DRIV*BK0+BK1*TABE)/BK0**2
IF(CRITE1-CRITE2)800,806,801
CRITE1=CRITE2
RF=RFS
GO TO 802
800) S=W/SQRT(RW**2-XW**2)
RETURN
END
SUBROUTINE COEF(A,A1,A2,HLOSS,TKDEP,DIFFR,T,AF,AB,BF,BB,TK,TD,
1TKDRV)
DIMENSION BIORD(60),BIIRD(60),BKORD(60),BKIRD(60),RRED(60)
DIMENSION TRED(50),TKRD(50),TDRD(50)
COMMON BIORD,BIIRD,BKORD,BKIRD,RRED
COMMON TRED,TKRD,TDRD
COMMON QBUIL
COMMON TM,TA,AMIN,TDEL,CA,CCONV,CRAD,SPD,THICK
COMMON TKXTD,TDXTD,TKMTD,TDMTD,NOPRP
A=1./((AF*AB)+1.)/(BF*BB)
A1=1./((AF+AB)
A2=1./((BF+BB)
IF(T)9,10,10
T=0.
CONTINUE
10) IF(T-.035)1,1,2
IF(T-.096)3,3,4
IF(T-.17)5,5,6
3 QBUIL=(1.542*T*(TM-672.))**3
GO TO 8
5 QBUIL=2265.11*(288./(T*(TM-672.)))**5.86
GO TO 8
6 QBUIL=32.4*(T*(TM-672.))**0.75

```



```

GO TO 8
QBOIL=0.
CONTINUE
CCUNV=0.
HLOSS=(1./(12.*CN))*2/(TK*THICK)*(CCUNV*T+QBOIL/TDEL+CRAD/TDEL*(
1T*TOEL+TA)*4-TA*4)
TKDEP=TDEL*TKDRV/(2.*TK)
DIFFR=AMIN/TO
RETURN
END
SUBROUTINE DIFEQ(TNEW, TMP, TID, T2D, T3D, T4D, AFD, ABD, BFD, BBD, TN)
DIMENSION BIORD(60), BIIRD(60), BKORD(60), BKIRD(60), RRED(60)
DIMENSION TRED(50), TKRD(50), TDRD(50)
COMMON BIORD, BIIRD, BKORD, BKIRD, RRED
COMMON QBOIL
COMMON TM, TA, AMIN, TOEL, CN, CCUNV, CRAD, SPD, THICK
COMMON TRED, TKRD, TDRD
COMMON TKXTD, TOXTD, TKMTD, TDMTD, NCPRP
T(A, A1, A2, AF, AB, BF, BB, HLOSS, TKDEP, DIFFR, T1, T2, T3, T4)=(A1*(T1*(1. /
1AF+DIFFR)+T2*(1./AB-DIFFR))+A2*(T3/BF+T4/BB)+TKDEP*(A1*2*(T1-T2)
2*2+A2*2*(T3-T4)*2)-HLOSS)/A
CALL PROP(TK, TKDRV, TO, TMP, TRED(1), TRED(1), TKRD(1),
1TDRD(1), TKRD(NOPRP), TDRD(NOPRP), TKXTD, TOXTD, TKMTD, TDMTD)
CALL COEF(AD, AID, A2D, AFD, ABC, BFD, BBD, HLOSS, TKOPD, DIFRD, TID, T2D, T3D,
1TKDRV)
TNEW=T(AD, AID, A2D, AFD, ABC, BFD, BBD, HLOSS, TKOPD, DIFRD, TID, T2D, T3D,
1T4D)
IF (TNEW-1.) 1, 2, 2
TNEW=.9999
CONTINUE
RETURN
END
FUNCTION AUTSCL(YO, DN)
DIMENSION BIORD(60), BIIRD(60), BKORD(60), BKIRD(60), RRED(60)
DIMENSION TRED(50), TKRD(50), TDRD(50)
COMMON BIORD, BIIRD, BKORD, BKIRD, RRED

```

1 8

2 1



```

COMMON TRED,TKRD,TDRD
COMMON TM,TA,AMIN,DEL,CN,CCONV,CRAD,SPD,THICK
COMMON TKXTD,TDXTD,TKMTD,TDMTD,NOPRP
J=YD/DN
ZARB1=J
IF(YD-ZARB1*DN)701,700,701
701 ZARB2=J+1
    AUTSCL=YD/ZARB2
    GO TO 702
702 AUTSCL=DN
    RETURN
    END
FUNCTION DRAJH(C,A,X,Y)
DIMENSION BIORD(60),BIIRD(60),BKORD(60),BKIRD(60),RRED(60)
DIMENSION TRED(50),TKRD(50),TDRD(50)
COMMON BIORD,BIIRD,BKORD,BKIRD,RRED
COMMON TRED,TKRD,TDRD
COMMON TM,TA,AMIN,DEL,CN,CCONV,CRAD,SPD,THICK
COMMON TKXTD,TDXTD,TKMTD,TDMTD,NOPRP
R=SQRT(X**2+Y**2)
CALL BESSEL(A*R,BIO,BII,BKO,BKI)
DRAJH=C*EXP(-A*X)*BKO
RETURN
END
005 1.00006 0025 3.1142 19.91
01 1.00025 05006 2.4271 9.8538
015 1.00056 07521 2.03 6.4775
02 1.001 1005 1.7527 4.776
025 1.00156 12598 1.5415 3.747
03 1.00226 15169 1.3725 3.056
035 1.00309 17769 1.2327 2.5591
04 1.00404 20403 1.1145 2.1844
045 1.00513 23074 1.0129 1.8915
05 1.00635 25789 0.92442 1.6564
055 1.00771 28553 0.84659 1.4637
06 1.0092 3137 0.77754 1.3028

```



065	101084	034247	071589	101668
07	101263	037188	066054	100503
075	101456	040199	06106	094957
08	101665	043286	056536	086178
085	101889	046456	052424	078467
09	10213	049713	048674	071653
095	102387	053004	045246	065597
10	102661	056516	042103	06019
105	102952	060075	039218	055341
101	103201	063749	036562	050975
1015	10359	067544	034114	047031
102	103937	071467	031853	043458
1025	104304	075528	029762	040211
103	104692	079732	027827	037254
1035	105102	08409	026033	034553
104	105534	088609	024368	032082
1045	105989	093298	022821	029817
105	106467	098167	021382	027737
1055	106971	100322	020044	025824
106	1075	100848	018797	024062
1065	108056	101395	017636	022436
107	10804	101963	016553	020934
1075	109252	102555	015541	019546
108	109895	103172	014597	01826
1085	20057	103814	013713	017069
109	201277	104482	012888	015964
1095	20218	105179	012115	014938
20	202795	105906	011393	013984
2005	203609	106664	010716	013097
201	204463	107455	010082	012272
2015	205356	10828	009487	011503
202	206291	109141	00893	010787
2025	207271	20004	008407	010118
203	208296	200978	007918	009495
2035	209369	201958	007458	008913
204	300492	202981	007026	008369



2.45	3.1668	2.405	0.6621	0.7861
2.5	3.2898	2.5167	0.624	0.7385
2.55	3.4185	2.6334	0.5882	0.6941
2.6	3.5532	2.7554	0.5545	0.6524
2.65	3.6941	2.8829	0.5229	0.6134
2.7	3.8416	3.0161	0.4931	0.5769
2.75	3.9958	3.1554	0.4651	0.5427
2.8	4.1573	3.3011	0.4388	0.5106
2.85	4.3261	3.4533	0.414	0.4825
2.9	4.5027	3.6120	0.3917	0.4523
2.95	4.6875	3.7792	0.3687	0.4258
3.	4.8818	3.9534	0.3482	0.4009
5.				
5.	45.5	2.31225		
11.	44.5	2.05584		
15.	43.5	1.857996		
2.	42.5	1.639509		
25.	41.8	1.412993		
31.	41.	1.204429		
35.	4.	0.99457		
4.	39.2	0.89311		
45.	38.5	0.82750		
51.	37.6	0.76845		
55.	36.8	0.7378		
61.	36.1	0.70351		
65.	35.2	0.65597		
71.	34.5	0.60482		
75.	33.8	0.58294		
81.	32.8	0.56199		
85.	32.	0.54489		
91.	31.2	0.52739		
95.	30.5	0.50348		
1.	29.7	0.47212		
1.5.	29.	0.44432		
11.	28.2	0.41701		
115.	27.5	0.40462		



12130	2608	039177
12500	2509	035928
13000	250	033571
13500	2405	031919
14000	2308	029566
14500	2301	027598
15000	2202	024465
15500	2108	022843
16000	2101	020136
16500	2006	014922
17000	200	014046
17500	1905	012877
18000	190	013384
18500	1808	01528
19000	1801	013214
19500	1708	02351
20000	1702	022717
20500	170	022453
21000	1605	021793
21500	1601	021264
22000	1509	021
22500	1505	020533
23000	1502	020165
23500	150	019935
24000	1409	019855
24500	1405	019379
25000	1401	018895
290	1350	
31300	0116	
5035		4098 019
1317E-08		
1808		
30	05	
10	035	05
2		
30		





APPENDIX 8
UTILITY PROGRAMS




```

C      THIS PROGRAM DETERMINES THE CORRELATION PARAMETERS
C      NECESSARY FOR COMPUTING THE MAXIMUM DIMENSIONS
C      OF THE WELD POOL.
C
C      DIMENSION VOLT(100),AMP(100),THICK(100),RN(100),DIST(100)
C      DIMENSION TIME(100),HW(100),X(100),WI(100),XI(100),SPD(100)
C      DIMENSION WN(100),XN(100),B(2),XL(100),WL(100),P(100)
C      DIMENSION PL(100),PW(100),RW(100),RL(100),PI(100),P2(100)
C      READ 1,N,N
C      READ 2,PC,TM,TA,AMIN,AVCON
C      READ 3,RL,(RN(I),VOLT(I),AMP(I),DIST(I),TIME(I),THICK(I),X(I),
C      HW(I),I=1,N)
C      1000 FORMAT(I3)
C      2000 FORMAT(4F10.5)
C      3000 FORMAT(8F10.5)
C      PC=5.415/(2.43.1416*AVCON*(TM-TA))
C      DISTV=1/(2.4*AMIN)
C      NW=0
C      NL=0
C      PRINT 11
C      11  FORMAT(4X,4HRUN#,6X,4HVOLT,5X,3HAMP,3X,10HVEL(FT/HR),1X,10HLENGTH(
C      IN),2X,9HWIDTH(IN),1X,6HLUMPAP,1X,9HTHICK(IN),/)
C      DO 1 I=1,N
C      SPD(I)=5.*DIST(I)/TIME(I)
C      DISTN=SPD(I)*DISTV
C      WN(I)=2.*HW(I)/(25.4*12.)*DISTN
C      XN(I)=X(I)/(25.4*12.)*DISTN
C      WI(I)=2.*HW(I)/25.4
C      XI(I)=X(I)/25.4
C      P(I)=PC*VOLT(I)*AMP(I)*12./THICK(I)
C      PRINT 12,RN(I),VOLT(I),AMP(I),SPD(I),XI(I),WI(I),P(I),THICK(I)
C      12  FORMAT(8F10.5)
C      IF(XN(I))3,32,3
C      30  NL=NL+1
C      XN(NL)=XN(I)

```



```

32  RL(NL)=RN(I)
33  P1(NL)=P(I)
34  IF(WN(I))33.1.33
35  NW=NW+1
36  WN(NW)=WN(I)
37  RW(NW)=RN(I)
38  P2(NW)=P(I)
39  CONTINUE
40  DO 4 I=1,NL
41  XL(I)=ALOG(XN(I))
42  PL(I)=ALOG(P1(I))
43  CONTINUE
44  DO 5 I=1,NW
45  WL(I)=ALOG(WN(I))
46  PW(I)=ALOG(P2(I))
47  CONTINUE
48  CALL LSFIT(NW,2,PW,WL,B)
49  A=EXP(B(1))
50  PRINT 13
51  FORMAT(///)
52  PRINT 14
53  FORMAT(2X,51H COEFFICIENT PLOTTING INFO MAX LENGTH AND WIDTH AND,
54  1///)
55  PRINT 15
56  FORMAT(4X,4HRUN#,5X,4HLOGP,8X,1HP,5X,9HLOGLENGTH,1X,9HNORLENGTH,/)
57  PRINT 16,(RL(I),PL(I),P1(I),XL(I),XN(I),I=1,NL)
58  FORMAT(5F10.5)
59  PRINT 13
60  PRINT 2
61  FORMAT(4X,4HRUN#,5X,4HLOGP,8X,1HP,6X,8HLOGWIDTH,2X,8HNORWIDTH,/)
62  PRINT 21,(RW(I),PW(I),P2(I),WL(I),WN(I),I=1,NW)
63  FORMAT(5F10.5)
64  PRINT 13
65  PRINT 17,A,B(2)
66  FORMAT(17H COEFF FOR WIDTH=F10.5,/,15H EXP FOR WIDTH=F10.5)
67  RN(1)=1,

```



```

100 DO 16, I=2,20
    RN(I)=RN(I-1)+1.
    DO 6, J=1,20
        WI(I)=A+RN(I)*R(2)
        PRINT I,3
        CALL LSFIT(NL,2,PL,XL,B)
        C=EXP(R(1))
        PRINT 18,C,R(2)
18    FORMAT(18H COEFF FOR LENGTH=F10.5,/,16H EXP FOR LENGTH=F10.5)
70    DO 7, I=1,20
        XI(I)=C+RN(I)*R(2)
        PRINT I,3
        PRINT 19
19    FORMAT(6X,23HVALUES FOR FITTED CURVE,/)
        PRINT 21
21    FORMAT(2X,6HUMP,2X,8HNRWIDTH,2X,9HNRLENGTH,/)
        PRINT 21,(RN(I),WI(I),XI(I),I=1,20)
        FORMAT(3F17.5)
15
3180  513.  12877  18.8  164  126  0.  3.56
410  25.  19.  3.87  125  0.  4.06
420  31.  175.  1.66  128  19.05  4.3
430  31.  169.  2.18  127  16.  3.45
440  31.  14.  2.44  126  0.  3.3
450  27.  16.  4.18  125  14.99  3.25
460  26.  19.  3.67  128  0.  3.25
470  27.  215.  3.59  131  18.03  4.
480  32.  20.  2.52  131  18.8  3.95
490  31.  215.  3.9  126  15.75  3.3
500  32.  125.  2.4  131  0.  3.05
510  28.  155.  4.29  132  20.83  3.65
520  29.  17.  1.74  125  20.83  3.56
530  30.  165.  4.  131  17.53  3.7
540  33.  18.  1.82  126  20.07  3.25
550  35.  19.  4.36  126

```



THIS PROGRAM DETERMINES THE CORRELATION PARAMETERS NECESSARY
FOR THE POOL CONTOUR EQUATION.

```

XL=.827
HW=.11
D1=(P(1)*XL/(P(1)+P(2)))*P(1)
D2=(XL-P(1)*XL/(P(1)+P(2)))*P(2)
XK=HW*D2/(D1+D2)
F=((XK*X(1)+P(1)*XL-X(1))*P(2))*.5)
D(1)=F/2.*P(ALOG(P(1)+P(2))-ALOG(P(1)*XL)+ALOG(X(1)))
D(2)=F/2.*P(ALOG(P(1)+P(2))-ALOG(P(2)*XL)+ALOG(XL-X(1)))
RETURN

```

END

```

PROBL4RUN#53 2 2 17 2 .0001 100 1
(2F10.5)

```

```

.02 .031
.059 .079
.198 .098
.138 .116
.177 .11
.216 .098
.256 .098
.295 .087
.335 .079
.374 .075
.413 .067
.453 .063
.492 .059
.531 .059
.571 .055
.689 .043

```

C
C
C
C
C
C
C
C



768 12
MAXIMUM 100 100
MINIMUM 1 1
PARAM 3 75



THIS PROGRAM REDUCES THE DATA AND PUTS THE RESULTS IN
SAME COORDINATE SYSTEM AS THE MODEL.

```

DIMENSION EMFCA(600),TCA(600)
DIMENSION XCHRT(100), YTEMP(100),XCCHRT(100),XCFOR(100),XNFOR(100)
DIMENSION TPN(100),X(50),Y(50),T(100),XTGR(100),TGRP(100)
DIMENSION XEMF(100)
DIMENSION YNDIT(100)
COMMON EMFCA,TCA
READ 3000,NOCA
READ 1000,(EMFCA(I),TCA(I),I=1,NOCA)
READ 3000,NRUN
DO 1 IRUN=1,NRUN
  READ 3000,NRUN
  READ 2000,ALMIN,TMELT,THIKRD,TAMB
  READ 2000,DISTV,TIME,VOLT,AMP
  READ 7000,FANRCA
  READ 3000,NOGRAP
  VEL=DISTV*50/TIME
  DISIN=24*ALMIN/IN/VEL
  TDIFF=TMELT-TAMB
1000 FORMAT(6F10.5)
2000 FORMAT(4F10.5)
3000 FORMAT(I3)
5000 FORMAT(2F10.5)
7000 FORMAT(F10.5)
8999 FORMAT(///)
PRINT 1000
1600 FORMAT(43X,34H EXPERIMENTAL DATA ,///)
PRINT 1500
1500 FORMAT(44X,32H SPECIMEN DATA ,///)

```



```

PRINT 4040
FORMAT (20X,39H MATERIAL .....AISI 1010 STEEL,/)
PRINT 4500, TMELT
FORMAT (51X,37H MELTING TEMPERATURE, (DEG,R) .....F12.5,/)
PRINT 5500, THICKRD
FORMAT (64X,25H THICKNESS, (IN,) .....F12.5,///)
PRINT 6500
FORMAT (45X,29H .....WELDING DATA .....//)
PRINT 7500, VEL
FORMAT (56X,33H SPEED OF TRAVEL, (FT/HR) .....F12.5,/)
PRINT 8000, VOLT
FORMAT (64X,25H VOLTAGE, (VOLTS) .....F12.5,/)
PRINT 8500, AMP
FORMAT (65X,24H CURRENT, (AMPS) .....F12.5,/)
READ 9501, EDIA, EANG
FORMAT (2F10.5)
PRINT 8502
FORMAT (20X,39H TYPE ELECTRODE .....7018-A1,/)
PRINT 8503, EDIA
FORMAT (51X,37H ELECTRODE DIA., (IN,) .....F12.5,/)
PRINT 8504, EANG
FORMAT (51X,37H ELECTRODE ANGLE, (DEG,) .....F12.5,///)
PRINT 9000
FORMAT (35X,49H .....COMPUTATION DATA AND ASSUMPTIONS .....//)
PRINT 9030, TAMB
FORMAT (52X,37H AMBIENT TEMPERATURE, (DEG,R) .....F12.5,/)
PRINT 9040, ALMIN
FORMAT (32X,55H MINIMUM VALUE OF THERMAL DIFFUSIVITY, (FT2/HR) .....
100,F12.5,/)
PRINT 9050, DISIN
FORMAT (53X,35H NORMALIZING DISTANCE, (IN,) .....F12.5,/)
PRINT 9060, TDIFF
FORMAT (48X,41H NORMALIZING TEMPERATURE, (DEG,R) .....F12.5,/)
DO 1004 ID=1, NCGRAP
READ 3000, ITCN
READ 7000, XTCPL

```



```

READ 7000,VELR
READ 3000,NQOP
READ 5001,(XCHRT(I),XEMF(I),I=1,NQOP)
XCCHRT(1)=0.
DO 1001 I=2,NQOP
  XCHRT(I)=XCHRT(I)/2.54
1001 XCCHRT(I)=XCCHRT(I-1)+XCHRT(I)
  XCONV=11.81*VELR/VFL
  XFORC=XCONV*XCPL
DO 1002 I=1,NQOP
1002 XCFOR(I)=XFORC-XCCHRT(I)
DO 1003 I=1,NQOP
  XNFOR(I)=XCFOR(I)/(XCONV*DISIN)
  TMPN(I)=(TMPF(XEMF(I))-TAMB+459.69)/TDIFF
1003 T(I)=TMPN(I)*TDIFF+TAMB
  READ 7000,YDIST
DO 8681 I=1,NQOP
8681 YNDIT(I)=YDIST/DISIN
  PRINT 8999
  PRINT 8680
8680 FORMAT(43X,33H MEASUREMENT DATA NEW MEASUREMENT,/)
  PRINT 13,NRJUN
  FORMAT(70X,19H RUN NUMBER 000000,I3,/)
  PRINT 1011,ITCN
1011 FORMAT(63X,28H THERMOCOUPLE NUMBER 000000,I3,/)
  PRINT 1010
1010 FORMAT(37X,52H TYPE OF THERMOCOUPLE USED IS 000000,CHROMEL-ALUMEL
1,/)
  PRINT 1040,YDIST
1040 FORMAT(28X,56H THERMOCOUPLE DISTANCE FROM LLINE OF SYMMETRY,(IN),0.
1,000,F12.5,/)
  VFLR=VELR/2.36
  PRINT 1030,VELR
1030 FORMAT(60X,29H CHART SPEED,(IN/MIN) 000000,F12.5,/)
  PRINT 2800
2800 FORMAT(33X,50H MEASURED TEMPERATURE DISTRIBUTION 000000,/)

```




```

1//)
PRINT 1250
1250 FORMAT(2X,32HDIM-LESS SIST, FROM LINE OF SYM,5X,25HDIM-LESS DIST,
1FROM ORIGIN,5X,25HDIM-LESS TEMPERATURE,5X,19HTEMPERATURE,(DEG,R))
PRINT 1800,(YNDIT(I),XNFOR(I),TMPN(I),T(I),I=1,NOOP)
1800 FORMAT(12X,F12.5,22X,F12.5,16X,F12.5,13X,F12.5)
1004 CONTINUE
1 CONTINUE
END
FUNCTION TMPF(EMFD)
DIMENSION EMFCA(600),TCA(600)
COMMON FAMBCA
COMMON EMFCA,TCA
VALF(A,B,C,D,E)=A+(B-A)/(D-C)*(E-C)
EMULT=1/(EMFCA(2)-EMFCA(1))
EMFD1=EMFD+EAMBCA
E=EMFD1*EMULT
I=E+1,
TMPF=VALF(TCA(I),TCA(I+1),EMFCA(I),EMFCA(I+1),EMFD1)
RETURN
END
END

```

537

THE FOLLOWING IS A LISTING OF MILLIVOLTS VS. DEGREES F
FOR CHROMEL-ALUMEL THERMOCOUPLES.

32°	01	36.4444	02	40.9629
45.4815	04	51.0	05	54.5454
59.1909	07	63.6363	08	68.1818
72.7272	10	77.2727	101	81.6614
86.1512	103	90.4095	104	94.7688
99.1281	106	103.6581	107	108.188
112.5628	109	116.9376	20	121.2917
125.6459	202	130.	203	134.3478
138.6956	205	143.0435	206	147.3913



2.7	151.7391	2.8	156.0869	2.9	160.4348
3.0	164.7826	3.1	169.1304	3.2	173.311
3.3	177.4916	3.4	181.8049	3.5	186.1181
3.6	190.4562	3.7	194.7943	3.8	199.1324
3.9	203.4794	4.0	207.8265	4.1	212.1742
4.2	216.5218	4.3	220.8696	4.4	225.2174
4.5	229.5652	4.6	233.9130	4.7	238.2609
4.8	242.6187	4.9	246.9565	5.0	251.3043
5.1	255.6522	5.2	260.0	5.3	264.5454
5.4	269.1919	5.5	273.4545	5.6	277.8182
5.7	282.33	5.8	286.8417	5.9	291.3791
6.0	295.9164	6.1	300.4595	6.2	305.1026
6.3	309.5457	6.4	314.091	6.5	318.6364
6.6	323.117	6.7	327.3776	6.8	331.8856
6.9	336.3926	7.0	340.9291	7.1	345.4645
7.2	350.0	7.3	354.5454	7.4	359.1909
7.5	362.6364	7.6	368.1818	7.7	372.5589
7.8	376.9361	7.9	381.449	8.0	385.9457
8.1	390.4792	8.2	395.13	8.3	399.5466
8.4	403.922	8.5	408.2578	8.6	412.774
8.7	417.2903	8.8	421.8294	8.9	426.3686
9.0	430.7658	9.1	435.1631	9.2	439.5613
9.3	444.0993	9.4	448.6383	9.5	453.1182
9.6	457.3721	9.7	461.8859	9.8	466.3937
9.9	470.7828	10.0	475.172	10.1	479.5611
10.2	483.9116	10.3	488.2602	10.4	492.6083
10.5	496.9563	10.6	501.4542	10.7	505.952
10.8	510.342	10.9	514.732	11.0	519.122
11.1	523.4732	11.2	527.8244	11.3	532.1728
11.4	536.5212	11.5	540.8692	11.6	545.2172
11.7	549.5652	11.8	553.913	11.9	558.2609
12.0	562.6187	12.1	566.9565	12.2	571.3043
12.3	575.6522	12.4	580.0	12.5	584.3478
12.6	588.6956	12.7	593.0435	12.8	597.3913
12.9	601.7391	13.0	606.1869	13.1	610.4348
13.2	614.7826	13.3	619.1314	13.4	623.4783



13.5	627.8261	13.6	632.1739	13.7	636.5217
13.8	640.733	13.9	644.9456	14.	649.1576
14.1	653.4946	14.2	657.8315	14.3	662.1774
14.4	666.5233	14.5	667.6448	14.6	668.7663
14.7	669.8878	14.8	677.9327	14.9	685.5776
15.	69.9856	15.1	695.9036	15.2	701.5118
15.3	7.5013	15.4	709.5482	15.5	713.7289
15.6	717.9096	15.7	722.2276	15.8	726.5456
15.9	730.75	16.	734.9546	16.1	739.1591
16.2	743.4955	16.3	747.8312	16.4	752.2277
16.5	756.2237	16.6	76.5288	16.7	764.8339
16.8	769.139	16.9	773.3163	17.	777.4936
17.1	781.8062	17.2	786.1187	17.3	79.3252
17.4	794.5316	17.5	798.7381	17.6	803.6696
17.7	807.4011	17.8	811.6017	17.9	815.8284
18.	82.	18.1	824.3478	18.2	828.6956
18.6	845.6897	18.7	85.	18.8	854.1667
18.9	858.3333	19.	862.5	19.1	866.6667
19.2	871.9677	19.3	875.2688	19.4	879.5699
19.5	883.7419	19.6	887.914	19.7	892.816
19.8	896.2492	19.9	900.4161	2.	94.583
20.1	9.87469	20.2	913.769	20.3	917.4038
20.4	921.6025	20.5	925.813	20.6	93.
20.7	934.1667	20.8	938.3333	20.9	942.6543
21.	946.9753	21.1	951.1768	21.2	955.3783
21.3	959.5798	21.4	963.9216	21.5	968.2633
21.6	972.455	21.7	976.6466	21.8	981.8196
21.9	984.9025	22.	989.1654	22.1	993.3326
22.2	997.4997	22.3	1001.812	22.4	1006.1206
22.5	1010.3264	22.6	1014.5323	22.7	1018.7382
22.8	1022.9192	22.9	1027.1832	23.	1031.3868
23.1	1035.6934	23.2	1040.	23.3	1044.1666
23.4	1048.3333	23.5	1052.49999	23.6	1056.6666
23.7	1060.9676	23.8	1065.2687	23.9	1069.5698
24.	1073.7418	24.1	1077.9139	24.2	1081.7367
24.3	1085.5595	24.4	1088.996	24.5	1092.4325



24.6	1096.869	24.7	1101.9193	24.8	1106.9697
24.9	1111.3131	25.	1115.6565	25.1	1120.
25.2	1124.1666	25.3	1128.3333	25.4	1132.6543
25.5	1136.9753	25.6	1141.1768	25.7	1145.3783
25.8	1149.5798	25.9	1153.7478	26.	1157.9159
26.1	1162.2316	26.2	1166.5474	26.3	1171.7513
26.4	1174.9552	26.5	1179.1592	26.6	1183.4955
26.7	1187.8318	26.8	1192.1775	26.9	1196.5233
27.	1200.7347	27.1	1204.946	27.2	1209.1577
27.3	1213.4946	27.4	1217.8315	27.5	1222.1275
27.6	1226.2235	27.7	1230.5286	27.8	1234.8228
27.9	1239.1389	28.	1243.4833	28.1	1247.8277
28.2	1252.125	28.3	1256.2223	28.4	1261.5278
28.5	1264.8333	28.6	1269.1388	28.7	1273.3161
28.8	1277.4935	28.9	1281.816	29.	1286.1186
29.1	1291.4565	29.2	1294.7944	29.3	1299.1324
29.4	1303.3122	29.5	1307.492	29.6	1311.851
29.7	1316.1182	29.8	1320.4562	29.9	1324.7943
30.	1329.1323	30.1	1333.4793	30.2	1337.8264
30.3	1342.6241	30.4	1346.2219	30.5	1350.5275
30.6	1354.8332	30.7	1359.1388	30.8	1362.4832
30.9	1367.8277	31.	1372.1749	31.1	1376.5222
31.2	1380.734	31.3	1384.9458	31.4	1389.1576
31.5	1393.4945	31.6	1397.8315	31.7	1402.1773
31.8	1406.5232	31.9	1410.8715	32.	1415.2179
32.1	1419.5652	32.2	1423.913	32.3	1428.2618
32.4	1432.6136	32.5	1436.9565	32.6	1441.3143
32.7	1445.6521	32.8	1450.	32.9	1454.5454
33.	1459.1909	33.1	1463.4545	33.2	1467.8181
33.3	1472.1687	33.4	1476.5189	33.5	1481.8679
33.6	1485.2165	33.7	1489.5651	33.8	1494.102
33.9	1498.639	34.	1503.1185	34.1	1507.3782
34.2	1511.735	34.3	1516.1828	34.4	1520.4319
34.5	1524.781	34.6	1529.1311	34.7	1533.6592
34.8	1538.1883	34.9	1542.563	35.	1546.9377
35.1	1551.4419	35.2	1555.9461	35.3	1560.3379



35.4	1564.7297	35.5	1569.1216	35.6	1573.4729
35.7	1577.8243	35.8	1582.3338	35.9	1586.8432
36.0	1591.38	36.1	1595.9168	36.2	1611.3178
36.3	1614.7188	36.4	1619.1198	36.5	1613.6532
36.6	1618.1866	36.7	1622.5619	36.8	1626.9372
36.9	1631.4416	37.0	1635.546	37.1	1641.4795
37.2	1645.13	37.3	1649.5466	37.4	1653.921
37.5	1658.2577	37.6	1662.7739	37.7	1667.2912
37.8	1671.8294	37.9	1676.3686	38.0	1681.9124
38.1	1685.4562	38.2	1691.38.3	38.3	1694.3478
38.4	1698.6956	38.5	1713.2173	38.6	1717.7391
38.7	1712.281	38.8	1716.8212	38.9	1721.3656
39.0	1725.91	39.1	1731.4551	39.2	1735.1123
39.3	1739.5454	39.4	1744.918	39.5	1748.6363
39.6	1753.1817	39.7	1757.7272	39.8	1762.4485
39.9	1767.1678	40.0	1771.7597	40.1	1776.3336
40.2	1781.889	40.3	1785.4445	40.4	1791.1
40.5	1794.5454	40.6	1799.199	40.7	1813.6363
40.8	1818.1818	40.9	1812.99	41.0	1817.6363
41.1	1822.2154	41.2	1826.7945	41.3	1831.3481
41.4	1835.9117	41.5	1841.611	41.6	1845.315
41.7	1851.1	41.8	1854.5454	41.9	1859.1
42.0	1863.6363	42.1	1868.182	42.2	1872.9.9
42.3	1877.6363	42.4	1882.2154	42.5	1886.7945
42.6	1891.5117	42.7	1896.227	42.8	1911.9763
42.9	1915.7256	43.0	1911.3312	43.1	1914.9348
43.2	1919.529	43.3	1924.2942	43.4	1929.149
43.5	1933.8113	43.6	1938.5715	43.7	1943.1429
43.8	1947.7143	43.9	1952.4395	44.0	1957.1648
44.1	1961.9185	44.2	1966.6723	44.3	1971.208
44.4	1975.864	44.5	1981.5758	44.6	1985.2879
44.7	1991.1	44.8	1994.7619	44.9	1999.5238
45.0	2004.2857	45.1	2019.476	45.2	2013.8095
45.3	2018.5714	45.4	2031.6666	45.5	2044.7619
45.9	2047.912	46.0	2052.389	46.1	2056.866
46.2	2061.5567	46.3	2066.2474	46.4	2071.159



46.5	2076.6776	46.6	218.8738	46.7	2985.677
46.8	209.4513	46.9	2095.2256	47.0	219.0
47.1	214.7619	47.2	219.5238	47.3	2114.5124
47.4	2119.511	47.5	2124.2733	47.6	2129.4455
47.7	2134.248	47.8	2139.141	47.9	2143.7849
48.0	2146.5657	48.1	2153.5371	48.2	2158.5185
48.3	2163.2966	48.4	2168.1847	48.5	2173.1494
48.6	2173.141	48.7	2182.8.84	48.8	2187.6.28
48.9	2192.5616	49.0	2197.52.5	49.1	22.2.5123
49.2	22.7.5.41	49.3	2212.5.24	49.4	2217.5.8
49.5	2222.5.14	49.6	2227.5.1	49.7	2232.3.77
49.8	2237.1154	49.9	2242.71	50.0	2247.266
50.1	2252.163	50.2	2257.6.61	50.3	2262.1037
50.4	2267.21.8	50.5	2272.2.08	50.6	2277.4.02
50.7	2282.44.1	50.8	2287.48	50.9	2292.488
51.0	2297.496	51.1	23.2.4976	51.2	23.7.4992
51.3	2312.4995	51.4	2317.4998	51.5	2322.7.82
51.6	2327.9166	51.7	2332.9513	51.8	2337.9861
51.9	2343.2.95	52.0	2348.4329	52.1	2353.462
52.2	2358.4912	52.3	2363.7224	52.4	2368.9537
52.5	2373.9747	52.6	2378.9957	52.7	2384.2358
52.8	2383.4759	52.9	2394.4873	53.0	2399.4988
53.1	2404.7494	53.2	241.0	53.3	2415.2631
53.4	242.84.3	53.5	2425.68.6	53.6	2431.0
53.7	2436.0	53.8	2441.0	53.9	2446.442
54.0	2451.8823	54.1	2456.7226	54.2	2462.0
54.2					
1.8	318.0	116	5.3.0		
1.287	19	29.0	135.0		
4.98					
9					
1					
1.89	51.0				
1					
1.37					
25.0					



27	0.	0.		
1	0.	0.		
5	0.	0.		
5	1.5	1.5		
5	4.	4.		
5	10.5	10.5		
5	15.	15.		
5	19.3	19.3		
5	22.	22.		
5	23.	23.		
5	23.	23.		
5	22.6	22.6		
5	21.	21.		
5	19.5	19.5		
5	18.	18.		
5	16.7	16.7		
5	15.1	15.1		
5	12.5	12.5		
5	10.5	10.5		
5	8.2	8.2		
5	7.	7.		
5	6.	6.		
5	5.	5.		
5	4.7	4.7		
5	4.	4.		
5	3.5	3.5		
7	2.	2.		
15				
110				
1287	318.	318.	316	503.
4.31	17	27.	27.	14.
9				
11				
189	51.			
1				



APPENDIX 9EXPERIMENTAL DATA

The experimental data has been arranged as follows:

- A. Data for the maximum dimensions of air welded pools.
- B. Data for the maximum dimensions of underwater pools.
- C. Pool contour measurements.
- D. Data for temperature distribution with sample thermocouple output tapes.



A. DATA FOR MAXIMUM DIMENSIONS OF AIR WELDED POOLS:

RUN#	VOLT	AMP	VEL (FT/HR)	Xm (IN)	* Xm	Ym (IN)	* Ym	P	PLATE THICKNESS (IN)
1	24	175	93.86	--	--	0.40	12.15	4.29	0.123
2	22	170	105.23	0.56	18.90	0.26	8.85	4.03	0.122
3	25	175	99.22	--	--	0.26	8.34	4.71	0.122
4	23	220	112.78	0.72	26.29	0.29	10.92	5.41	0.123
5	28	175	82.55	0.64	17.14	--	--	5.12	0.126
6	25	225	130.13	0.87	36.80	0.29	12.27	6.07	0.122
7	27	200	111.21	0.95	34.28	0.36	12.95	5.78	0.123
8	22	185	79.09	0.73	18.64	0.33	8.46	4.42	0.121
9	29	230	116.25	1.32	49.46	0.41	15.40	7.19	0.122
10	24	172	100.96	0.84	27.27	--	--	4.28	0.127



RUN#	VOLT	AMP	VEL (FT/HR)	Xm (IN)	Xm [*]	Ym (IN)	Ym [*]	P	PLATE THICKNESS (IN)
11	21	143	84.14	--	--	0.26	7.07	3.14	0.126
12	24	170	71.21	--	--	0.30	6.91	4.26	0.126
13	22	175	98.18	0.56	17.76	0.27	8.51	4.02	0.126
14	23	170	90.37	--	--	--	--	3.93	0.131
15	26	200	93.79	0.94	28.68	--	--	5.34	0.128
16	28	210	107.27	0.88	30.47	0.35	12.14	5.95	0.130
17	25	225	110.48	1.11	39.55	0.27	9.57	5.87	0.126
18	25	225	110.22	0.81	28.78	0.38	13.56	5.83	0.127
19	27	245	134.00	1.29	55.99	0.34	14.68	6.85	0.127



RUN#	VOLT	AMP	VEL (FT/HR)	Xm (IN)	* Xm	Ym (IN)	* Ym	P	PLATE THICKNESS (IN)
21	32	225	108.33	0.66	23.32	--	--	2.47	0.387
22	30	250	115.33	0.86	32.32	0.44	16.46	2.62	0.381
23	32	250	130.62	1.11	47.00	0.33	13.98	2.79	0.382
24	30	255	122.57	--	--	--	--	2.64	0.385
25	29	250	117.03	1.04	39.21	0.33	12.52	2.50	0.386
26	28	240	118.60	0.74	28.56	0.33	12.69	2.32	0.385
27	30	220	117.99	--	--	--	--	2.29	0.384
28	26	240	118.89	0.96	36.80	0.32	12.11	2.14	0.388
29	25	225	121.59	0.75	29.43	0.30	11.93	1.97	0.379
30	26	220	112.83	0.55	19.98	0.31	11.49	2.01	0.379



RUN#	VOLT	AMP	VEL (FT/HR)	Xm (IN)	* Xm	Ym (IN)	* Ym	P	PLATE THICKNESS (IN)
31	30	240	121.99	--	--	0.44	17.40	2.50	0.383
32	27	240	108.89	0.88	31.21	0.28	9.85	2.25	0.383
33	27	230	116.89	--	--	0.40	15.19	2.14	0.386
34	29	210	107.65	0.81	28.11	0.37	13.03	2.12	0.382
35	26	210	103.16	0.67	22.34	0.35	11.69	1.91	0.381
36	26	190	86.15	0.47	13.06	0.33	9.33	1.70	0.387
37	25	185	88.81	0.79	22.63	0.33	9.50	1.61	0.38
38	21	186	95.17	0.61	18.67	0.34	10.42	1.33	0.390



B. DATA FOR MAXIMUM DIMENSIONS OF UNDERWATER POOL

RUN#	VOLT	AMP	VEL (FT/HR)	Xm (IN)	Xm*	Ym (IN)	Ym*	P	PLATE THICKNESS (IN)
41	25	190	117.99	--	--	0.28	10.70	4.90	0.126
42	31	175	80.58	--	--	0.32	8.33	5.65	0.125
43	31	169	72.67	0.75	17.63	0.34	7.96	5.32	0.128
44	31	140	93.85	0.63	19.12	0.27	8.25	4.45	0.127
45	27	160	116.11	--	--	0.26	9.76	4.45	0.126
46	26	190	107.94	0.59	20.61	0.26	8.94	5.14	0.125
47	27	205	112.19	--	--	0.26	9.29	5.62	0.218
48	32	200	114.55	0.71	26.31	0.31	11.67	6.35	0.131
49	31	215	114.70	0.74	27.47	0.31	11.54	6.61	0.131



RUN#	VOLT	AMP	VEL (FT/HR)	Xm (IN)	* Xm	Ym (IN)	* Ym	P	PLATE THICKNESS (IN)
50	32	125	85.71	0.62	17.20	0.26	7.21	4.13	0.126
51	28	155	112.89	--	--	0.24	8.77	4.31	0.131
52	29	170	108.75	0.82	28.86	0.29	10.11	4.86	0.132
53	30	165	125.00	0.82	33.17	0.28	11.34	5.15	0.125
54	33	180	113.75	0.69	25.40	0.29	10.72	5.90	0.131
55	35	190	145.33	0.79	37.16	0.25	12.03	6.86	0.126
56	34	195	125.29	--	--	0.36	14.52	3.42	0.252
57	33	205	136.43	0.83	36.64	0.30	13.38	3.51	0.251
58	34	225	174.09	--	--	0.31	17.30	3.95	0.252
59	37	210	146.94	--	--	0.30	14.41	3.98	0.254
60	38	220	175.89	--	--	0.32	18.37	4.30	0.253
61	38	230	182.14	0.87	51.28	0.31	18.33	4.53	0.251



C. POOL CONTOUR MEASUREMENTS

RUN #4		RUN #8		RUN #13	
X (mm)	Y (mm)	X (mm)	Y (mm)	X (mm)	Y (mm)
0.0	0.0	0.0	0.0	0.0	0.0
1.5	2.0	1.4	2.9	1.1	2.8
2.5	2.8	2.4	3.6	2.1	3.1
3.5	3.1	3.4	3.9	3.1	3.3
4.5	3.6	4.4	4.2	4.1	3.4
5.5	3.8	5.4	4.0	5.1	3.2
6.5	3.7	6.4	3.6	6.1	3.0
7.5	3.5	7.4	3.4	7.1	2.8
8.5	3.3	8.4	3.1	8.1	2.4
9.5	3.1	9.4	3.2	9.1	2.3
10.5	2.8	10.4	2.9	10.1	2.1
11.5	2.5	11.4	3.0	11.1	1.6
12.5	2.2	12.4	2.8	12.1	0.9
13.5	1.8	13.4	2.7	13.1	0.5
14.5	1.4	14.4	2.6	14.2	0.0
15.5	1.1	15.4	2.5		
16.5	0.7	16.4	2.2		
17.5	0.4	17.4	2.0		
18.3	0.0	18.5	0.0		



RUN #53

X (mm)	Y (mm)
0.0	0.0
0.5	0.8
1.5	2.0
2.5	2.5
3.5	2.7
4.5	2.8
5.5	2.5
6.5	2.5
7.5	2.2
8.5	2.0
9.5	1.9
10.5	1.7
11.5	1.6
12.5	1.5
13.5	1.5
14.5	1.4
17.5	1.1
19.5	0.5
21.0	0.0

RUN #16

X (mm)	Y (mm)
0.0	0.0
1.8	2.4
3.8	2.7
5.8	2.4
7.8	2.6
9.8	2.4
11.8	2.2
13.8	2.1
15.8	2.0
17.8	1.7
19.8	1.2
21.8	0.7
22.3	0.0



D. WELDING VARIABLES FOR TEMPERATURE DISTRIBUTION

RUN#	VOLT	AMP	BEAD LENGTH (IN)	TIME (MIN)	PLATE THICK. (IN)	AMBIENT TEMP.
100*	33	155	6.13	0.22	0.255	73
101	34	170	1.61	0.06	0.116	43
102	28	140	4.87	0.24	0.116	43
103	29	145	5.30	0.23	0.116	43
104	26	150	5.77	0.23	0.116	43
105	28	145	6.03	0.24	0.116	43
106	27	145	4.75	0.17	0.116	43
107	30	140	4.58	0.17	0.116	43
108	29	135	4.98	0.19	0.116	43
109	27	143	5.47	0.20	0.255	43
110	27	140	4.31	0.17	0.116	43
111	27	140	4.96	0.18	0.116	43



THERMOCOUPLE OUTPUT TAPES

CHART SPEED: 25 mm/sec

VERTICAL SCALE: 1 MILLIVOLT/mm



RUN NO. 106, T/C NOS. 5,3,4

EVENT MARK

106-5

3

106-3

4

106-4

6

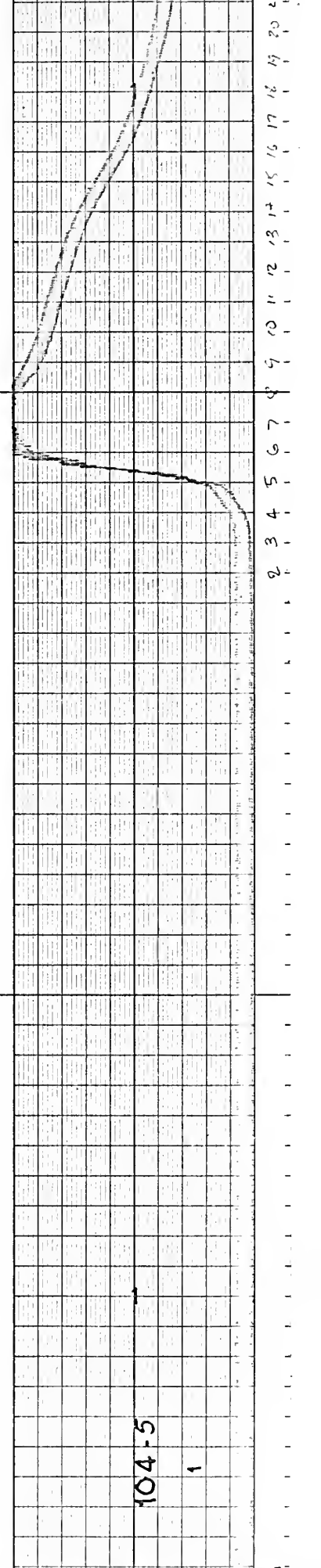
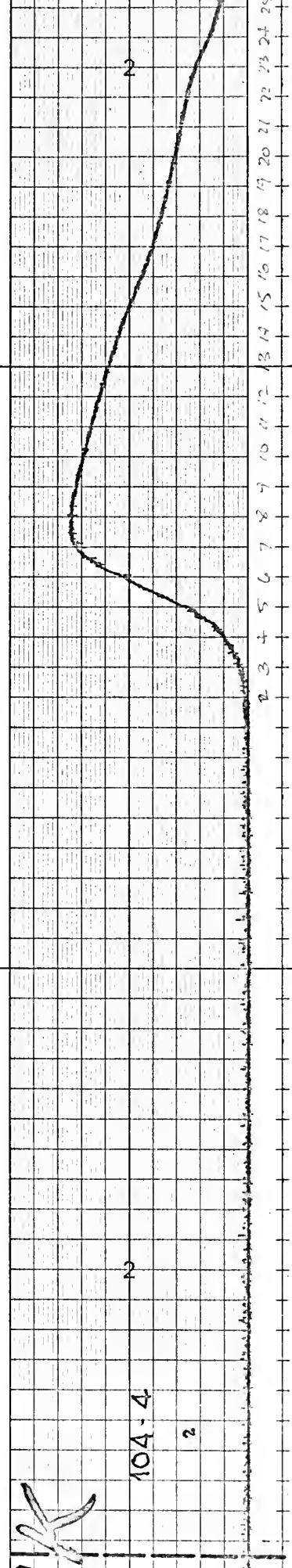
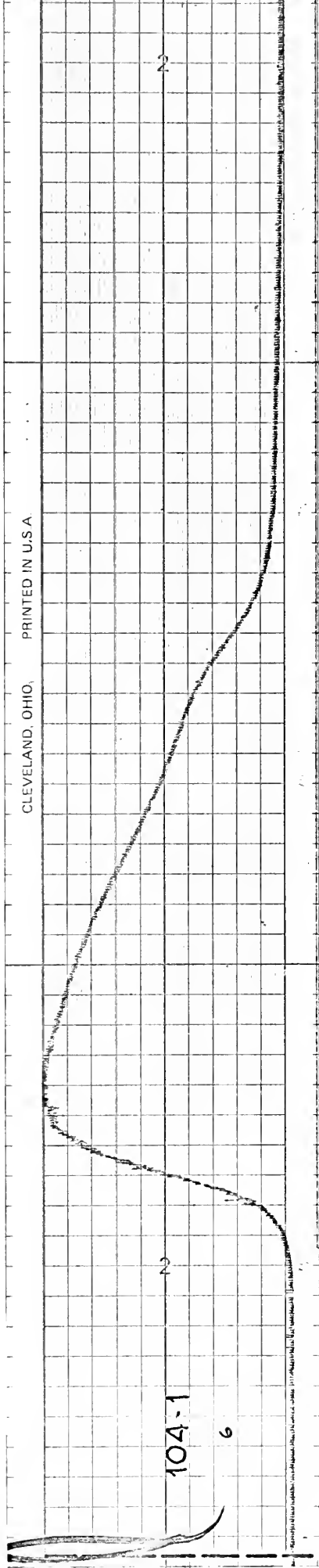
2

2



RUN NO. 104, T/C NOS. 1, 4, 5

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RUN NO. 108, T/C NO. 1

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108-1

2

2

RUN NO. 110, T/C NO. 11

110-11

1

2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26

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